

April 24, Week 14

Today: Chapter 13, Kepler's Laws

Exam #4, Next Friday, April 26

Practice Exam on Website.

Review Sessions: Thursday, April 25, 5:15PM, 114 Regener Hall

After the exam, we will start chapter 14.

Kepler's Laws

Before Newton, all astronomical work had been observational. Using the data of Danish astronomer Tycho Brahe (1546-1601), the German mathematician Johannes Kepler (1571-1630) was able to deduce (but not explain), three statements about planetary motion.

Kepler's Laws

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Kepler's Laws:

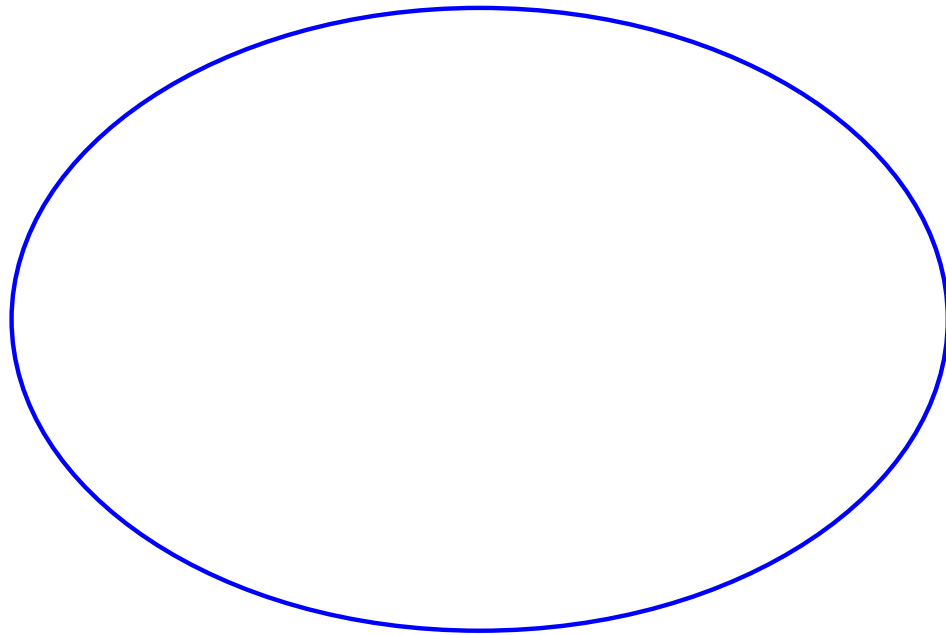
- 1: Each planet's orbit traces out the shape of an ellipse with the sun located at one focus.
- 2: The imaginary line from the sun to a planet sweeps out equal areas in equal times.
- 3: The period of the planet's motion is proportional to the orbit's semi-major axis to the $\frac{3}{2}$ power.

Ellipses - The Geometric Approach

Ellipse - ovals.

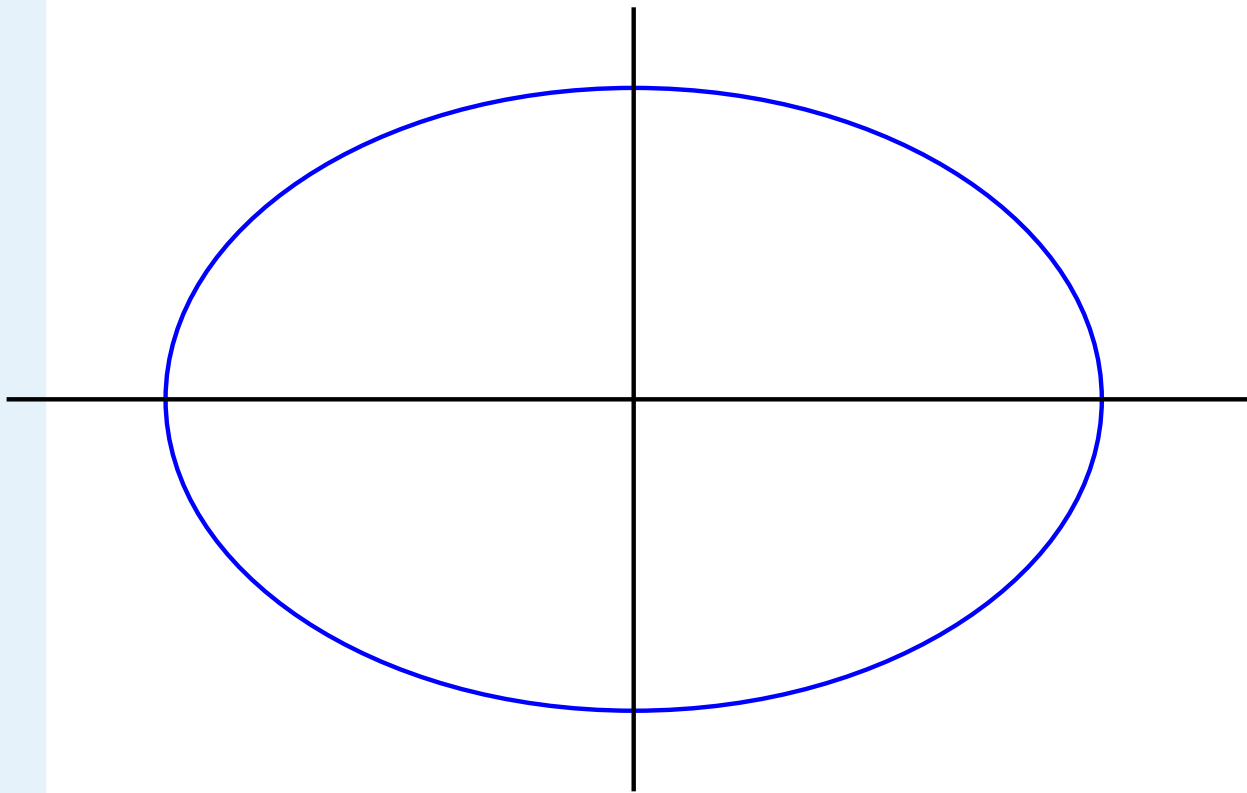
Ellipses - The Geometric Approach

Ellipse - ovals.



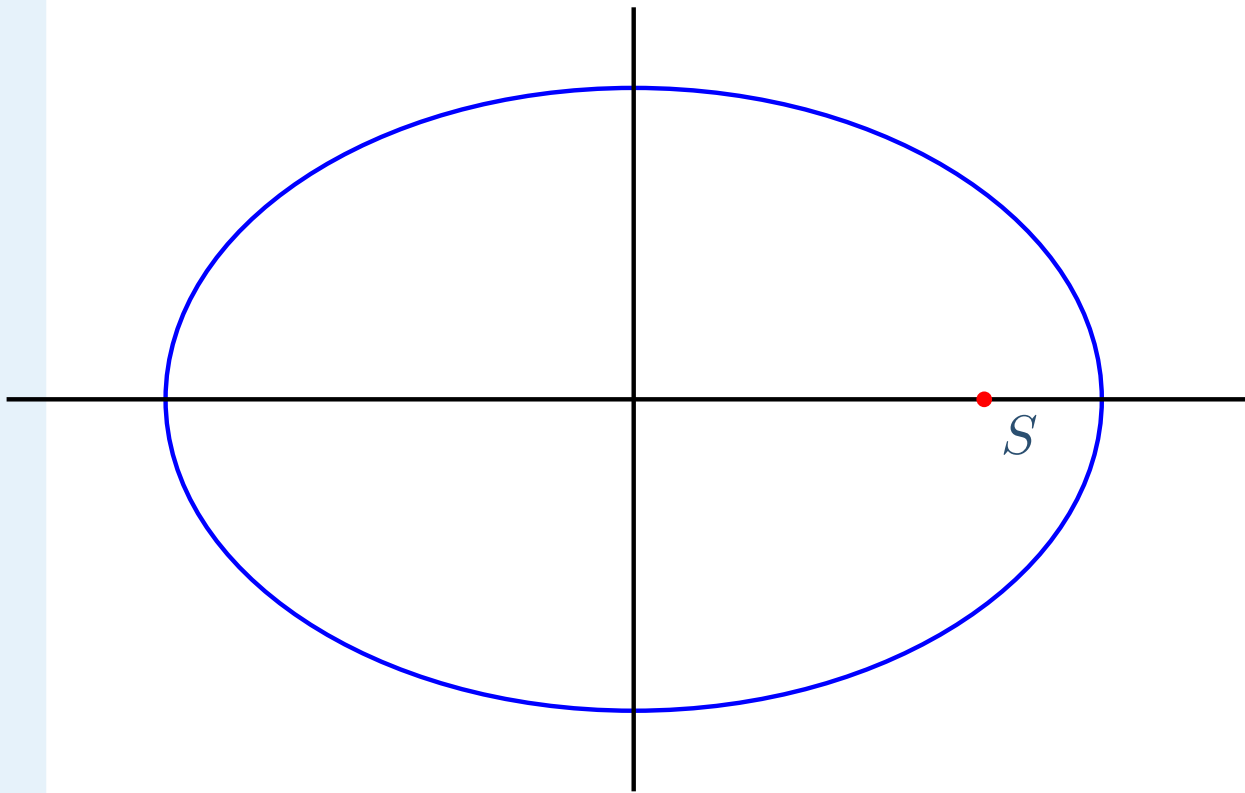
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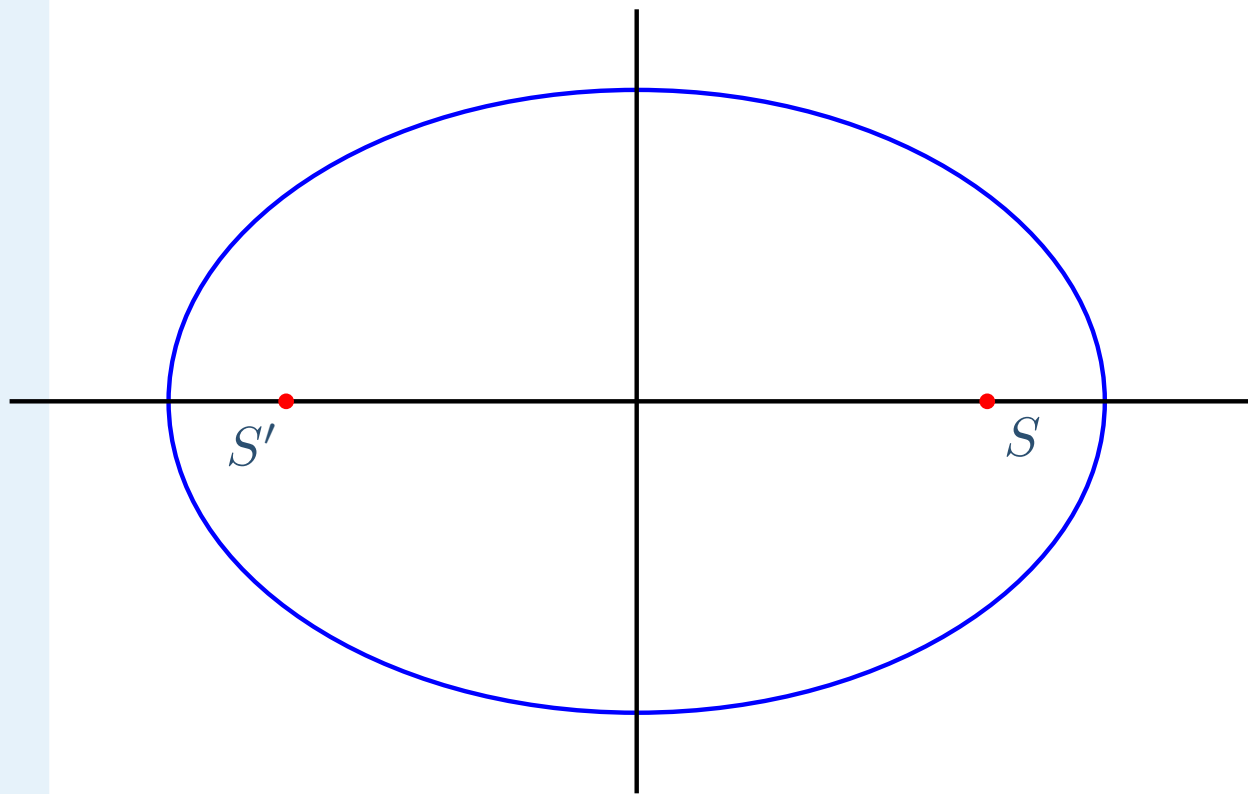
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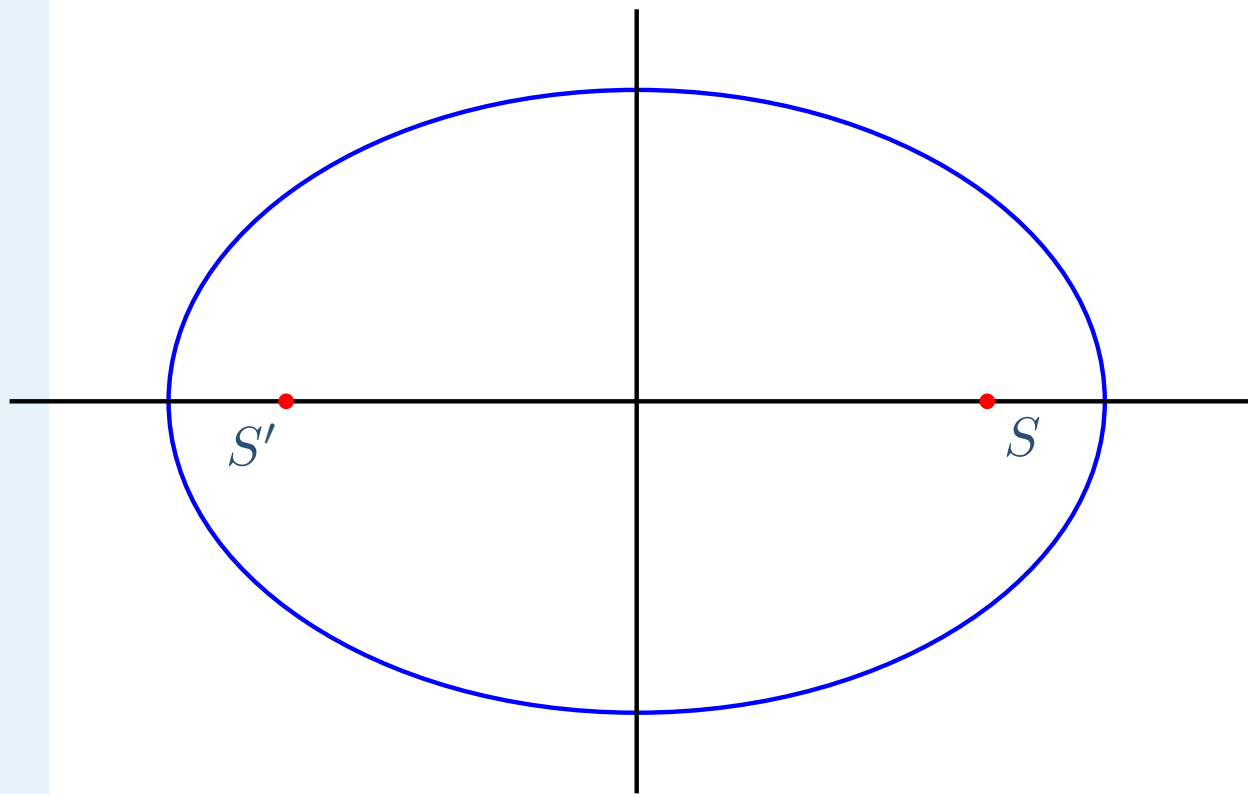
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Ellipses - The Geometric Approach

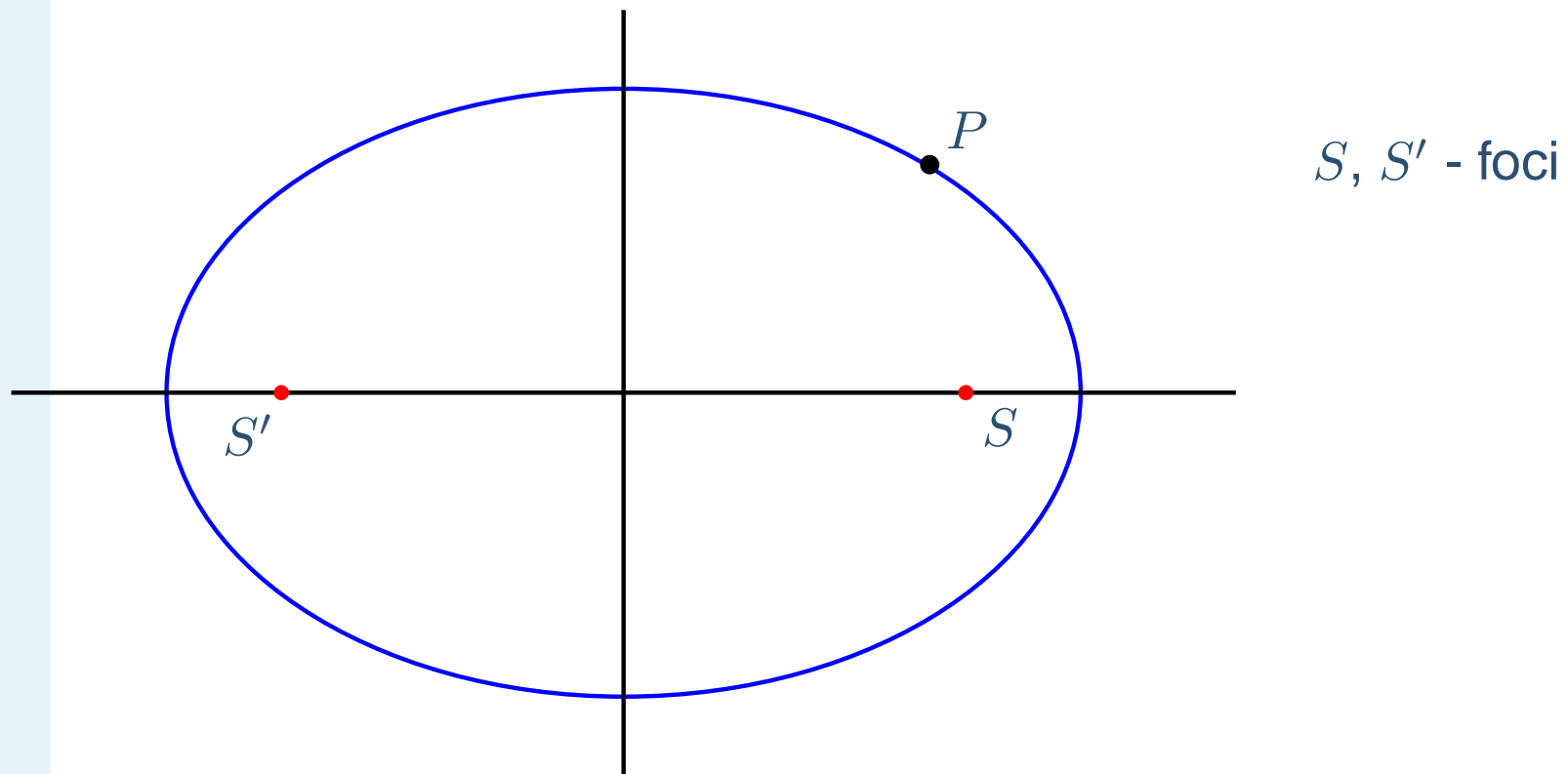
Ellipse - ovals.



S, S' - foci

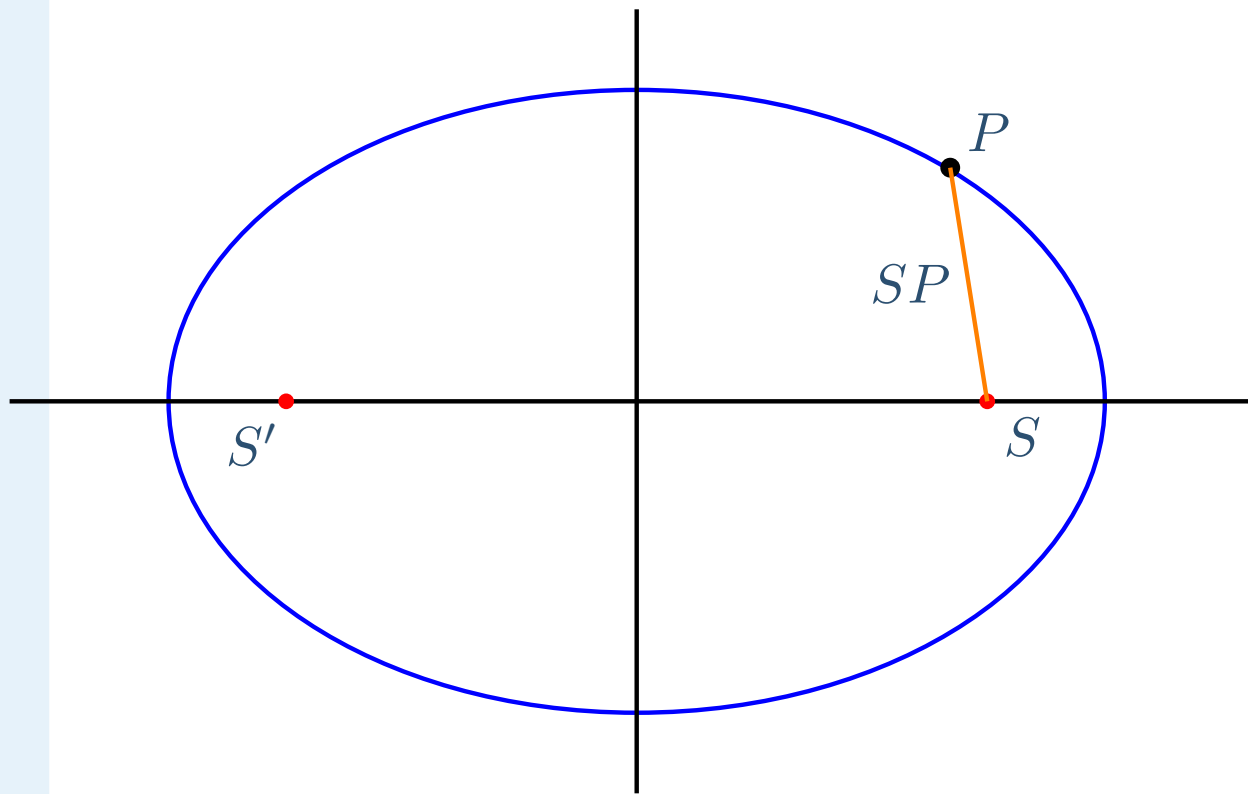
Ellipses - The Geometric Approach

Ellipse - ovals.



Ellipses - The Geometric Approach

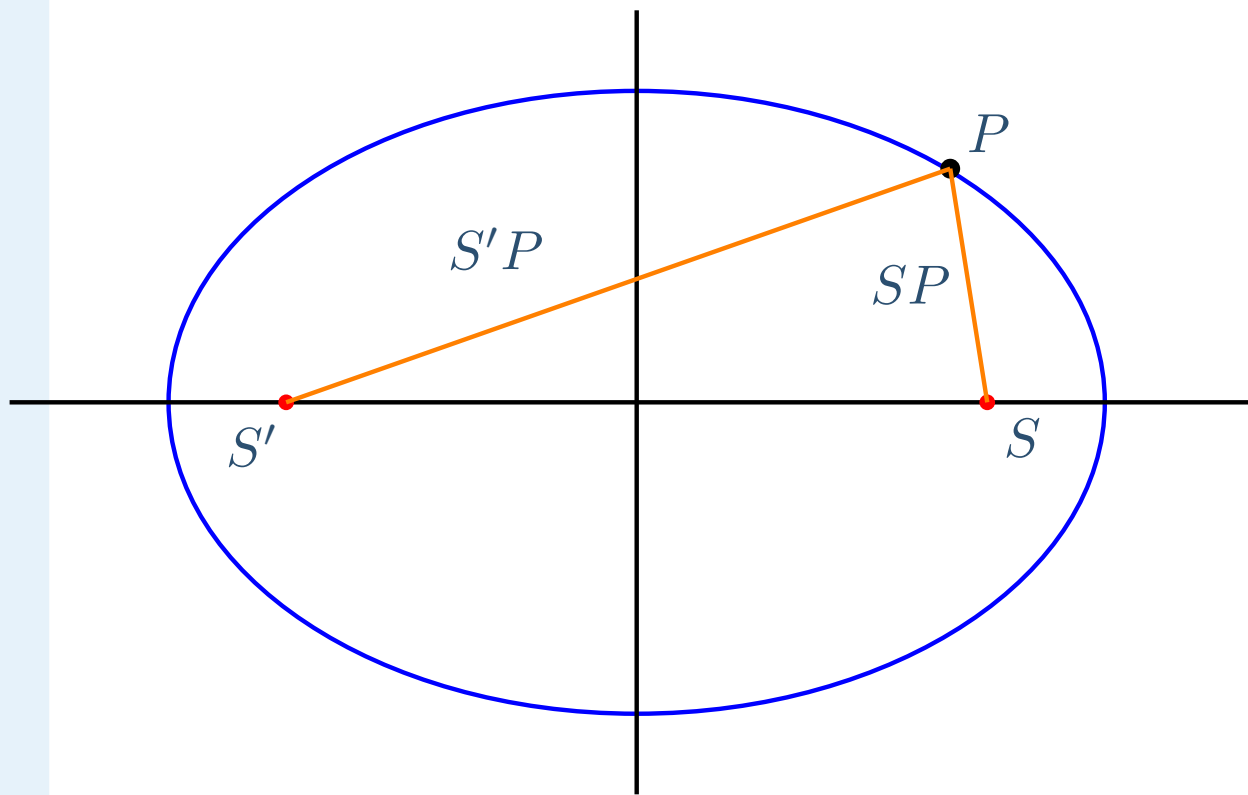
Ellipse - ovals.



S, S' - foci

Ellipses - The Geometric Approach

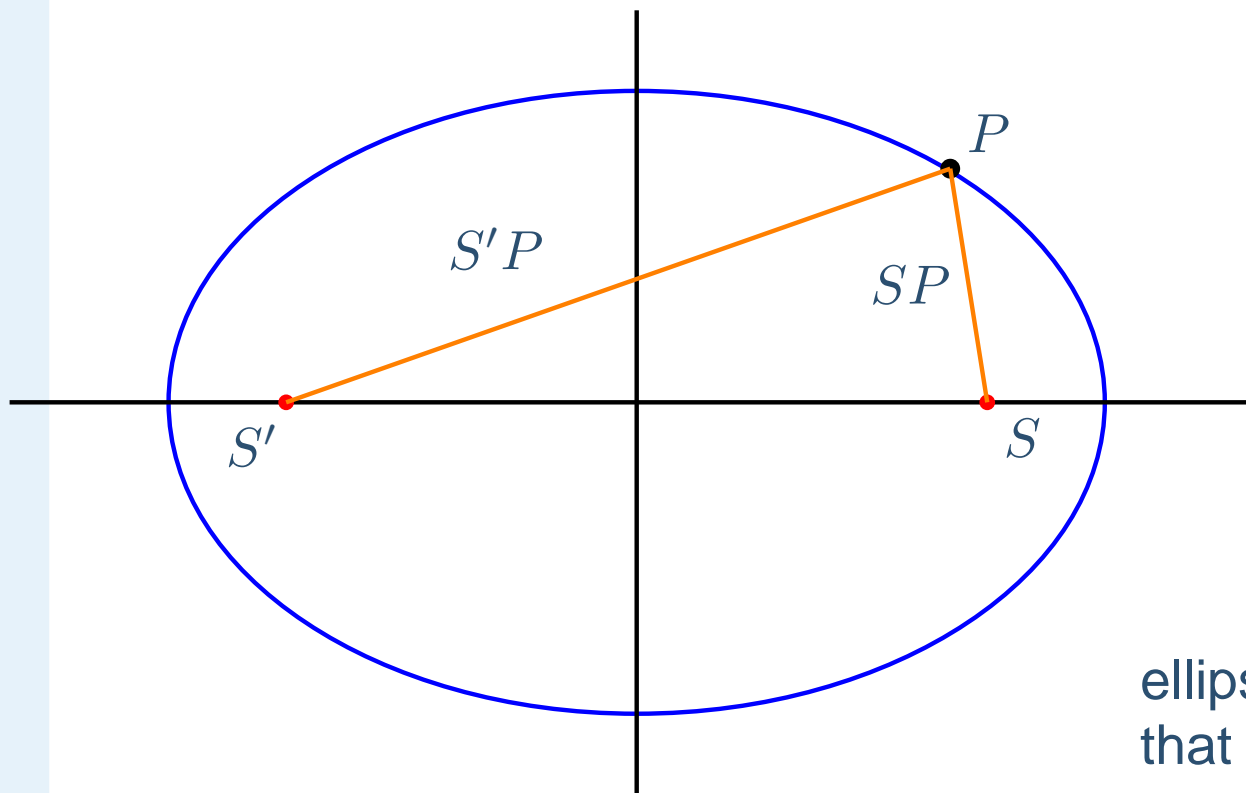
Ellipse - ovals.



S, S' - foci

Ellipses - The Geometric Approach

Ellipse - ovals.

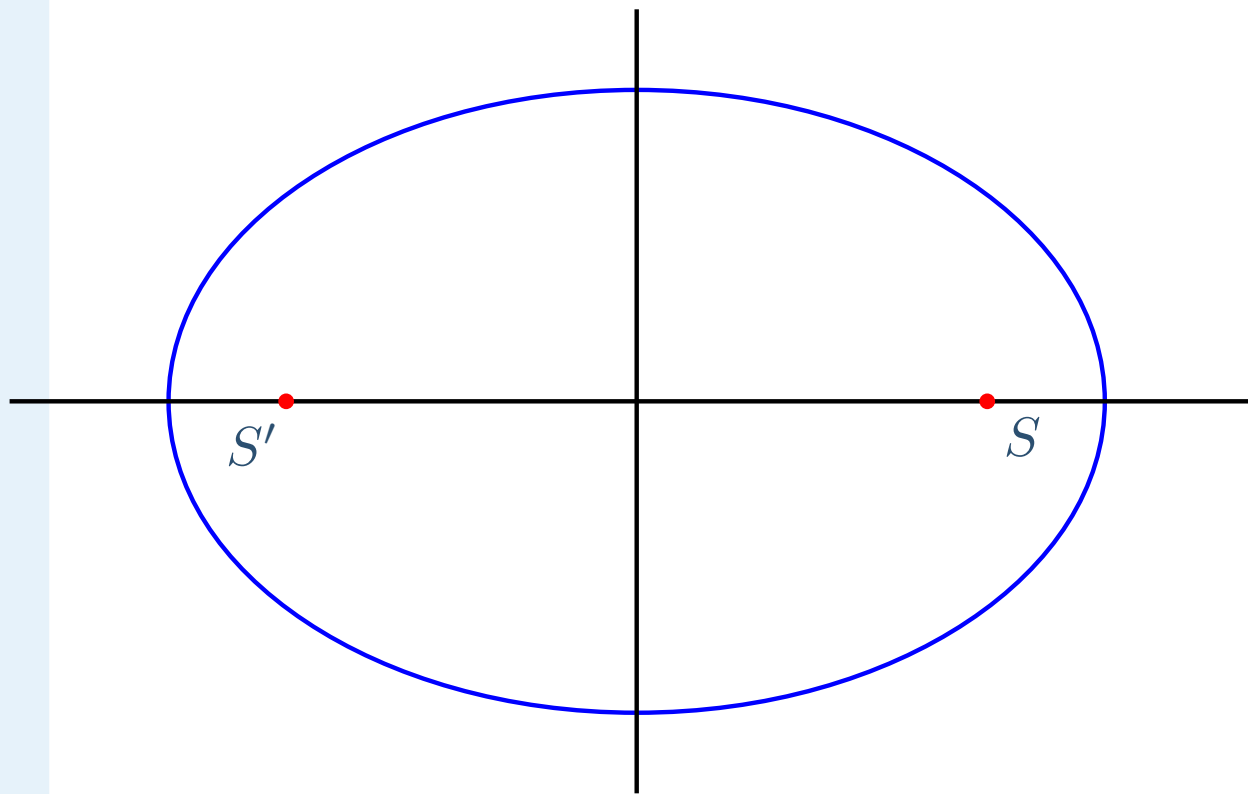


S, S' - foci

ellipse = all points P such
that $SP + S'P = \text{constant}$

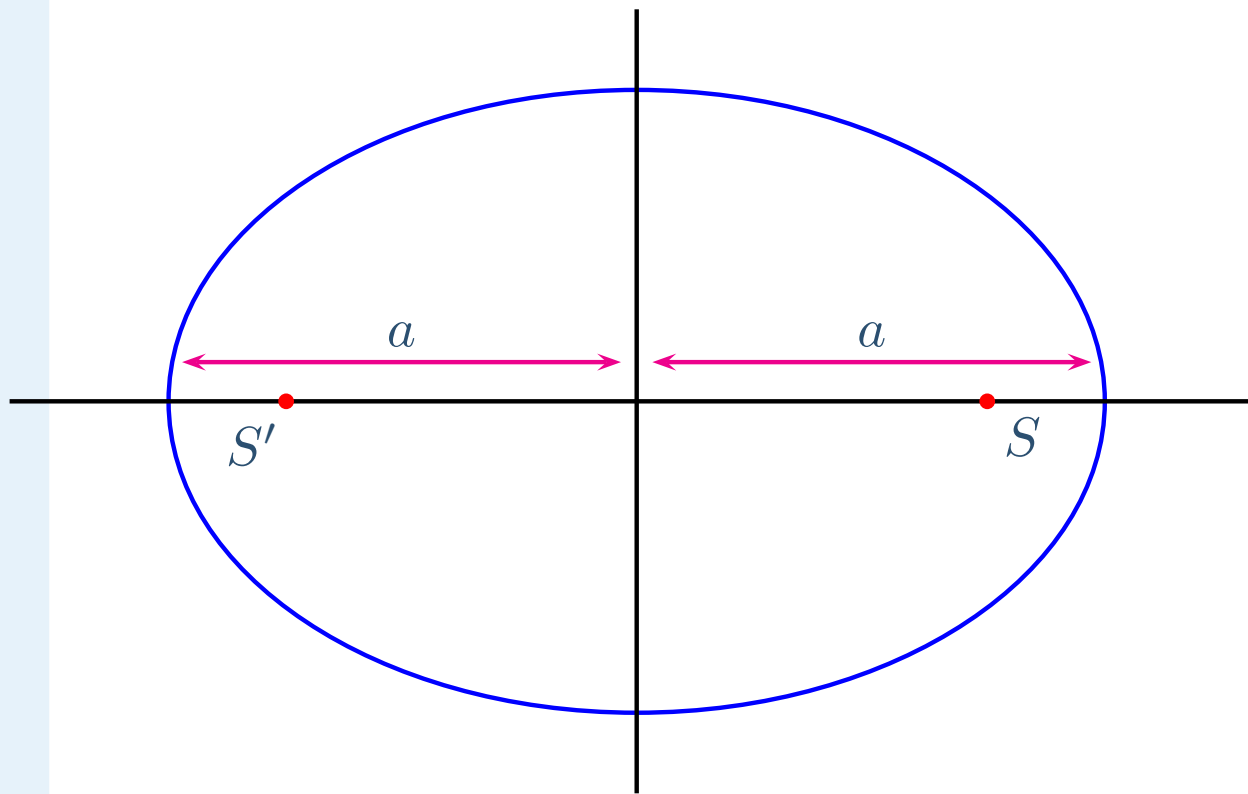
Ellipses - The Algebraic Approach

Ellipse - ovals.



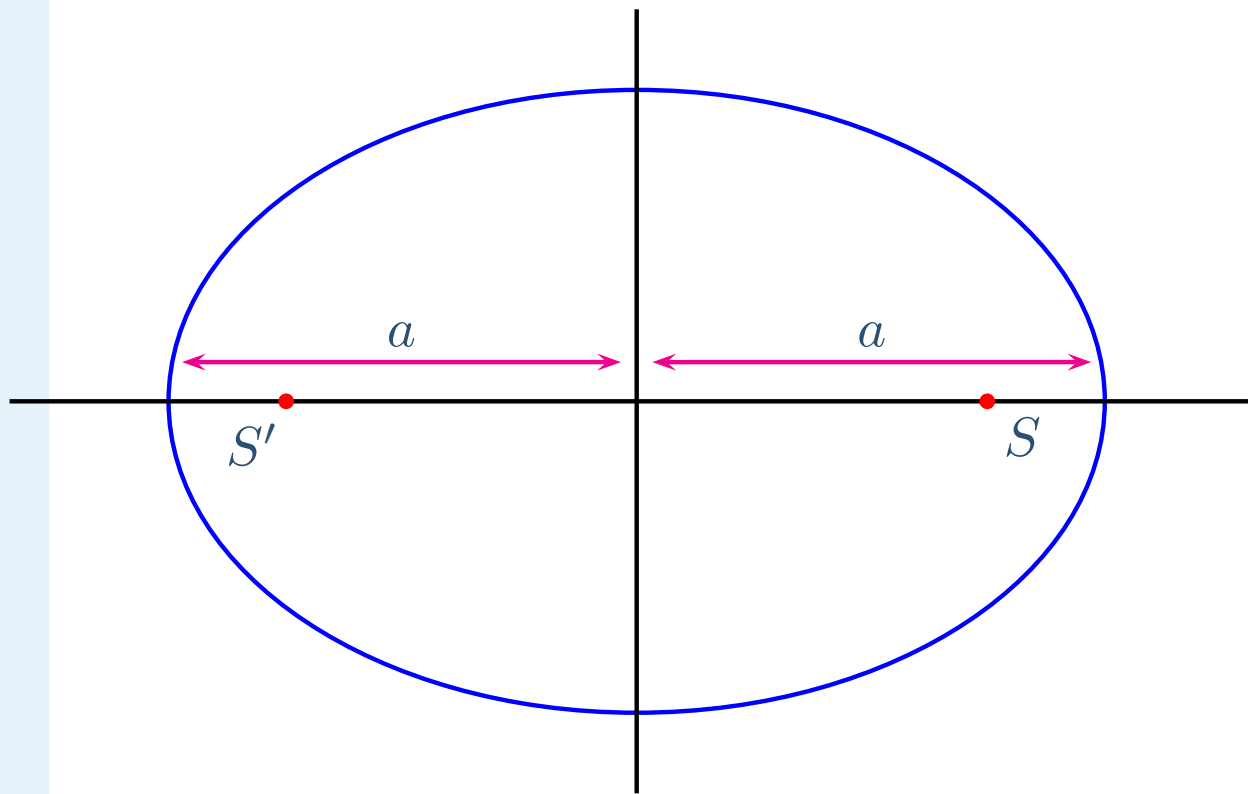
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Ellipses - The Algebraic Approach

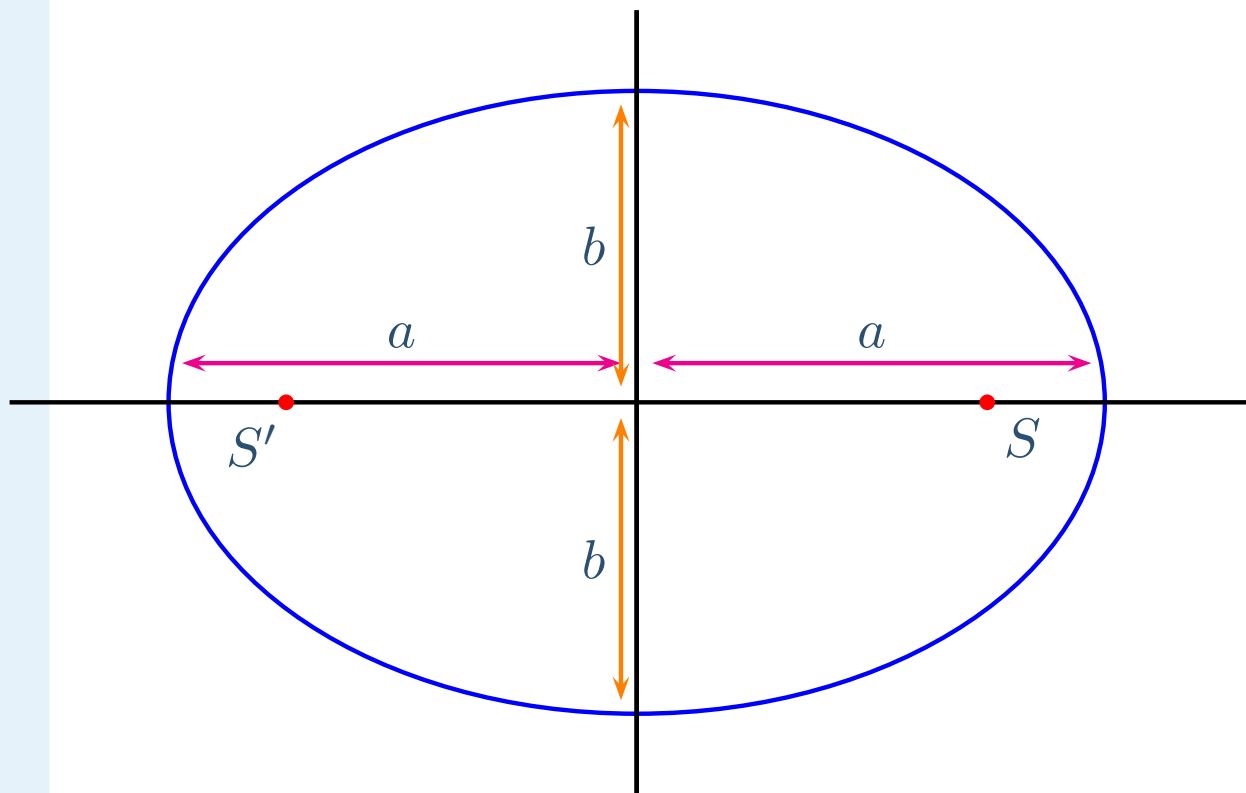
Ellipse - ovals.



a : semi-major axis

Ellipses - The Algebraic Approach

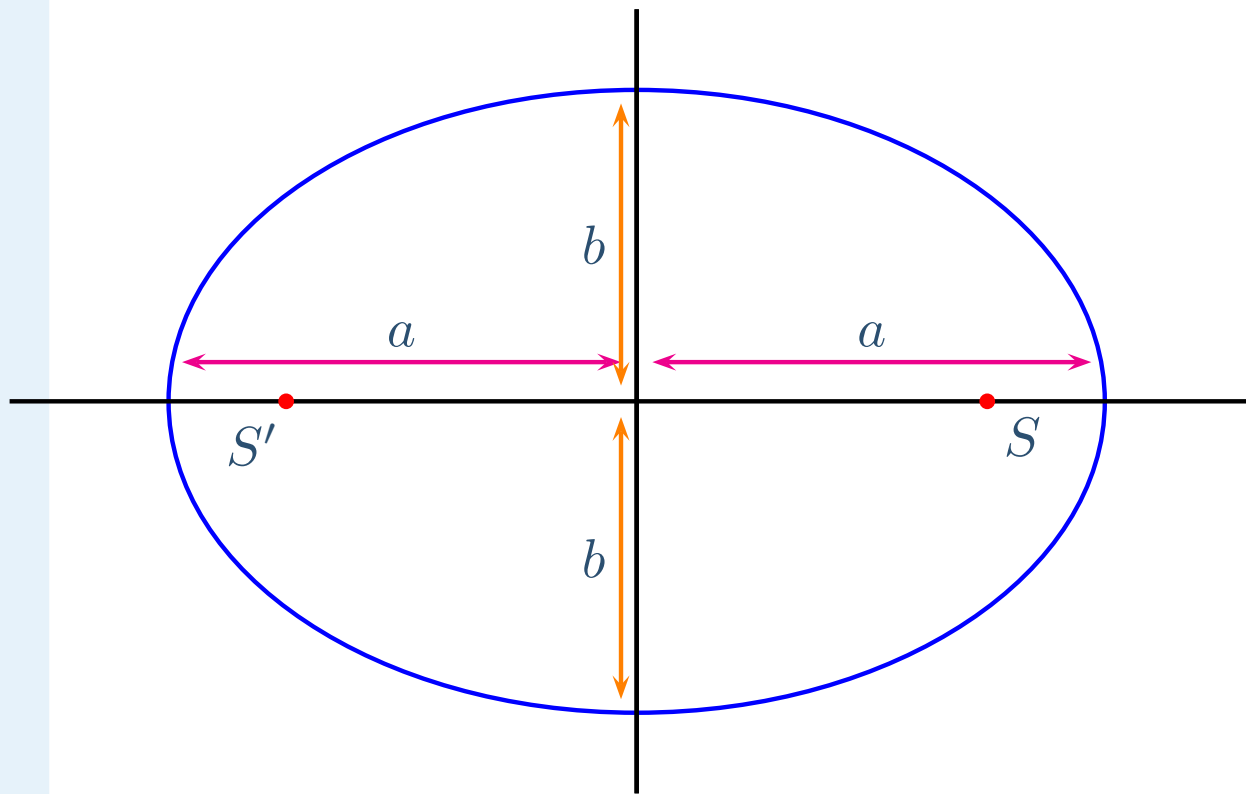
Ellipse - ovals.



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Ellipses - The Algebraic Approach

Ellipse - ovals.

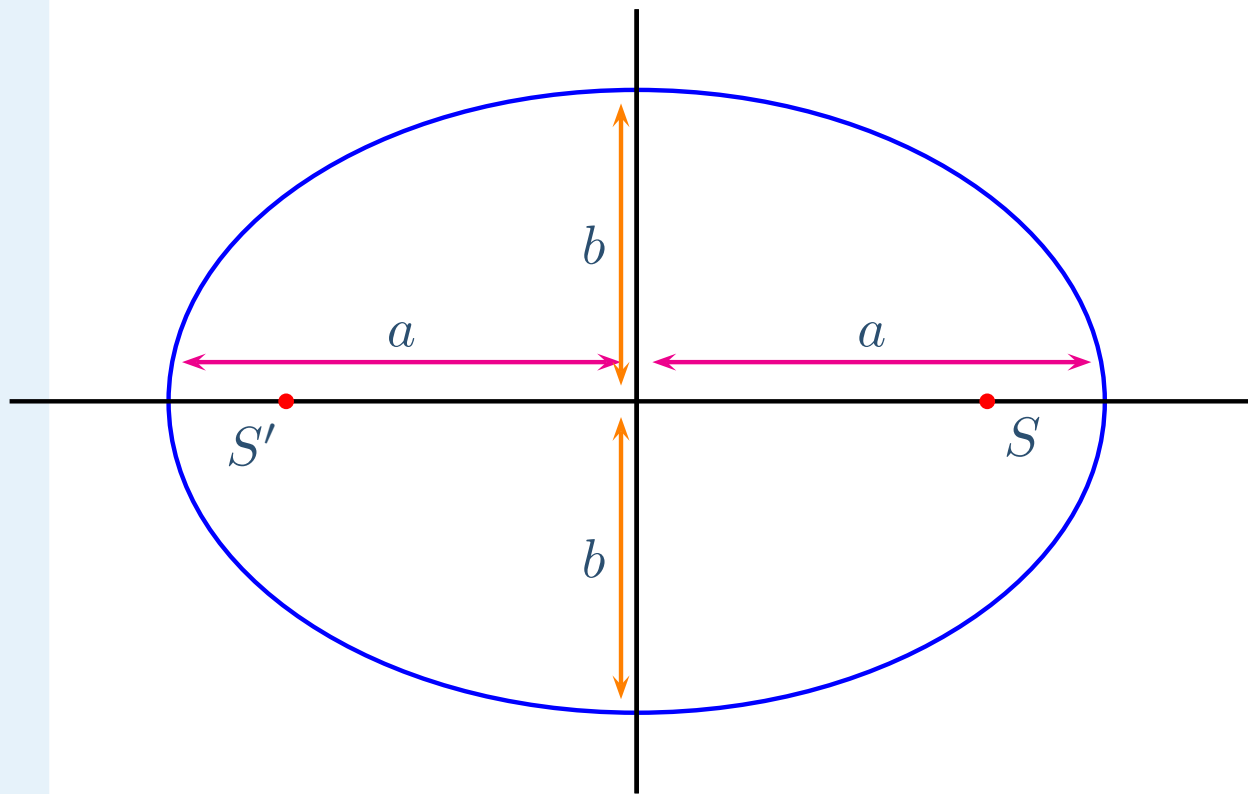


a : semi-major axis

b : semi-minor axis

Ellipses - The Algebraic Approach

Ellipse - ovals.



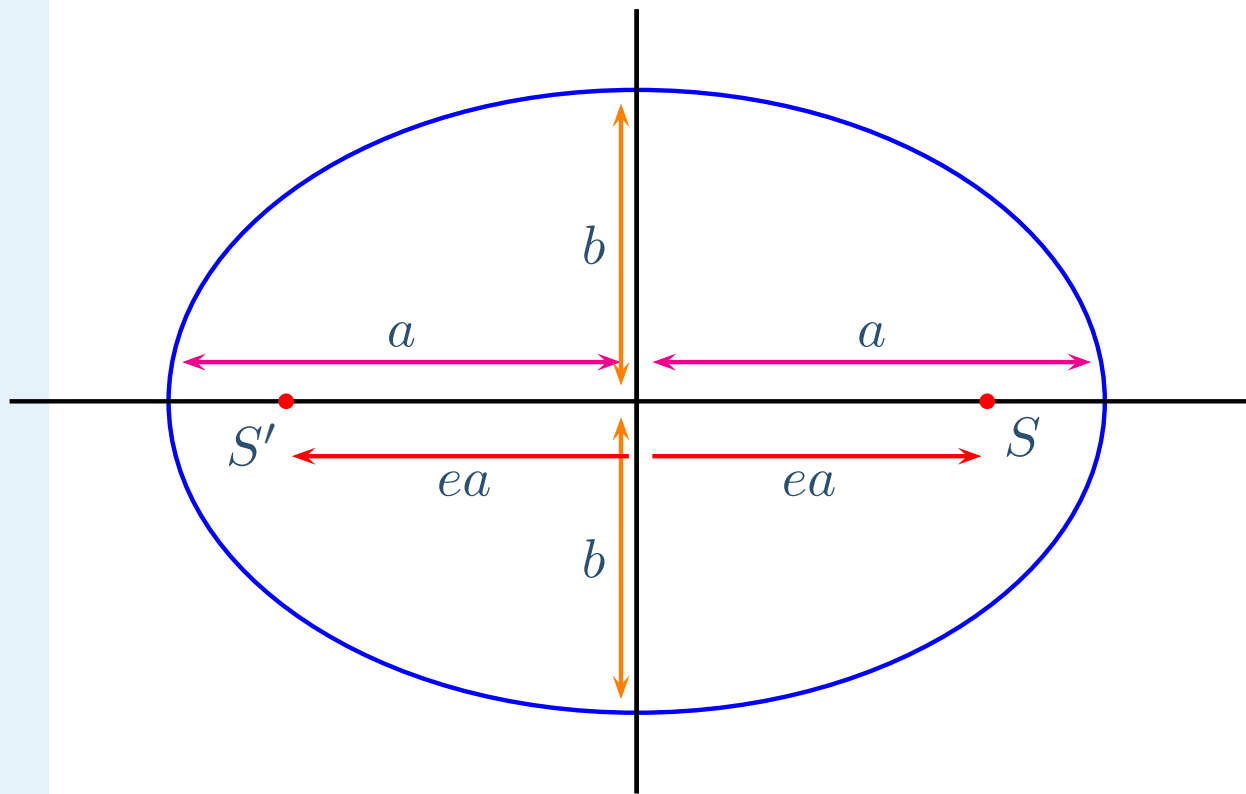
a : semi-major axis

b : semi-minor axis

$$\left(\frac{x}{a}\right)^2 + \left(\frac{y}{b}\right)^2 = 1$$

Ellipses - The Algebraic Approach

Ellipse - ovals.



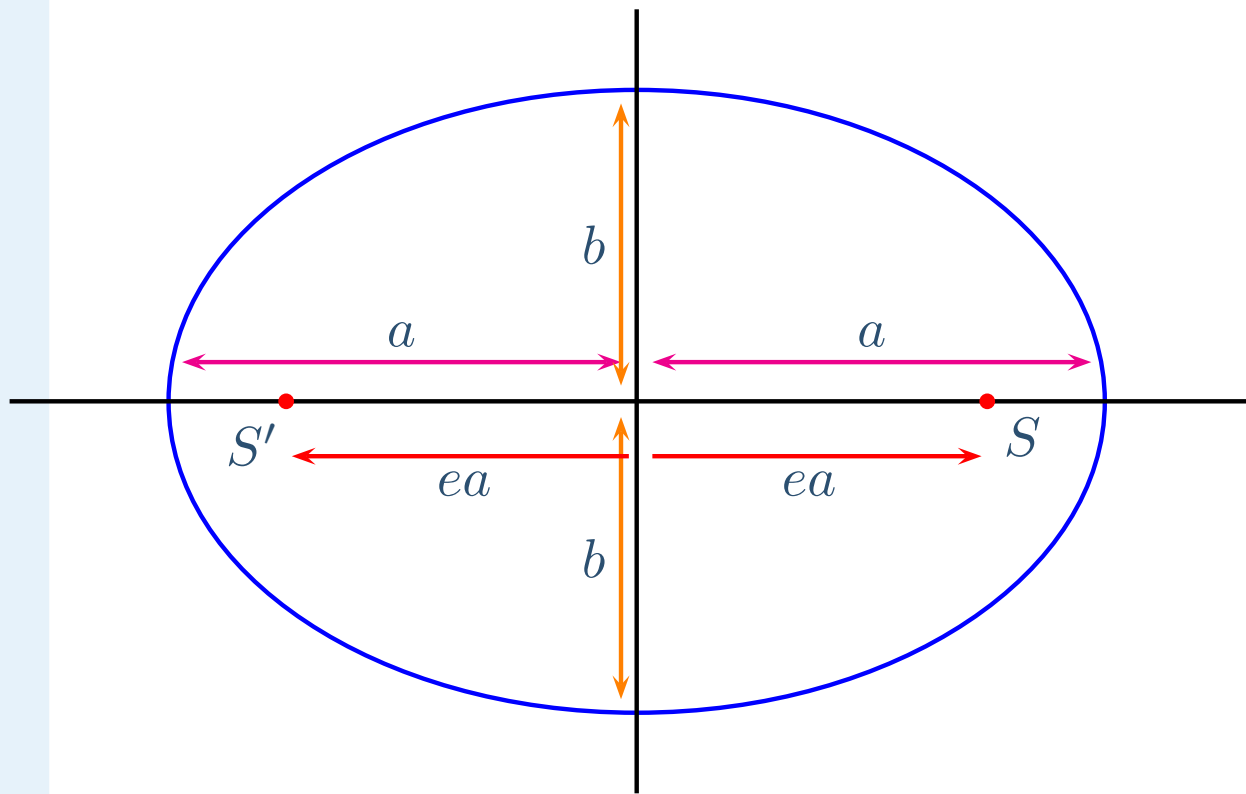
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Ellipse - ovals.



a : semi-major axis

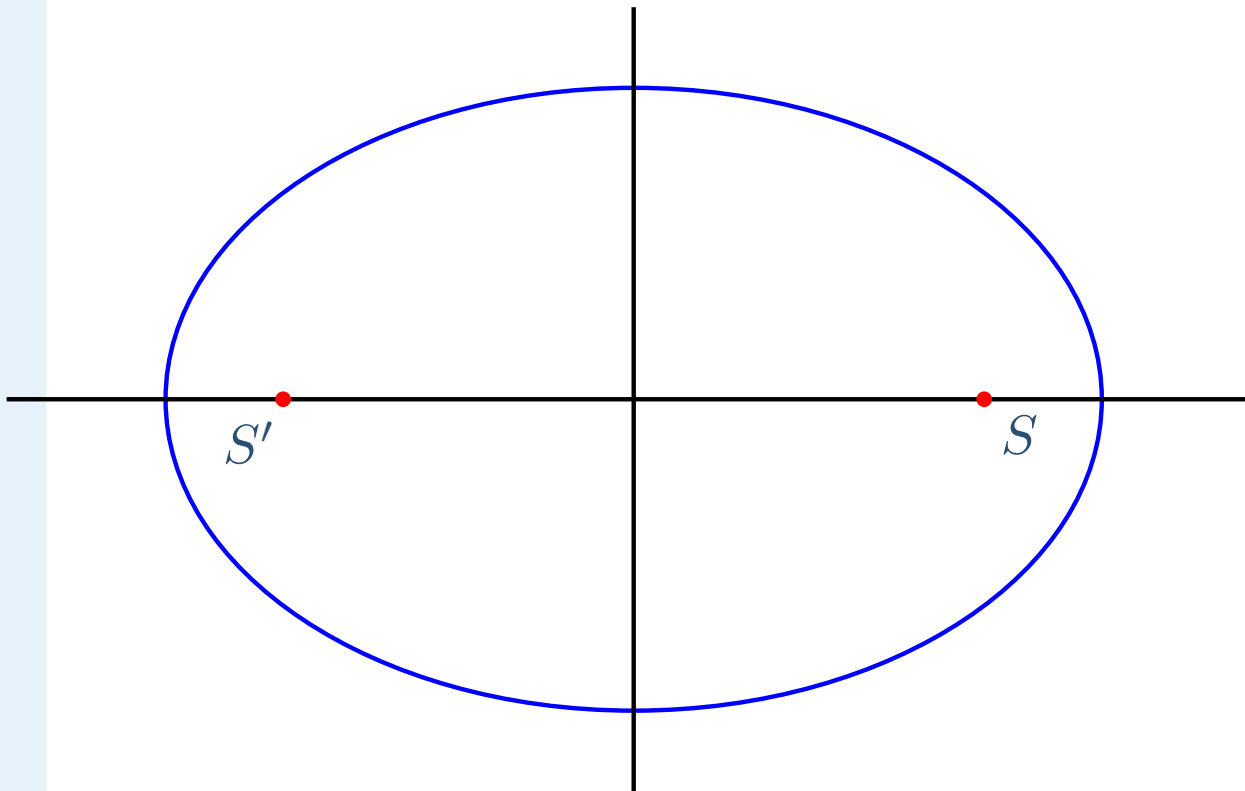
b : semi-minor axis

$$\left(\frac{x}{a}\right)^2 + \left(\frac{y}{b}\right)^2 = 1$$

e : eccentricity

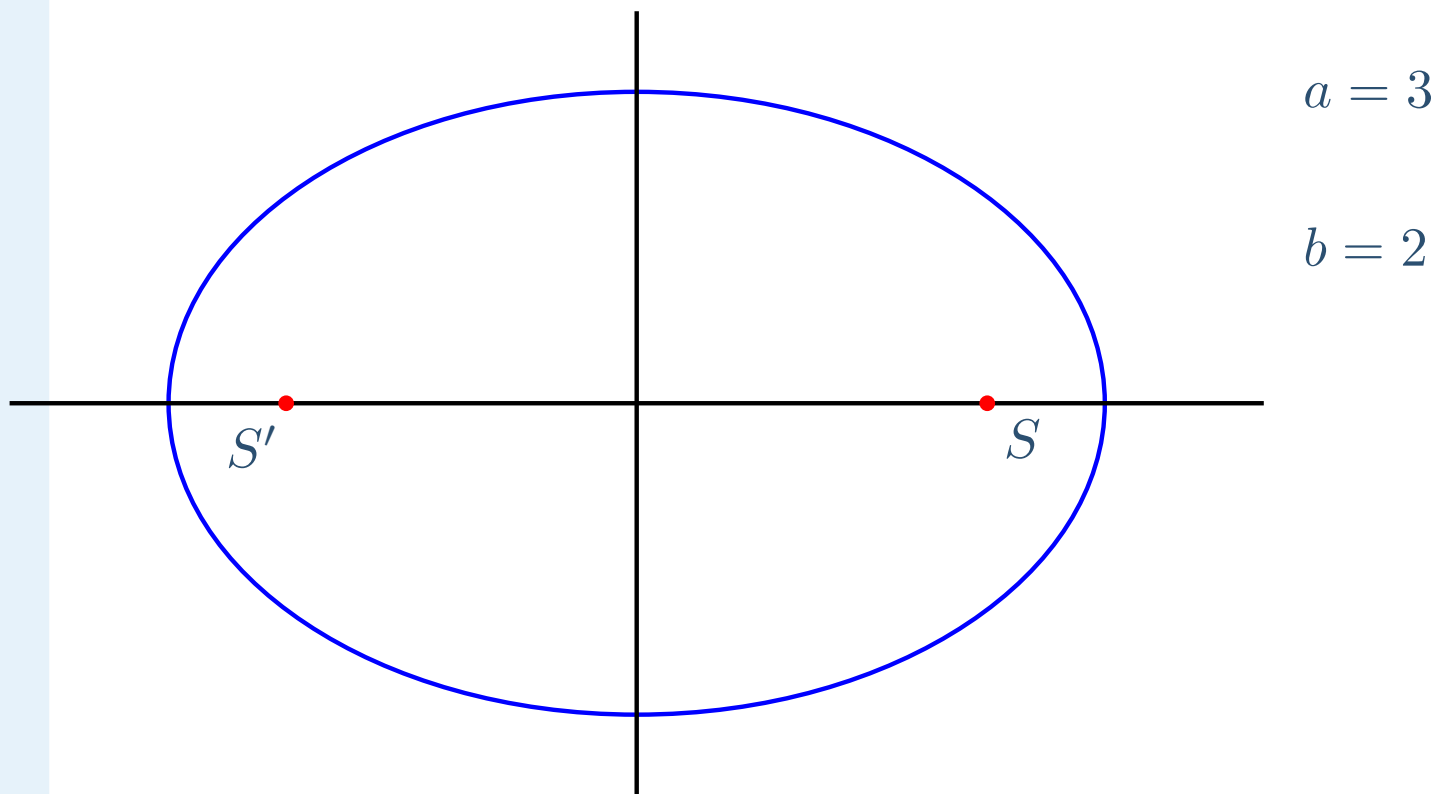
Eccentricity

The eccentricity gives the amount of “oval-ness” of the ellipse.



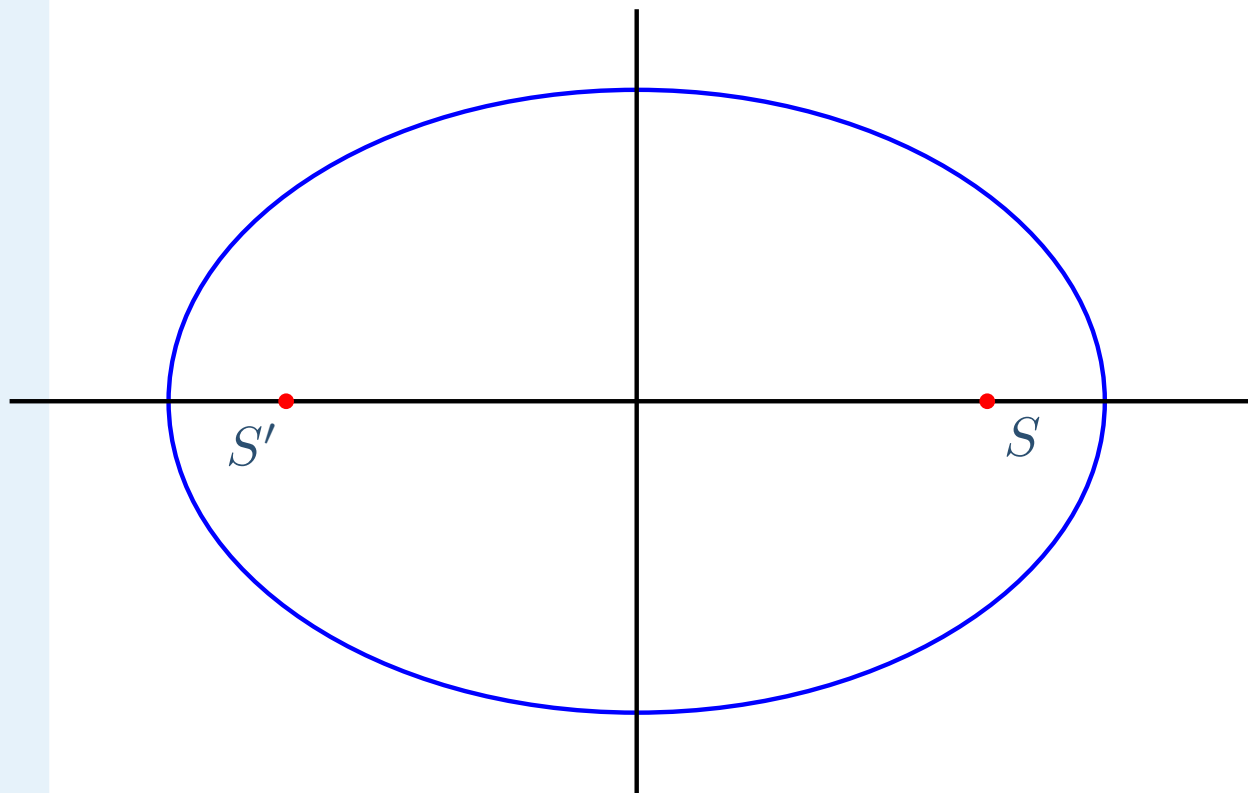
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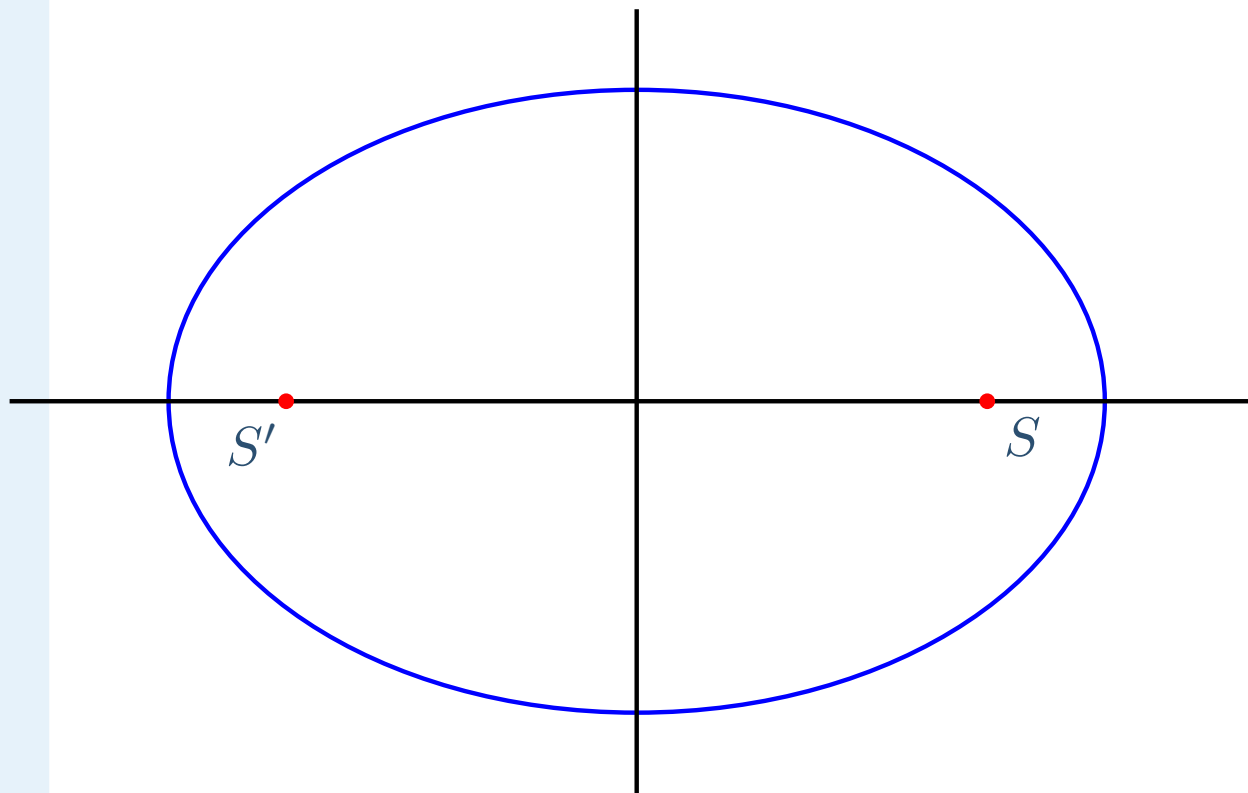
$$a = 3$$

$$b = 2$$

$$e = \sqrt{1 - \left(\frac{b}{a}\right)^2}$$

Eccentricity

The eccentricity gives the amount of “oval-ness” of the ellipse.



$$a = 3$$

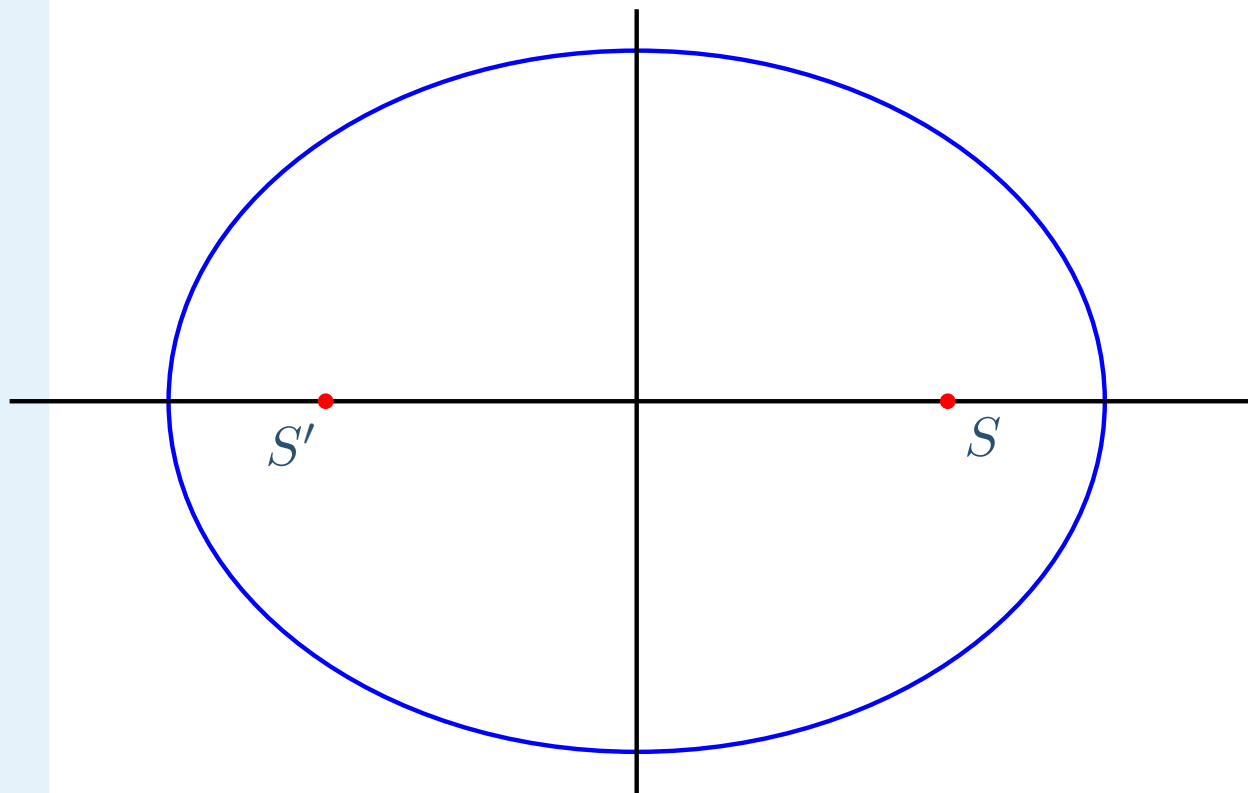
$$b = 2$$

$$e = \sqrt{1 - \left(\frac{b}{a}\right)^2}$$

$$e = 0.745$$

Eccentricity

The eccentricity gives the amount of “oval-ness” of the ellipse.



$$a = 3$$

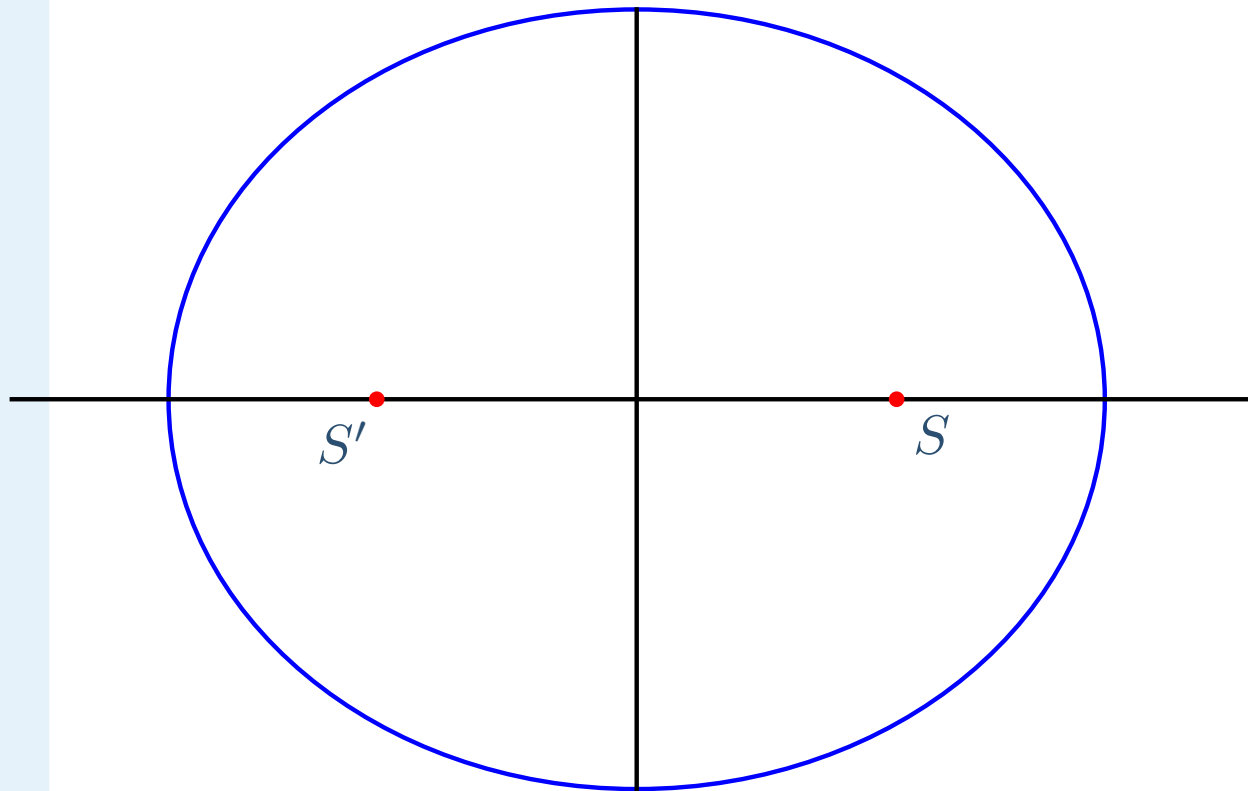
$$b = 2.25$$

$$e = \sqrt{1 - \left(\frac{b}{a}\right)^2}$$

$$e = 0.661$$

Eccentricity

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$$a = 3$$

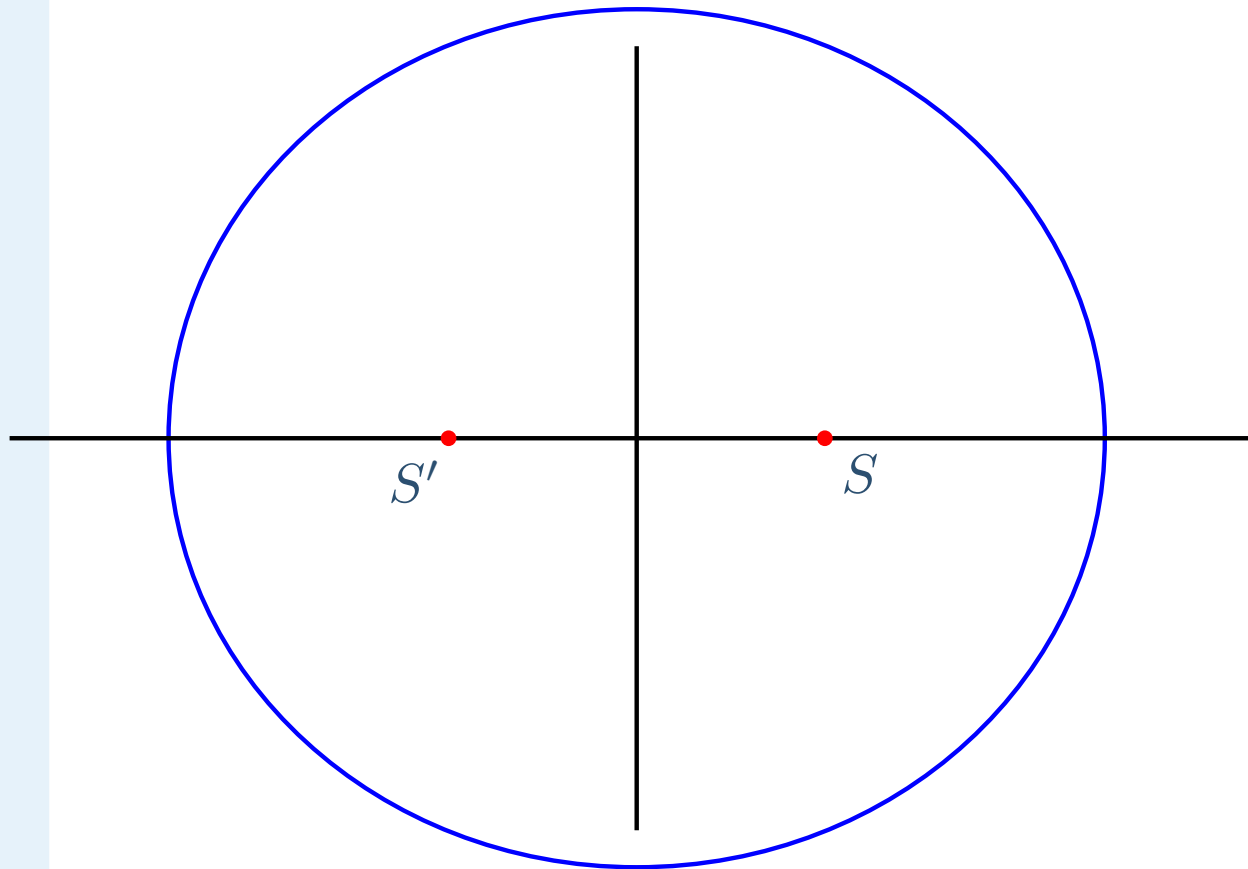
$$b = 2.5$$

$$e = \sqrt{1 - \left(\frac{b}{a}\right)^2}$$

$$e = 0.553$$

Eccentricity

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$$a = 3$$

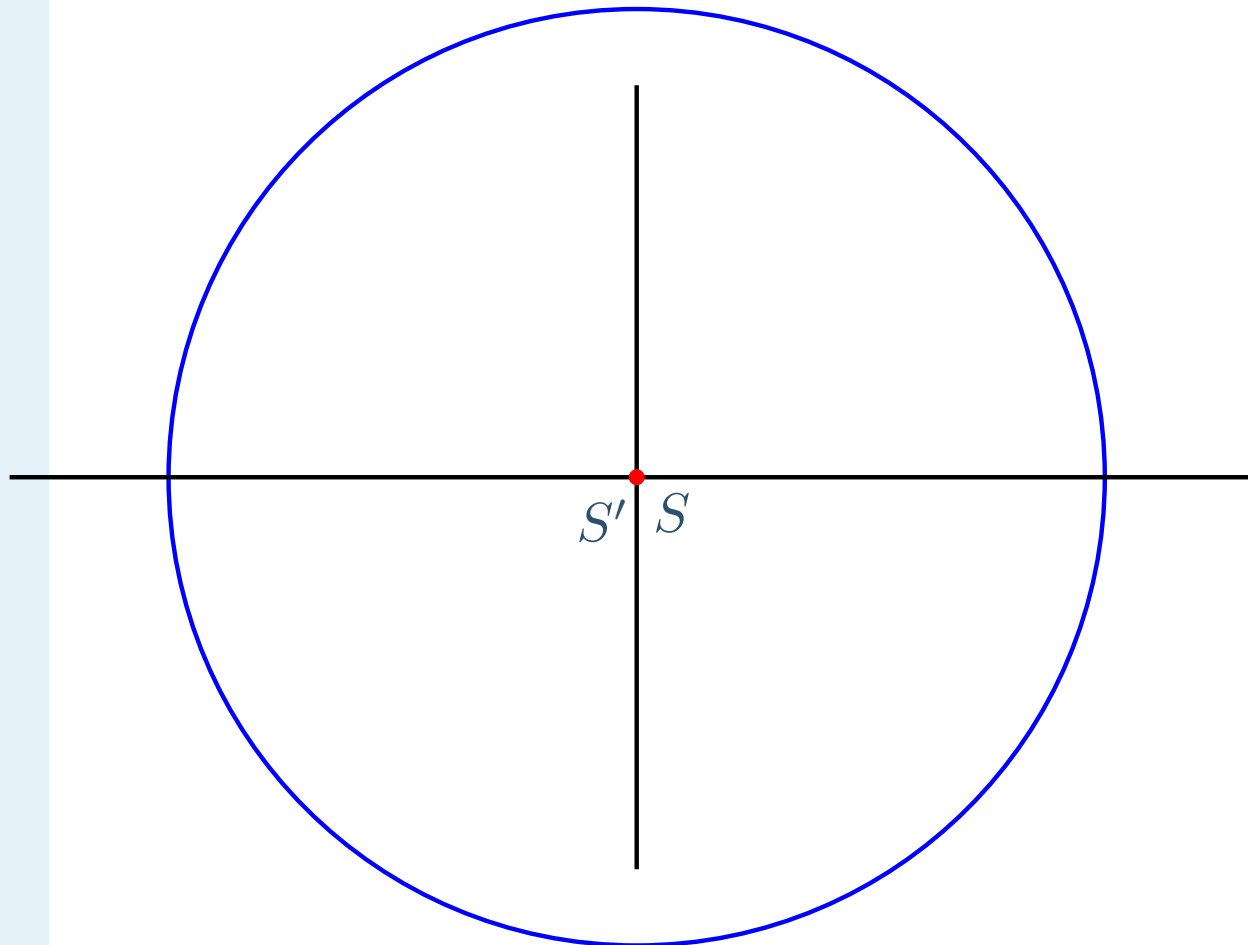
$$b = 2.75$$

$$e = \sqrt{1 - \left(\frac{b}{a}\right)^2}$$

$$e = 0.400$$

Eccentricity

The eccentricity gives the amount of “oval-ness” of the ellipse.



$$a = 3$$

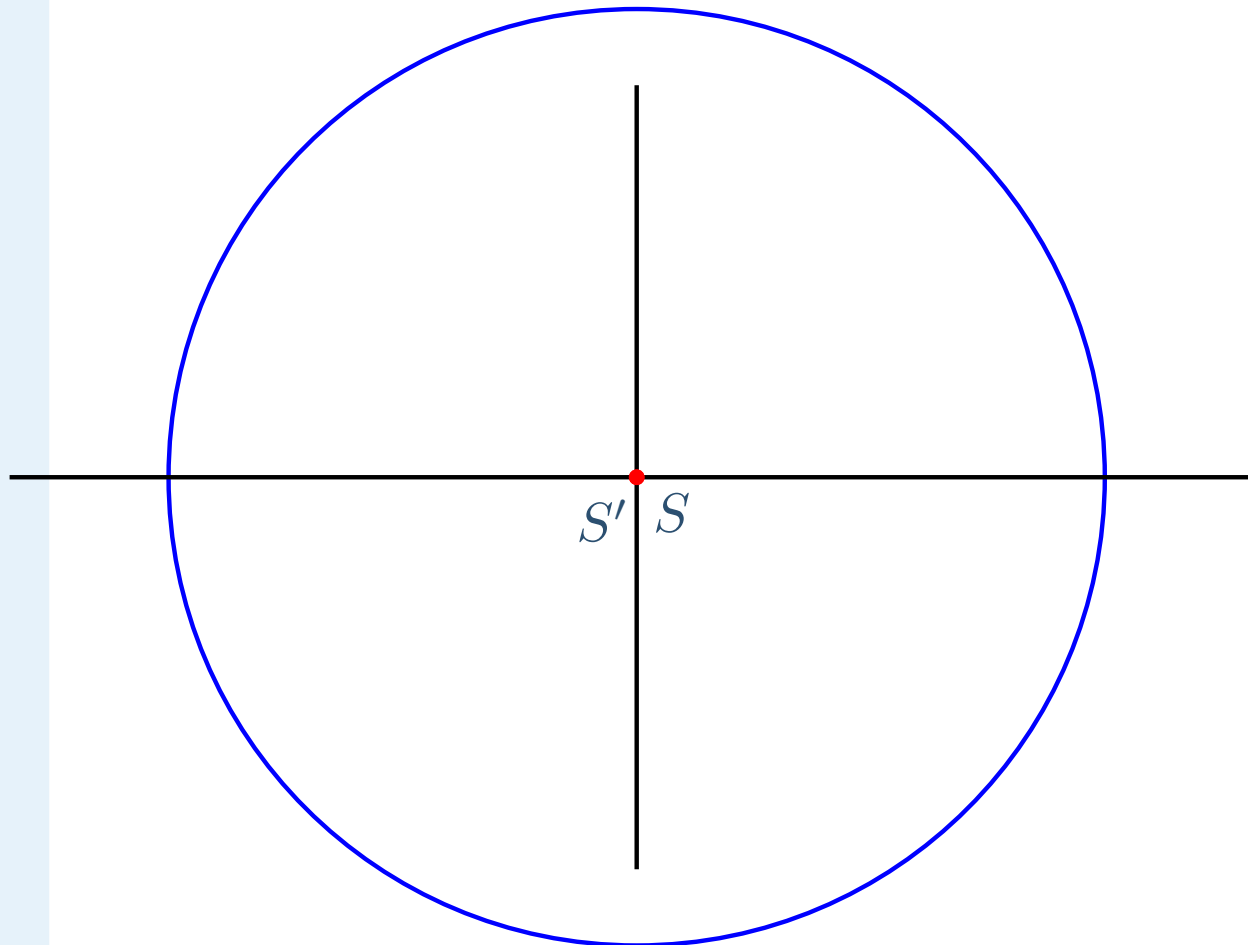
$$b = 3$$

$$e = \sqrt{1 - \left(\frac{b}{a}\right)^2}$$

$$e = 0$$

Eccentricity

The eccentricity gives the amount of “oval-ness” of the ellipse.



$$a = 3$$

$$b = 3$$

$$e = \sqrt{1 - \left(\frac{b}{a}\right)^2}$$

$$e = 0$$



Circle

Kepler's First Law

Kepler's Laws:

- 1: Each planet's orbit traces out the shape of an ellipse with the sun located at one focus.

Kepler's First Law

Kepler's Laws:

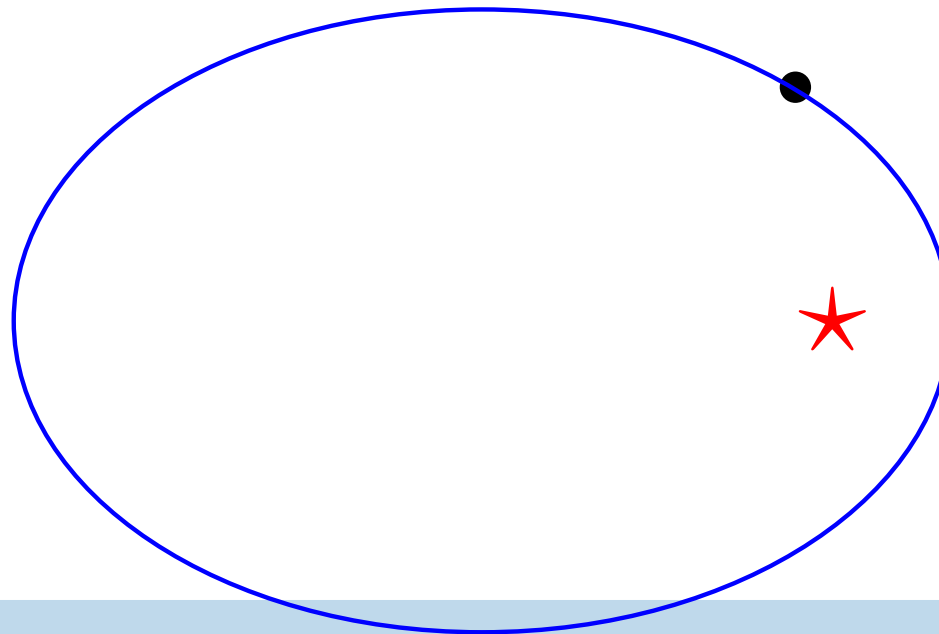
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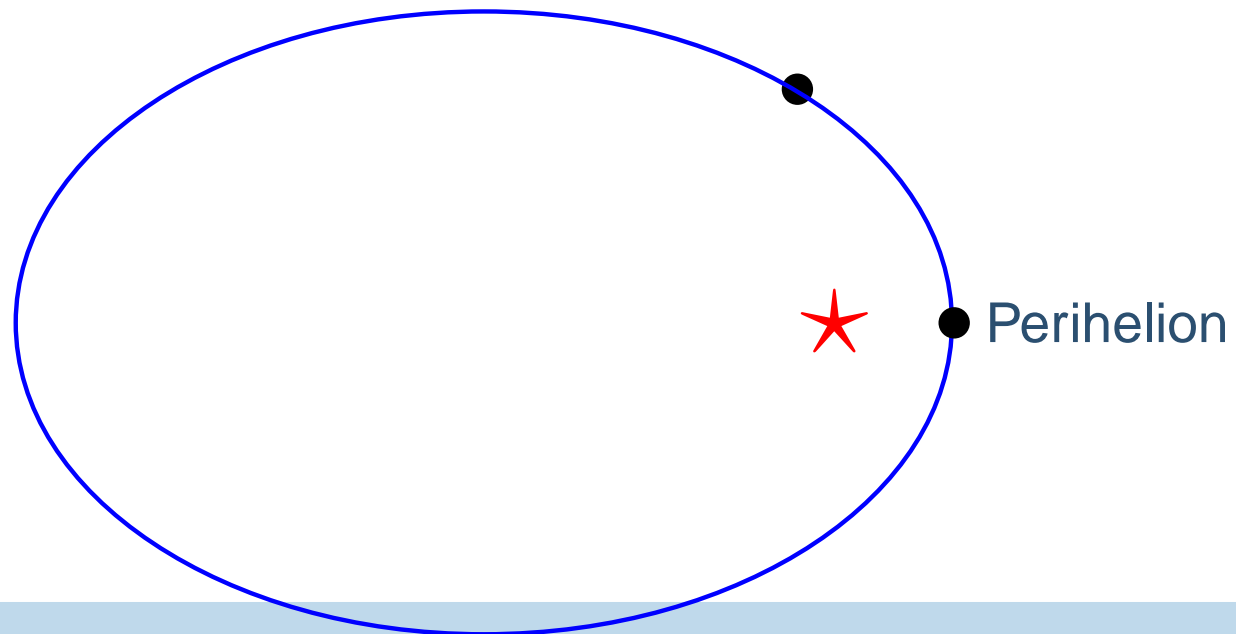
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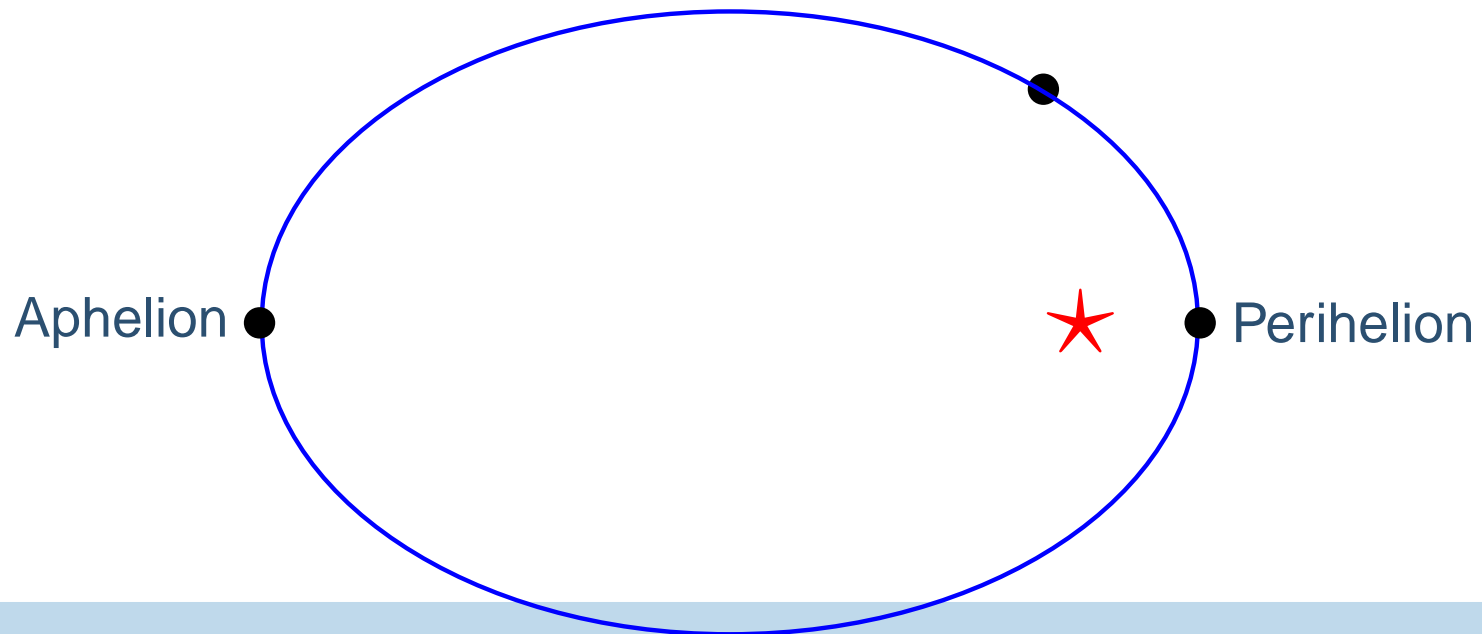
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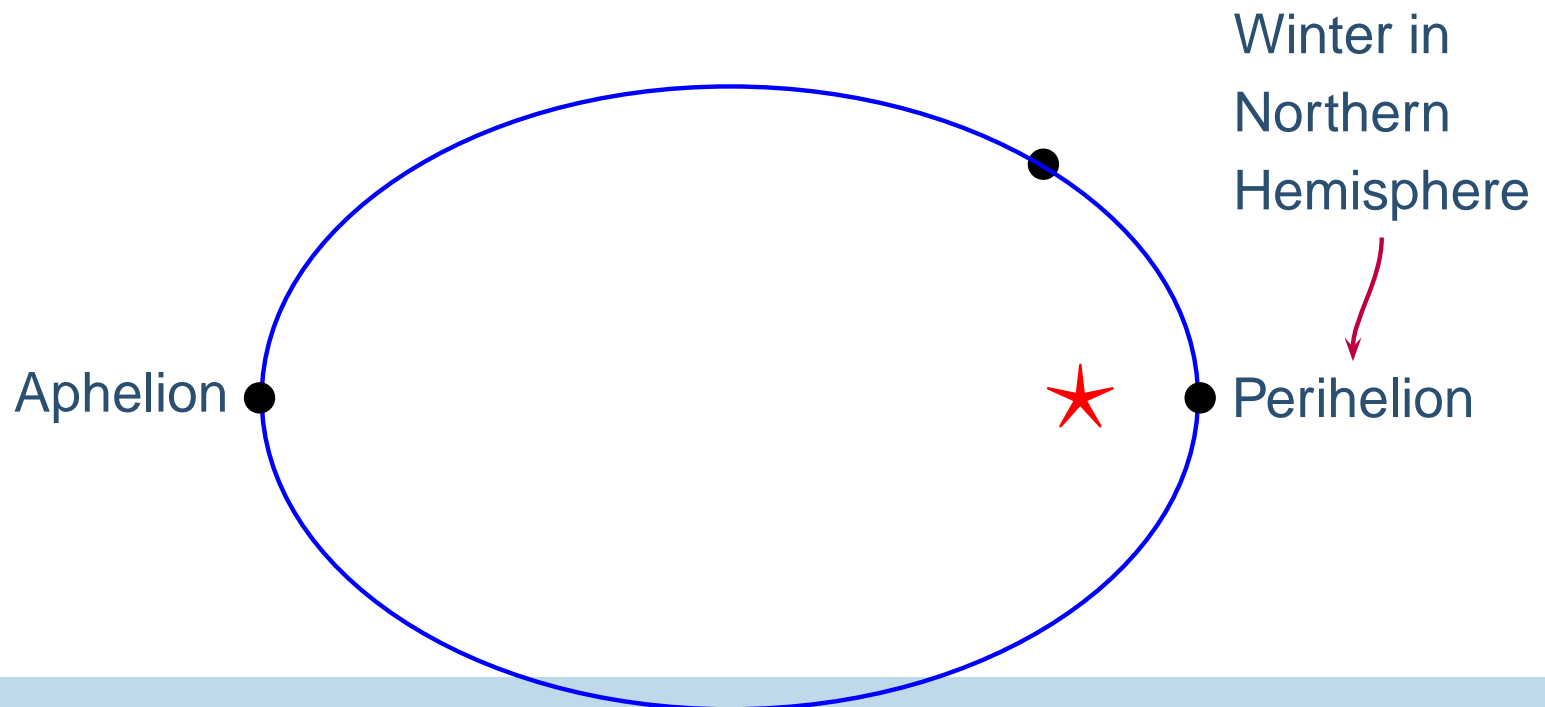
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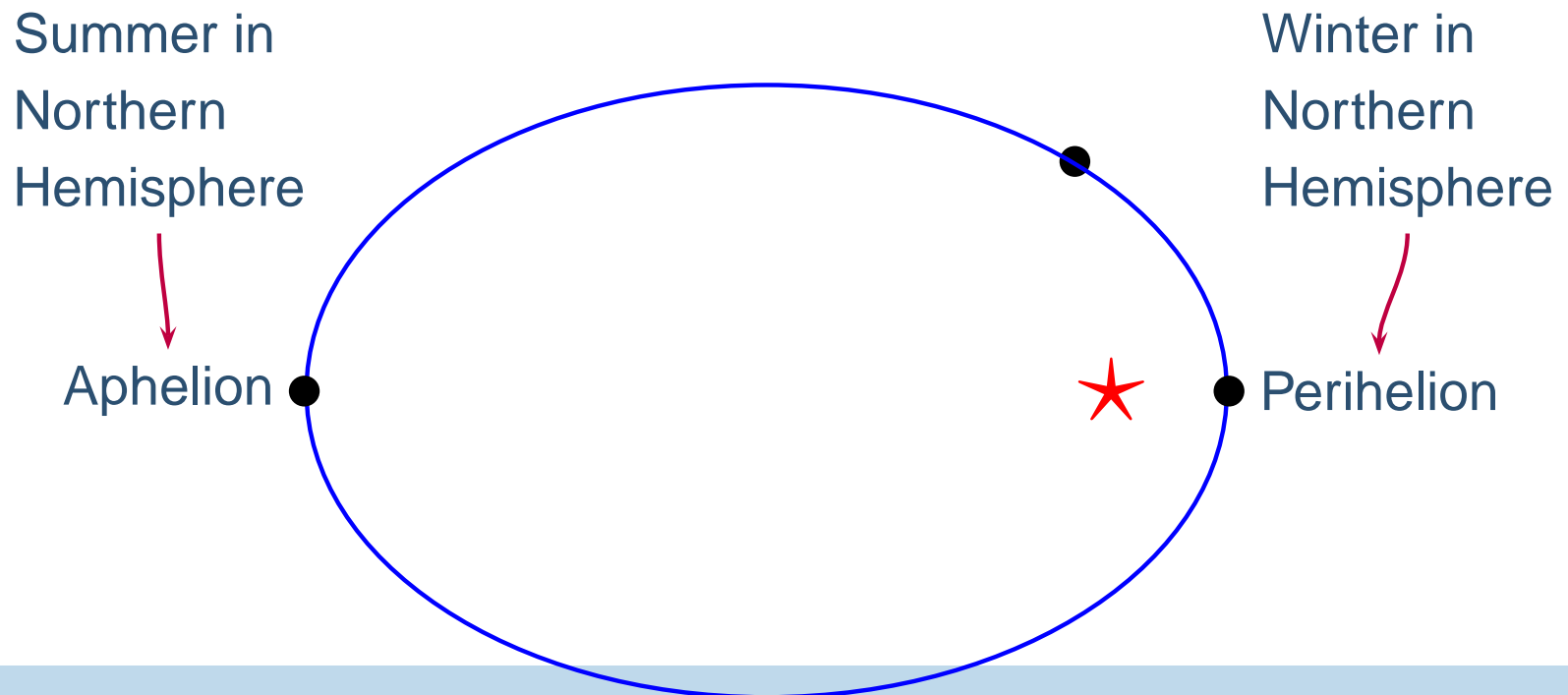
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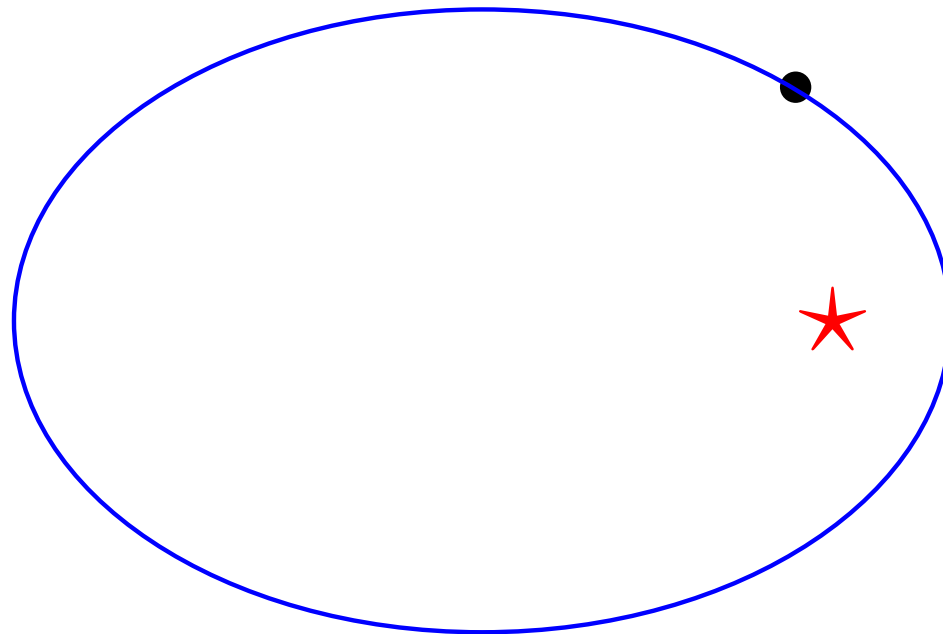
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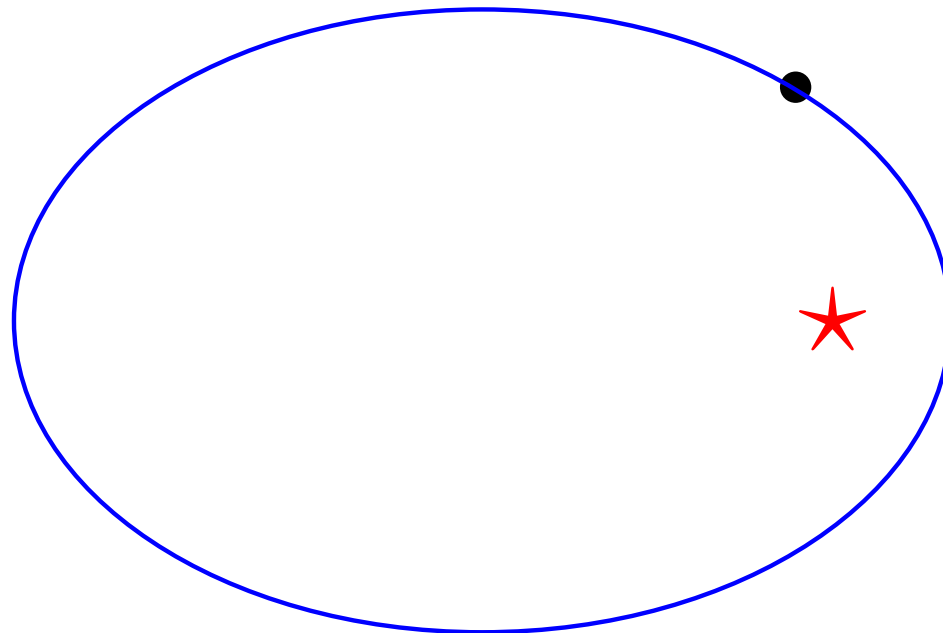
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Kepler's First Law II

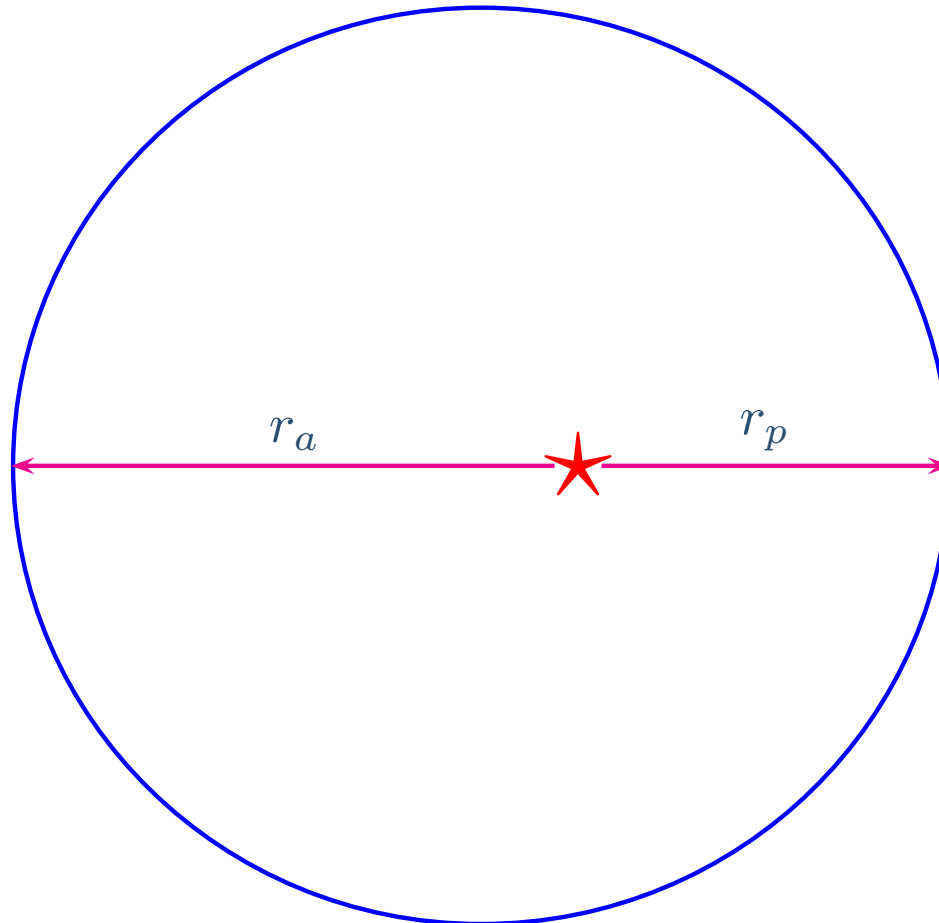


Kepler's First Law II



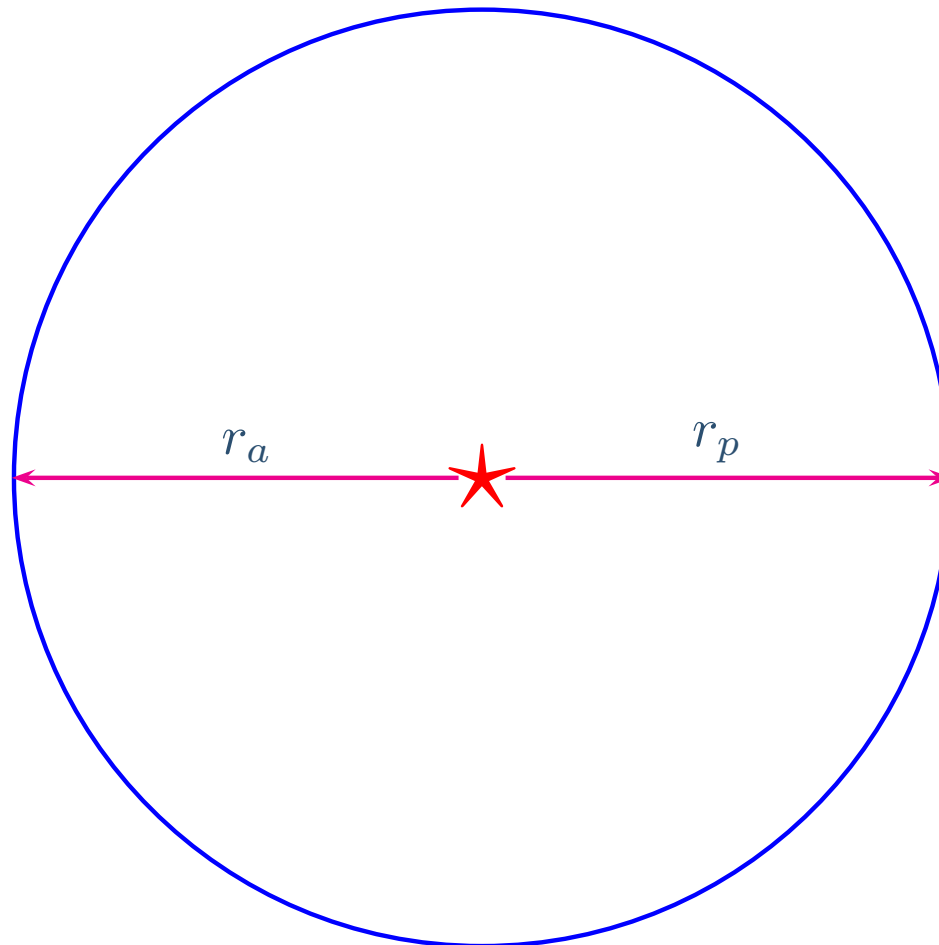
Planet	e

Kepler's First Law II



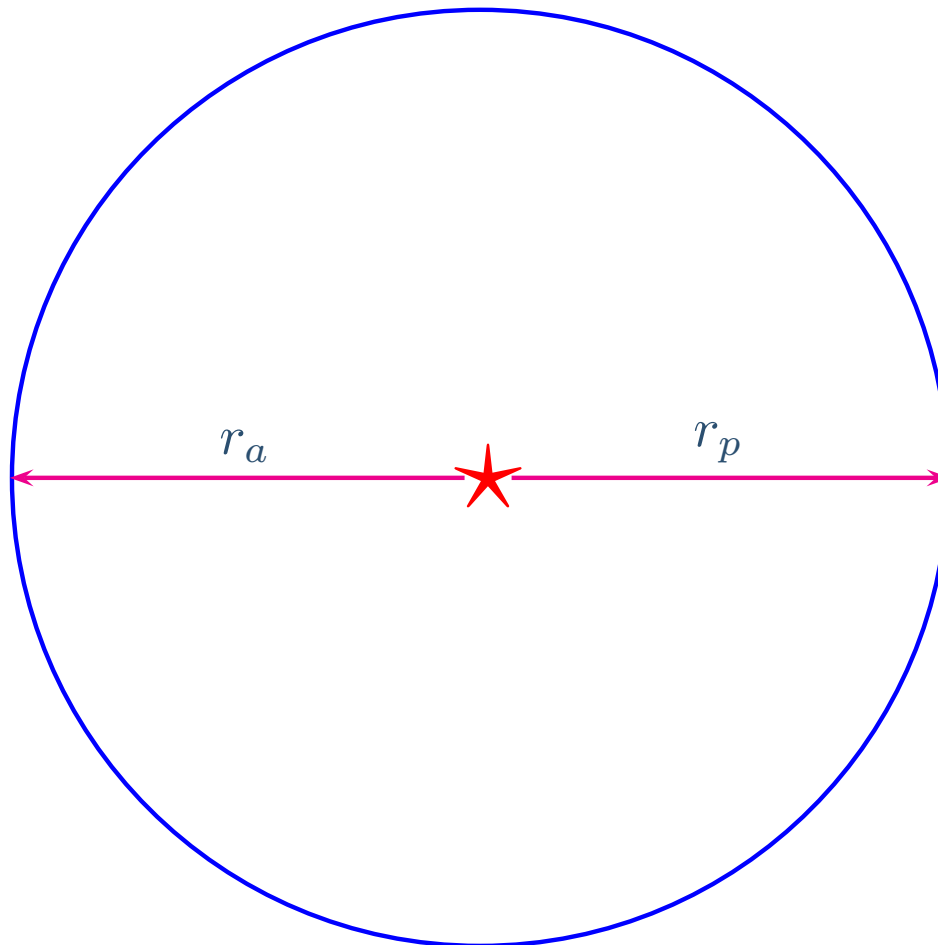
Planet	e
Mercury	0.206

Kepler's First Law II



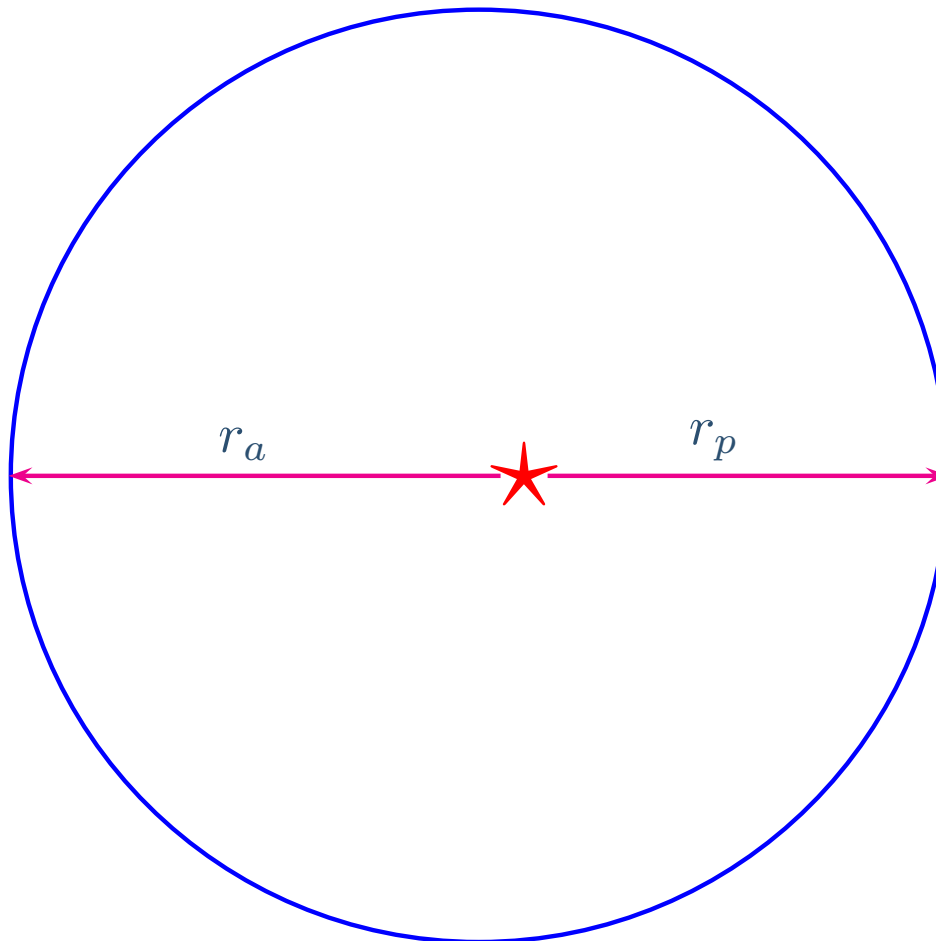
Planet	e
Mercury	0.206
Venus	0.007

Kepler's First Law II



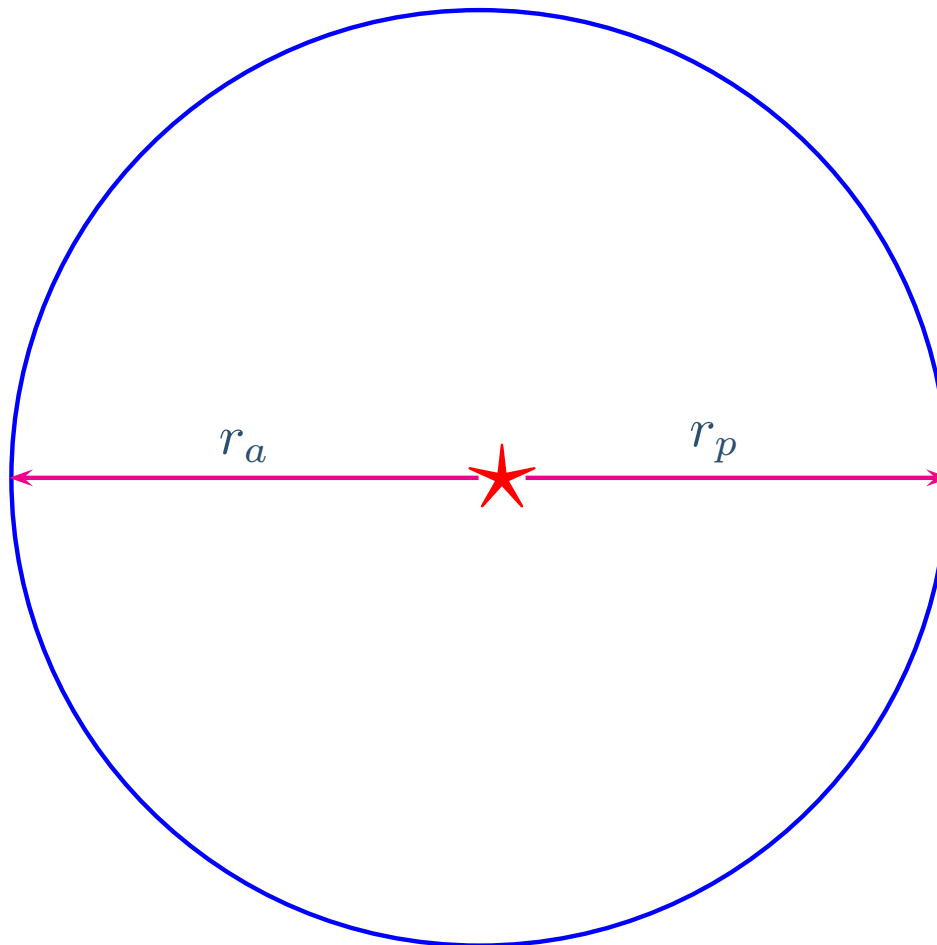
Planet	e
Mercury	0.206
Venus	0.007
Earth	0.017

Kepler's First Law II



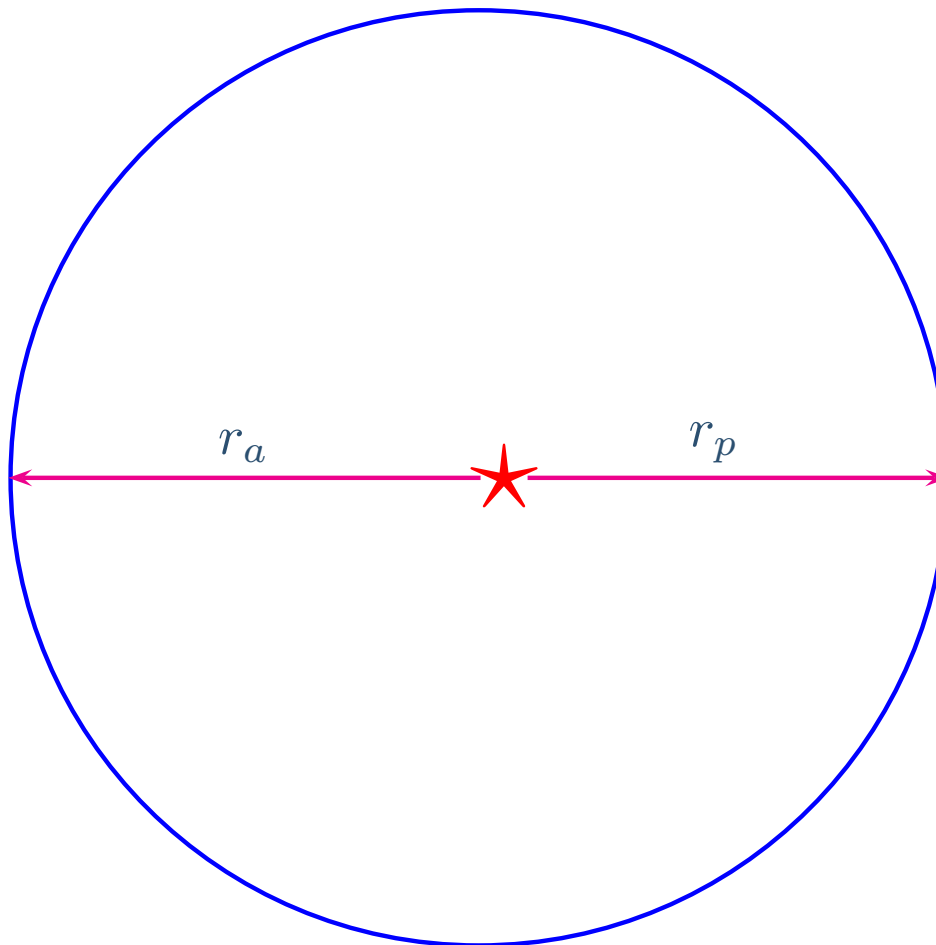
Planet	e
Mercury	0.206
Venus	0.007
Earth	0.017
Mars	0.093

Kepler's First Law II



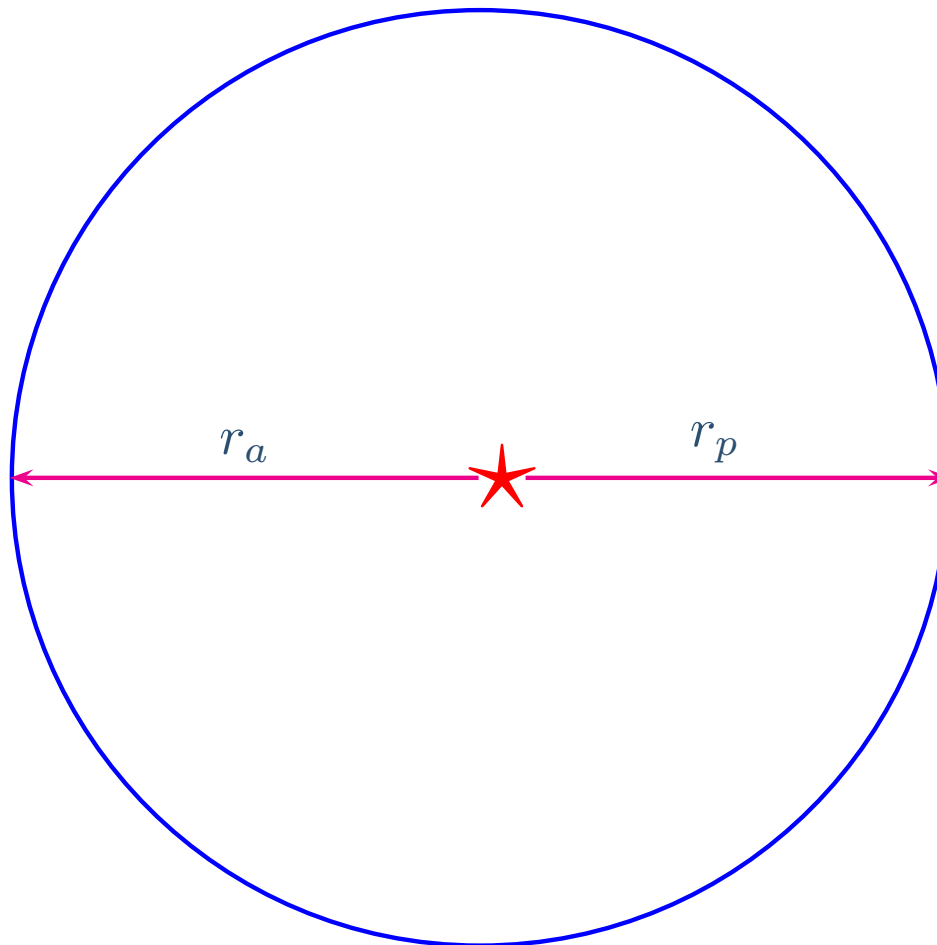
Planet	e
Mercury	0.206
Venus	0.007
Earth	0.017
Mars	0.093
Jupiter	0.048

Kepler's First Law II



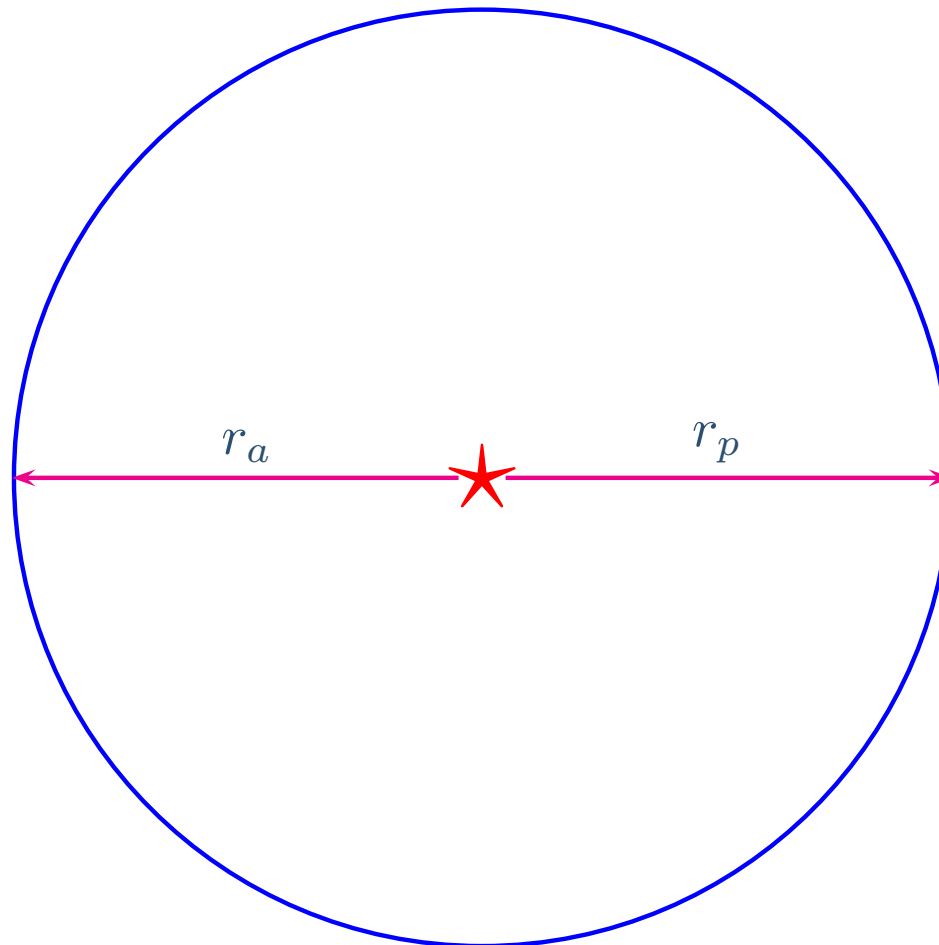
Planet	e
Mercury	0.206
Venus	0.007
Earth	0.017
Mars	0.093
Jupiter	0.048
Saturn	0.054

Kepler's First Law II



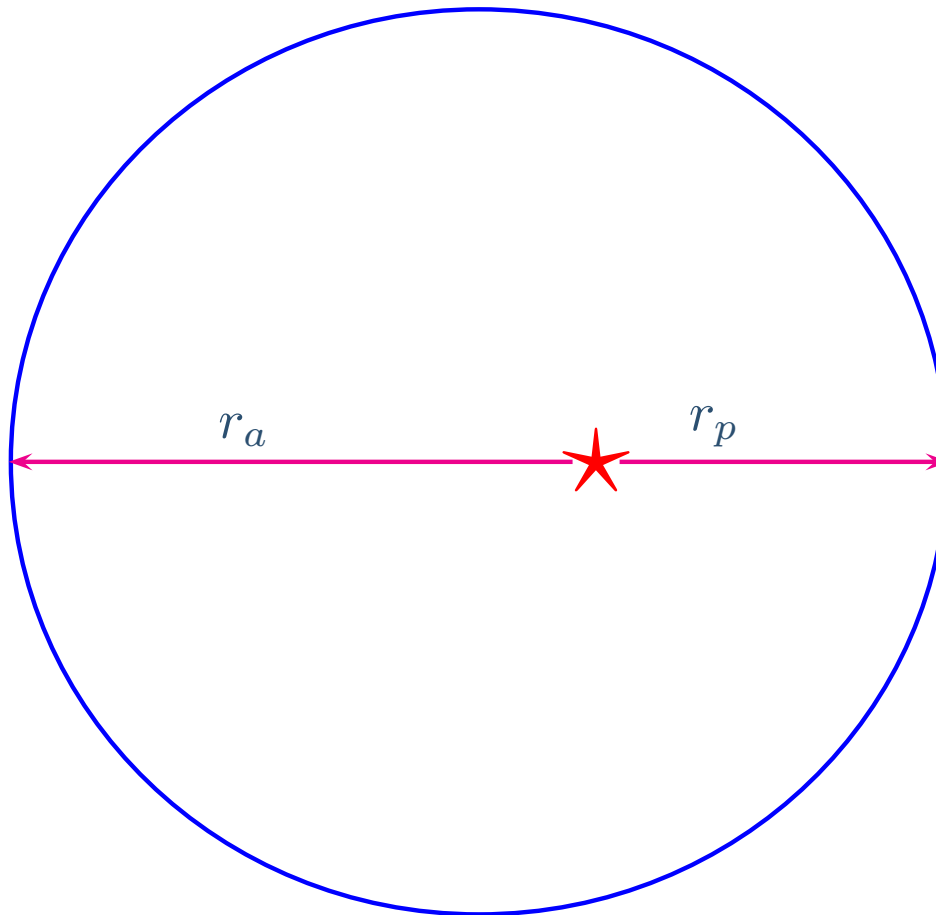
Planet	e
Mercury	0.206
Venus	0.007
Earth	0.017
Mars	0.093
Jupiter	0.048
Saturn	0.054
Uranus	0.047

Kepler's First Law II



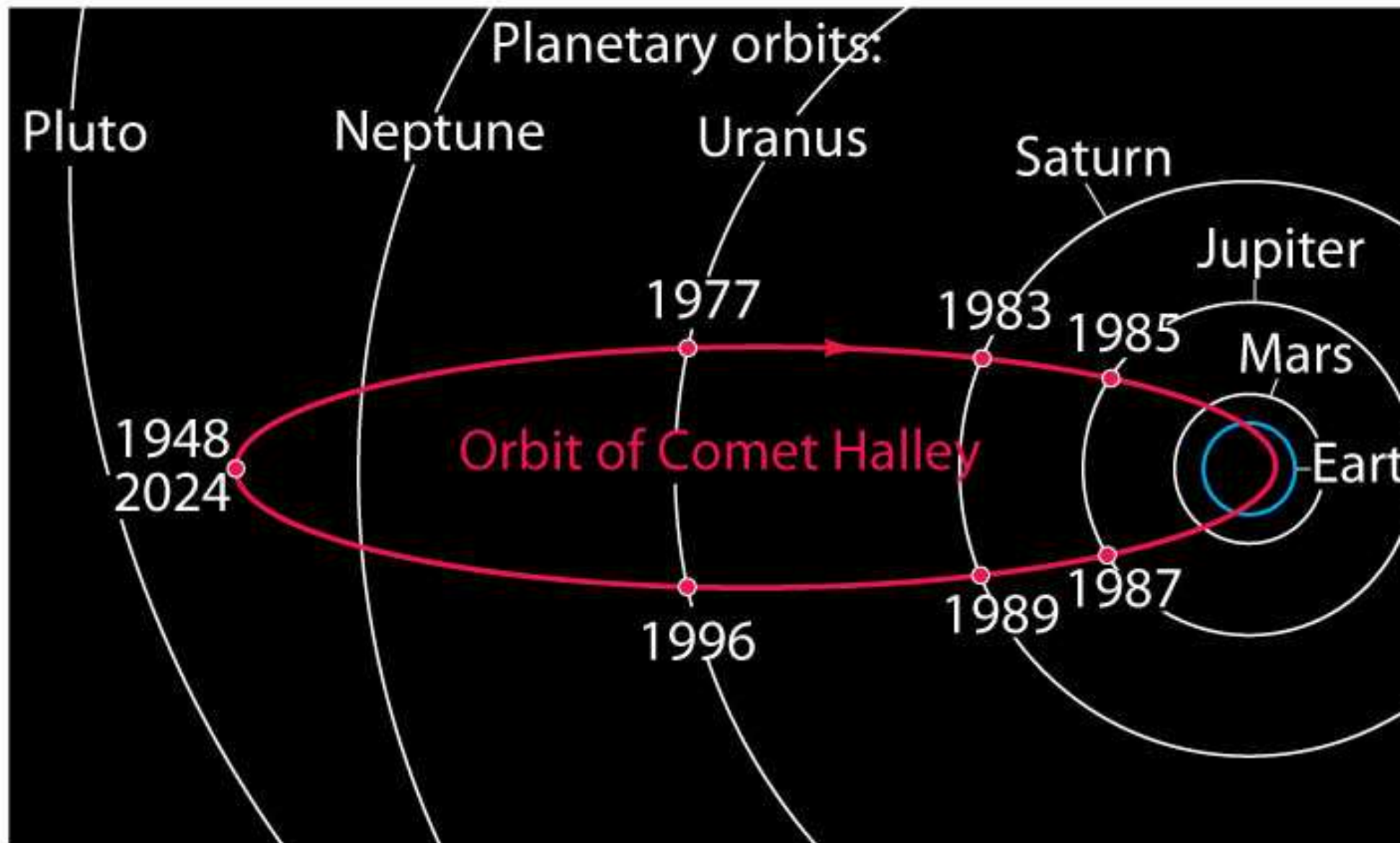
Planet	e
Mercury	0.206
Venus	0.007
Earth	0.017
Mars	0.093
Jupiter	0.048
Saturn	0.054
Uranus	0.047
Neptune	0.009

Kepler's First Law II



Planet	e
Mercury	0.206
Venus	0.007
Earth	0.017
Mars	0.093
Jupiter	0.048
Saturn	0.054
Uranus	0.047
Neptune	0.009
Pluto	0.249

Kepler's First Law III



Kepler's Second Law

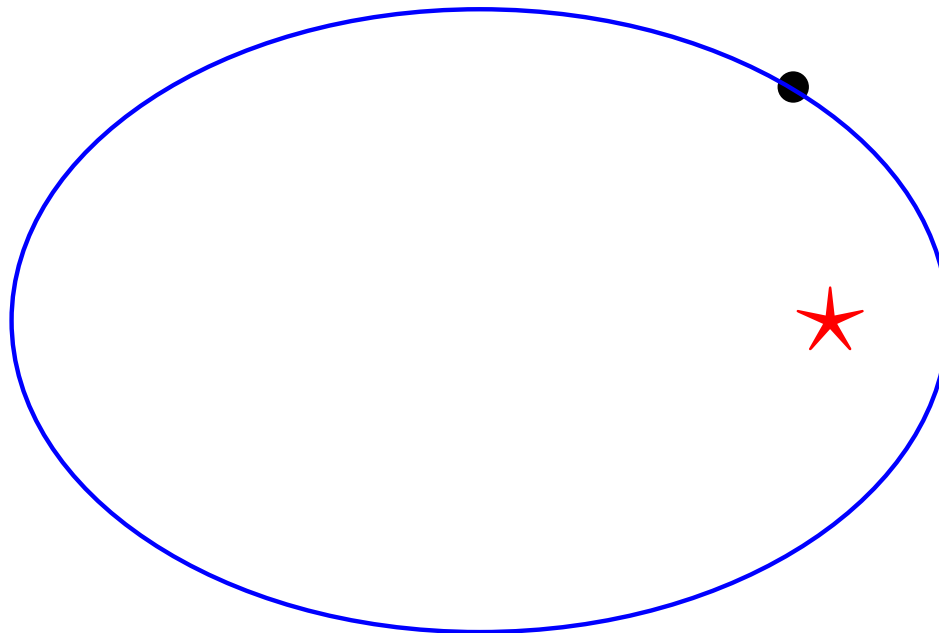
Kepler's Laws:

2: The imaginary line from the sun to a planet sweeps out equal areas in equal times.

Kepler's Second Law

Kepler's Laws:

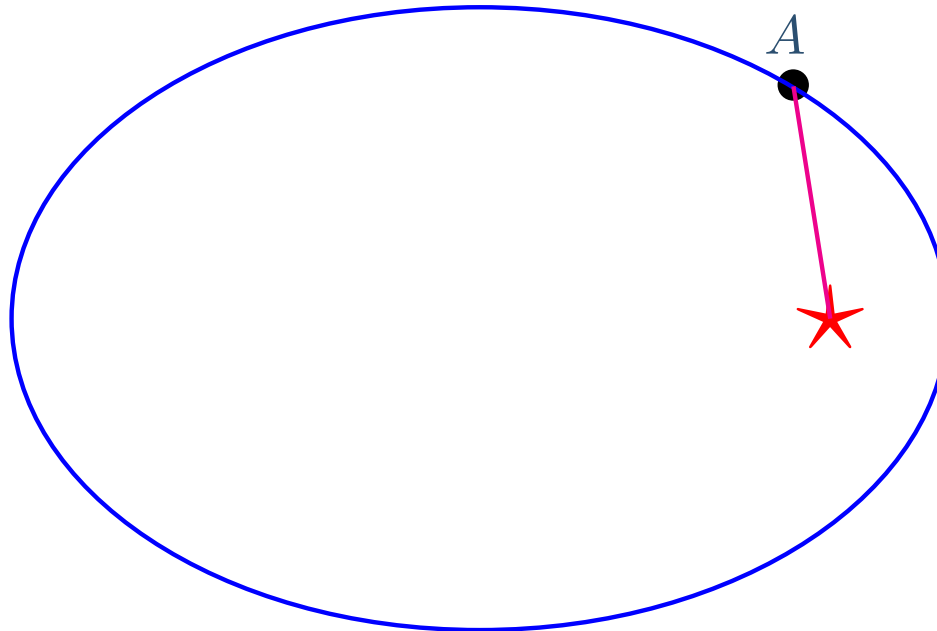
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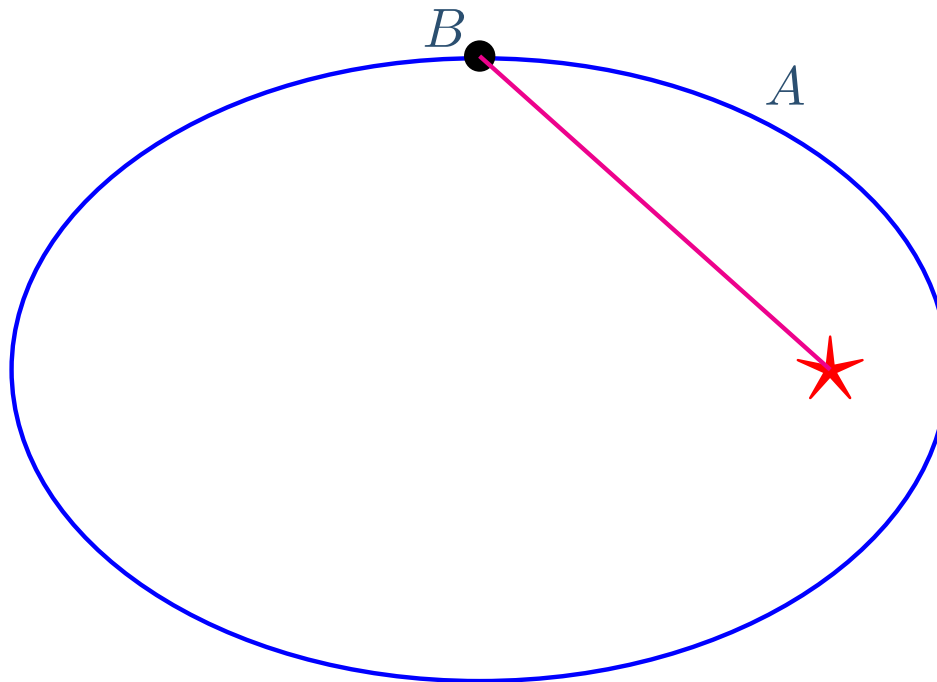
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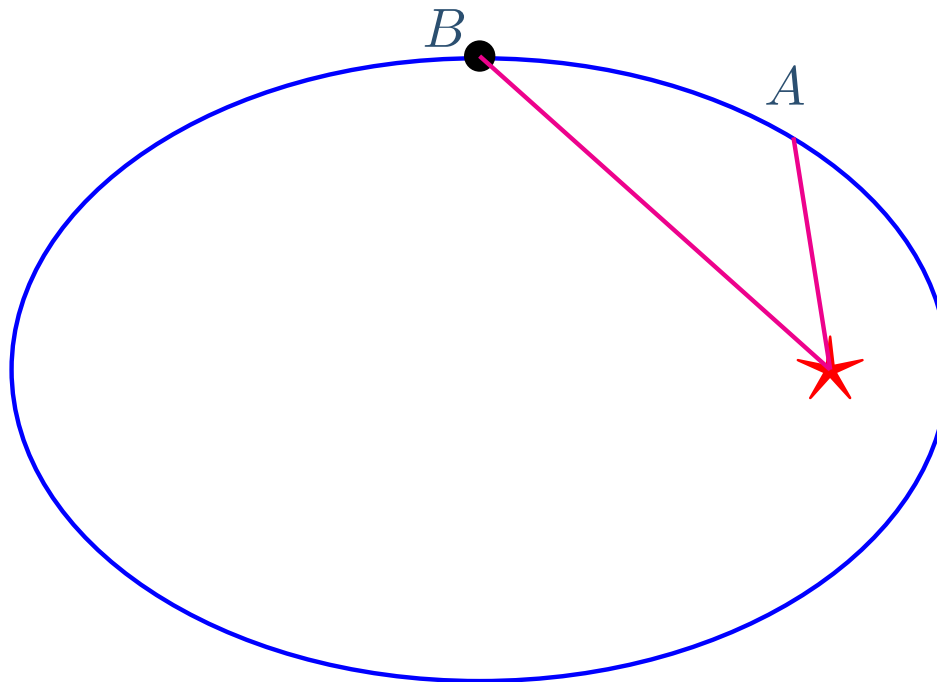
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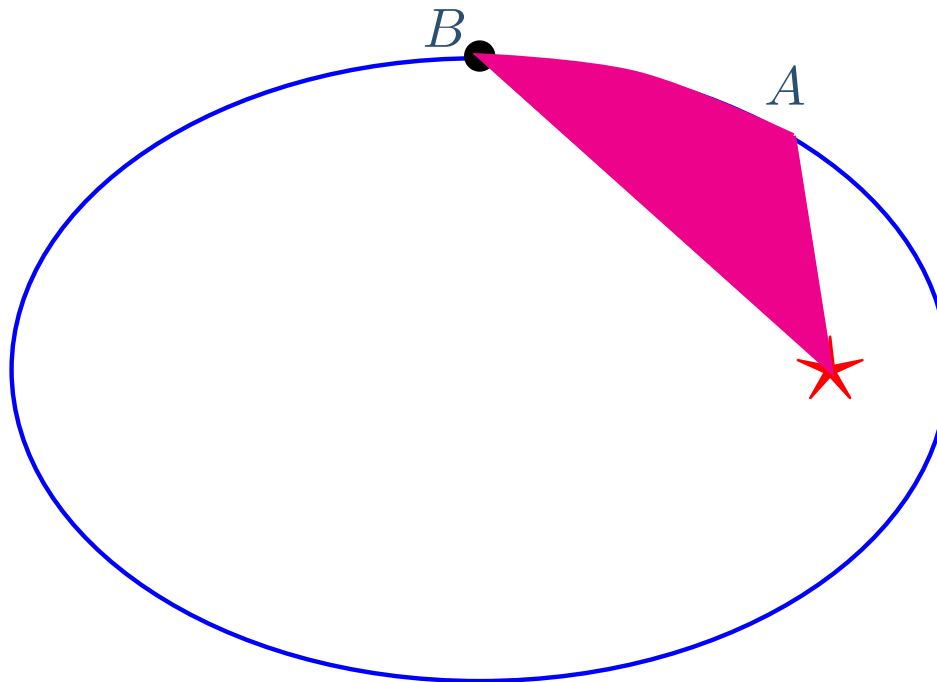
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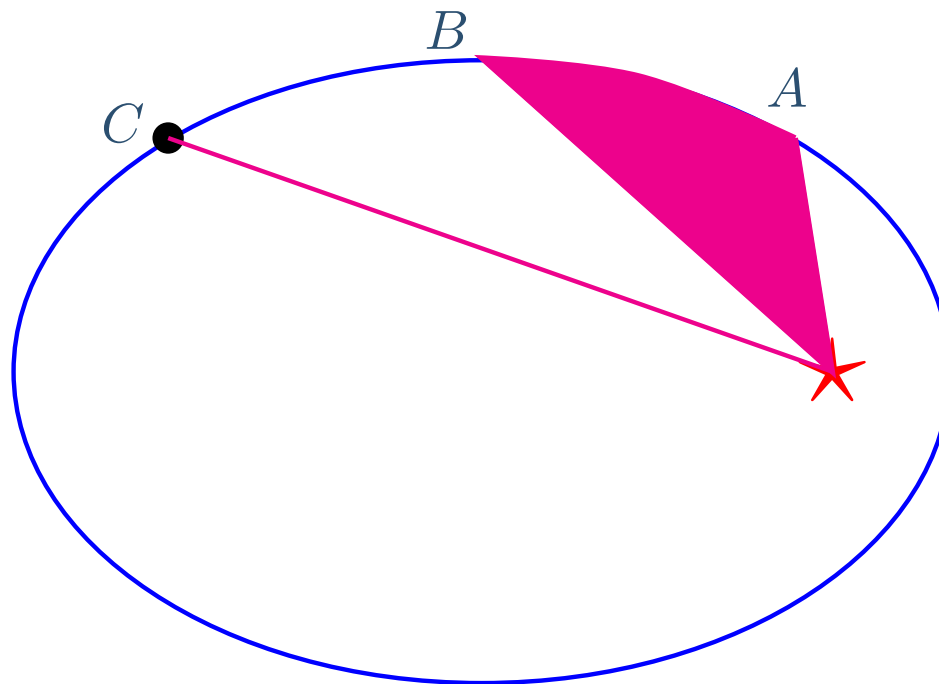
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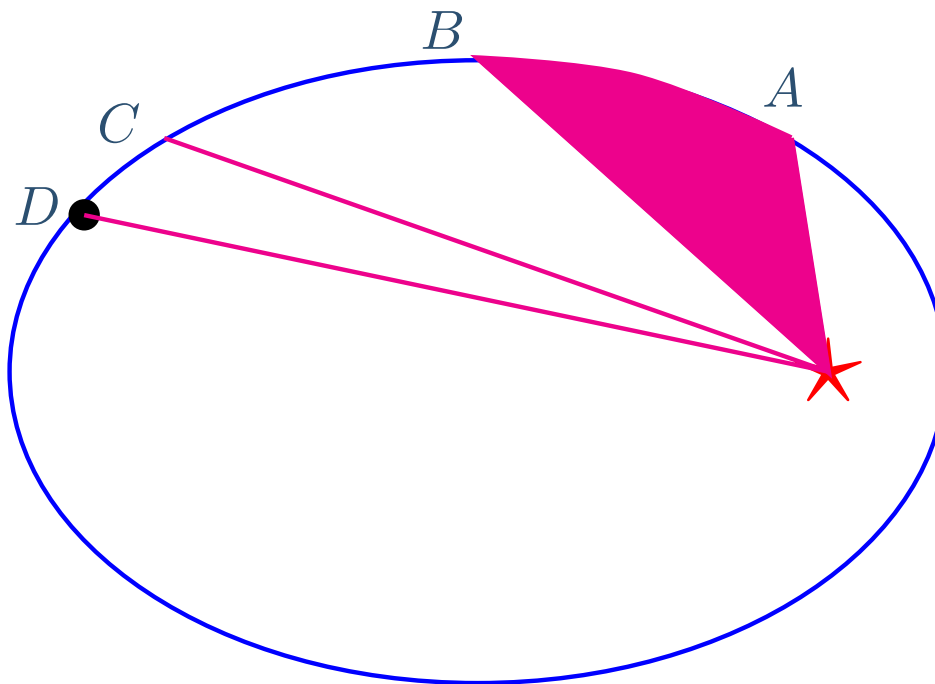
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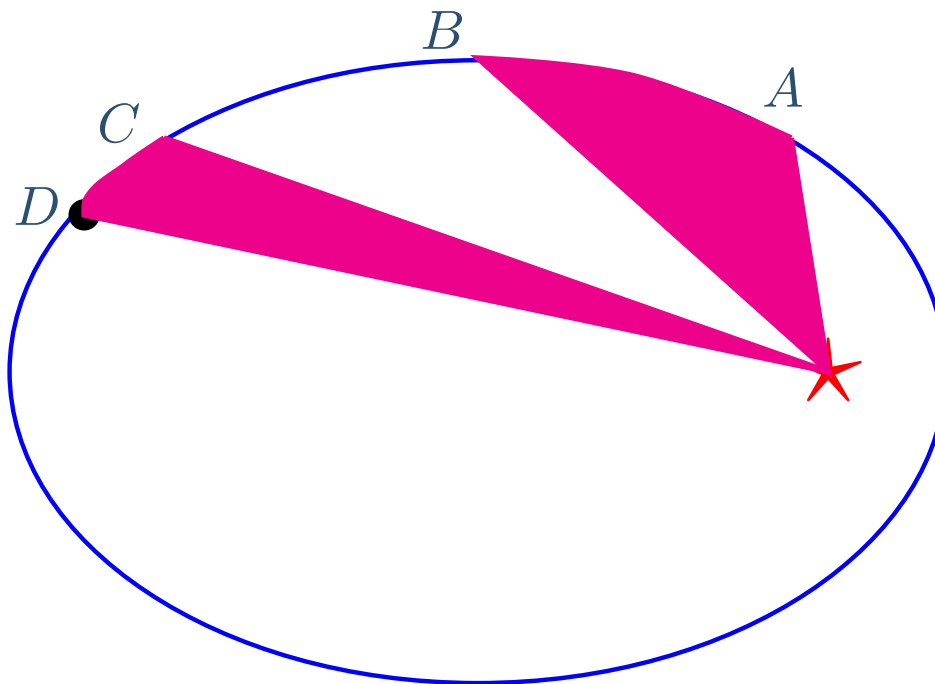
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Kepler's Second Law

Kepler's Laws:

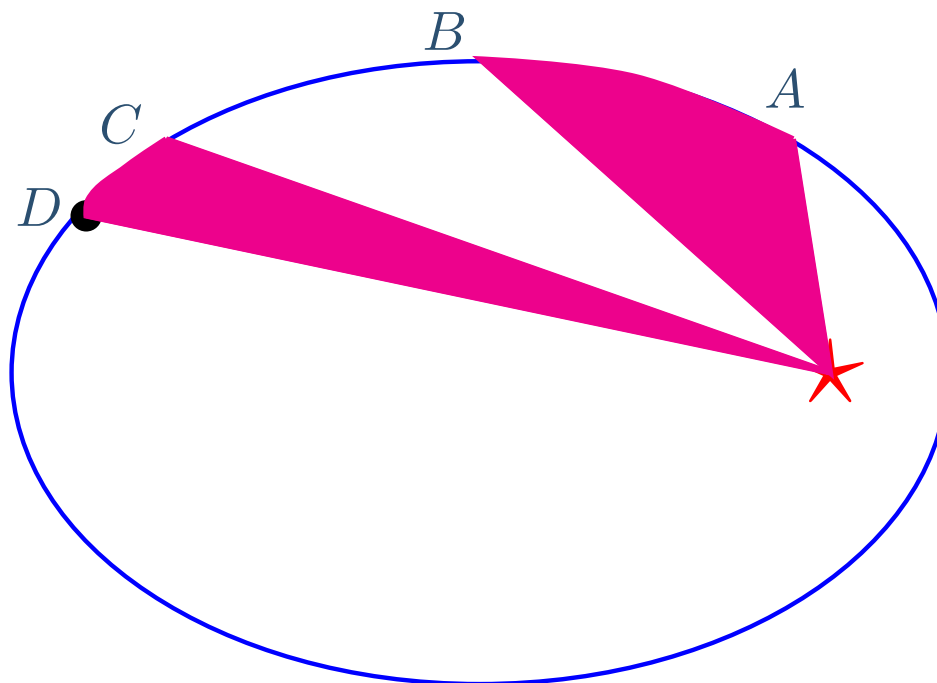
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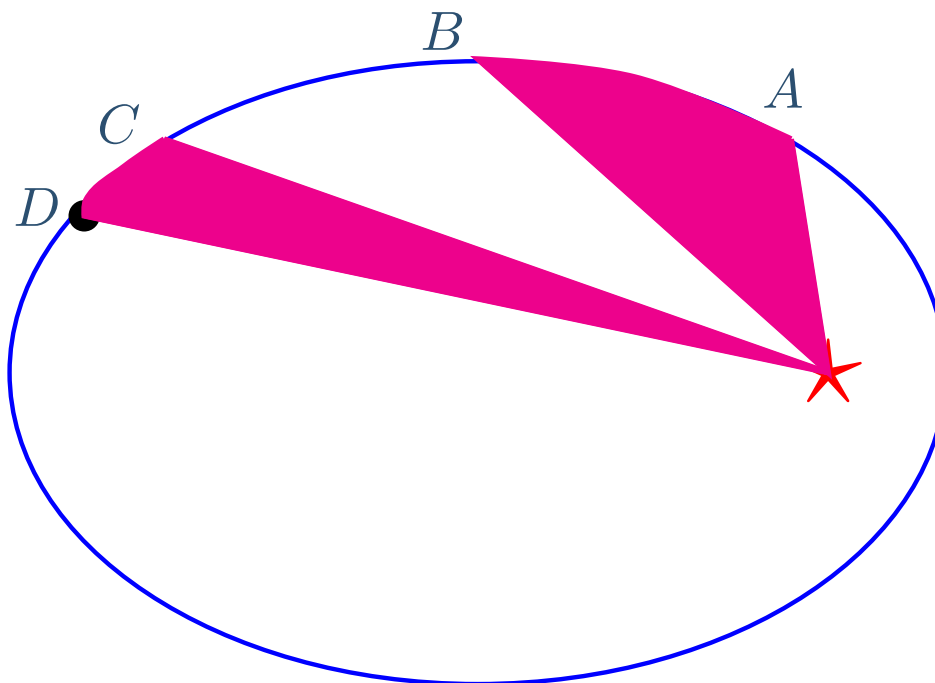


If these areas are equal then the planet takes the same amount of time going from A to B as it does going from C to D

Kepler's Second Law

Kepler's Laws:

2: The imaginary line from the sun to a planet sweeps out equal areas in equal times.



If these areas are equal then the planet takes the same amount of time going from A to B as it does going from C to D

The speed of *any* object in an elliptical orbit is NOT constant

Kepler's Second Law II

Proof:

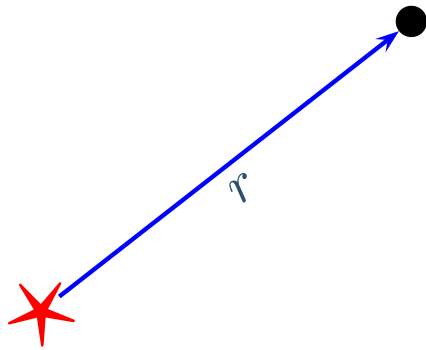
Kepler's Second Law II

Proof:



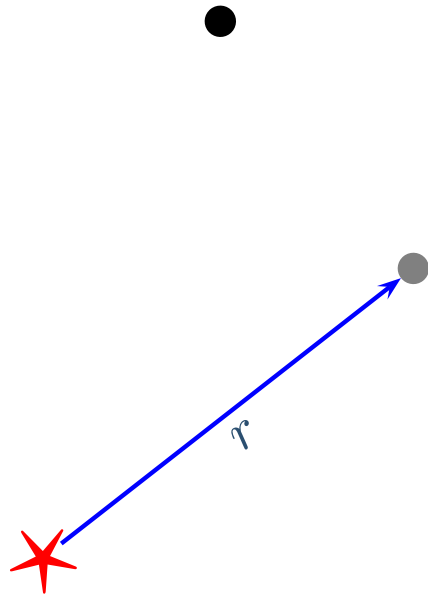
Kepler's Second Law II

Proof:



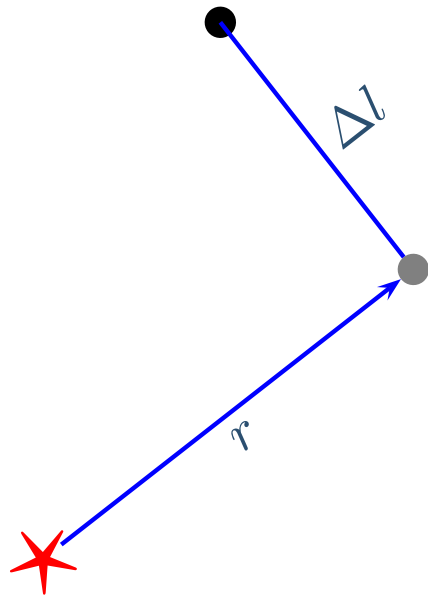
Kepler's Second Law II

Proof:



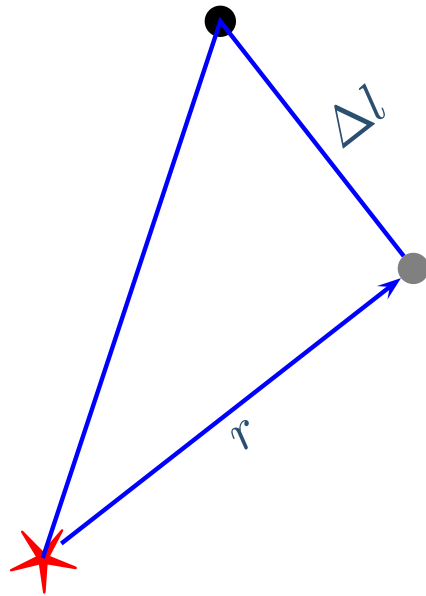
Kepler's Second Law II

Proof:



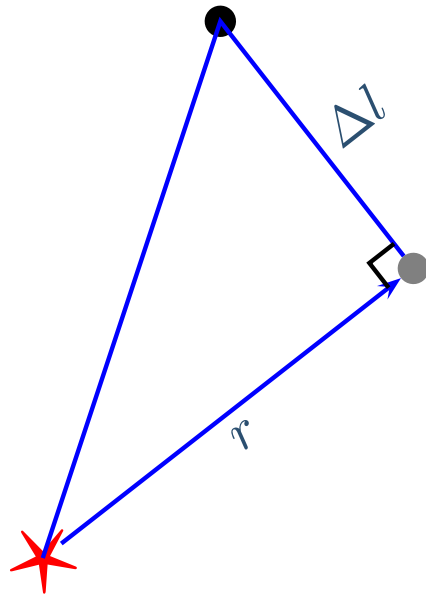
Kepler's Second Law II

Proof:



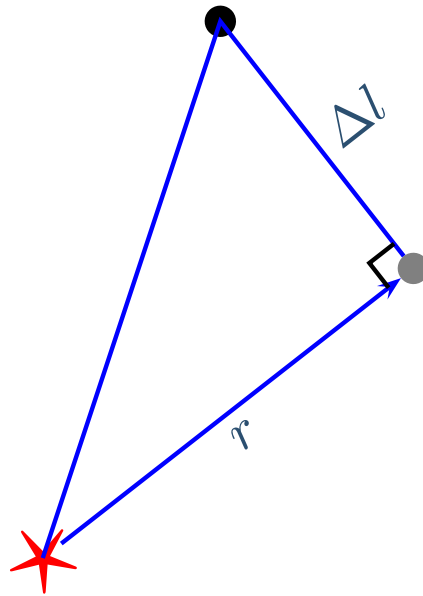
Kepler's Second Law II

Proof:



Kepler's Second Law II

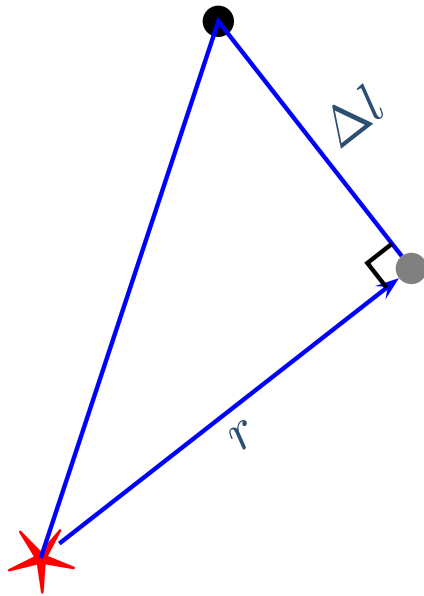
Proof:



$$A \approx \frac{1}{2}(r)(\Delta l)$$

Kepler's Second Law II

Proof:

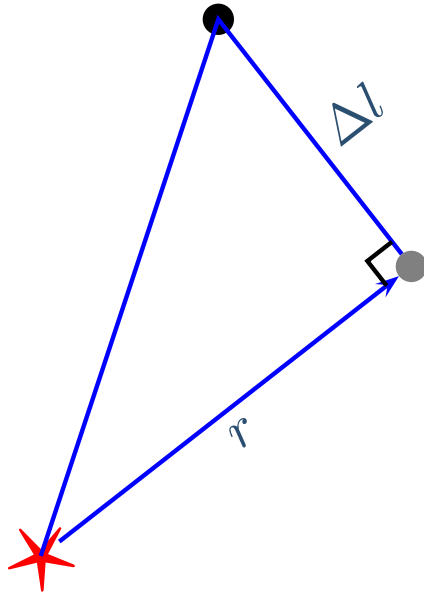


$$A \approx \frac{1}{2}(r)(\Delta l)$$

$$\frac{\Delta A}{\Delta t} = \frac{1}{2}(r) \left(\frac{\Delta l}{\Delta t} \right)$$

Kepler's Second Law II

Proof:



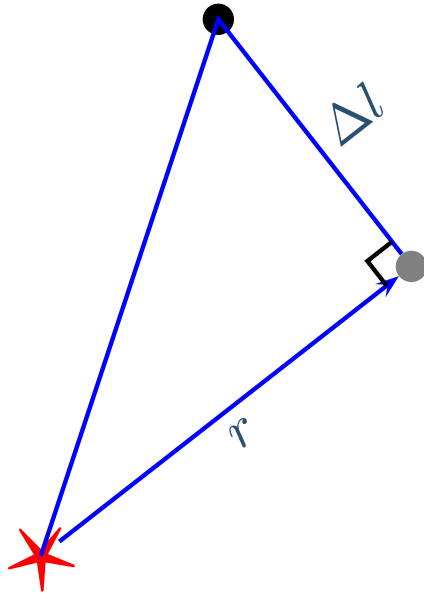
$$A \approx \frac{1}{2}(r)(\Delta l)$$

$$\frac{\Delta A}{\Delta t} = \frac{1}{2}(r) \left(\frac{\Delta l}{\Delta t} \right)$$

$$\text{As } \Delta t \rightarrow 0 \Rightarrow \frac{dA}{dt} = \frac{1}{2}(r) \left(\frac{dl}{dt} \right)$$

Kepler's Second Law II

Proof:



$$A \approx \frac{1}{2}(r)(\Delta l)$$

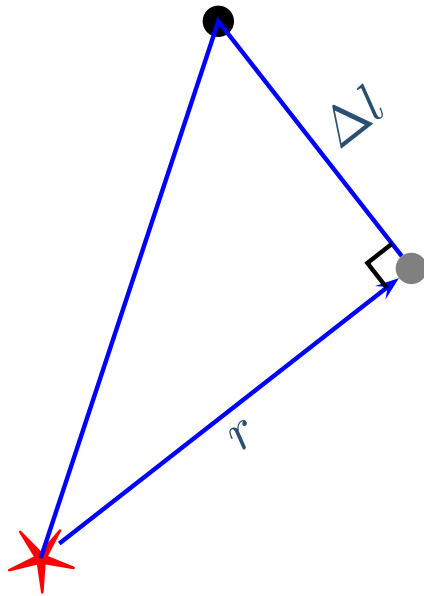
$$\frac{\Delta A}{\Delta t} = \frac{1}{2}(r) \left(\frac{\Delta l}{\Delta t} \right)$$

$$\text{As } \Delta t \rightarrow 0 \Rightarrow \frac{dA}{dt} = \frac{1}{2}(r) \left(\frac{dl}{dt} \right)$$

$$\frac{dA}{dt} = \frac{1}{2}(r)v_t$$

Kepler's Second Law II

Proof:



$$A \approx \frac{1}{2}(r)(\Delta l)$$

$$\frac{\Delta A}{\Delta t} = \frac{1}{2}(r) \left(\frac{\Delta l}{\Delta t} \right)$$

$$\text{As } \Delta t \rightarrow 0 \Rightarrow \frac{dA}{dt} = \frac{1}{2}(r) \left(\frac{dl}{dt} \right)$$

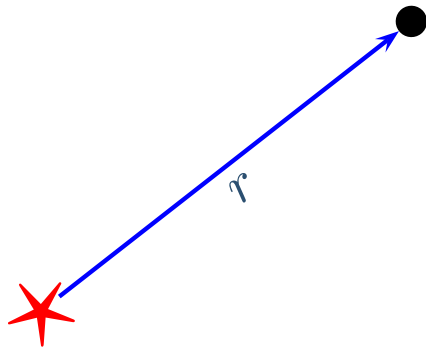
$$\frac{dA}{dt} = \frac{1}{2}(r)v_t \leftarrow \text{Tangential Velocity}$$

Kepler's Second Law III

Proof:

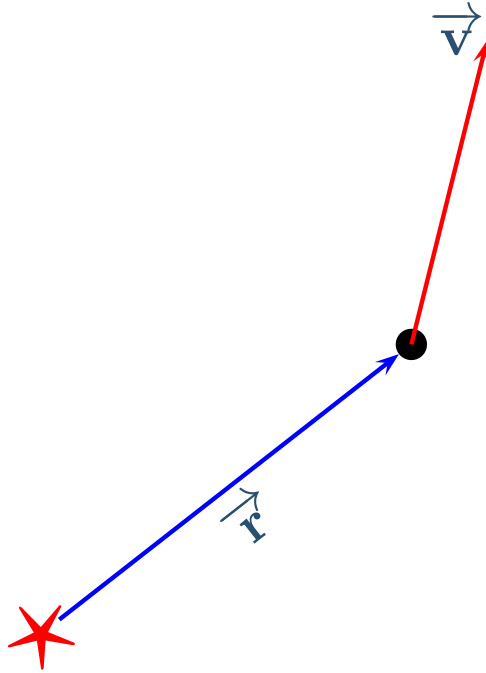
$$\frac{dA}{dt} = \frac{1}{2}(r)v_t$$

On an ellipse, \vec{r} and \vec{v} are *not* perpendicular!



Kepler's Second Law III

Proof:

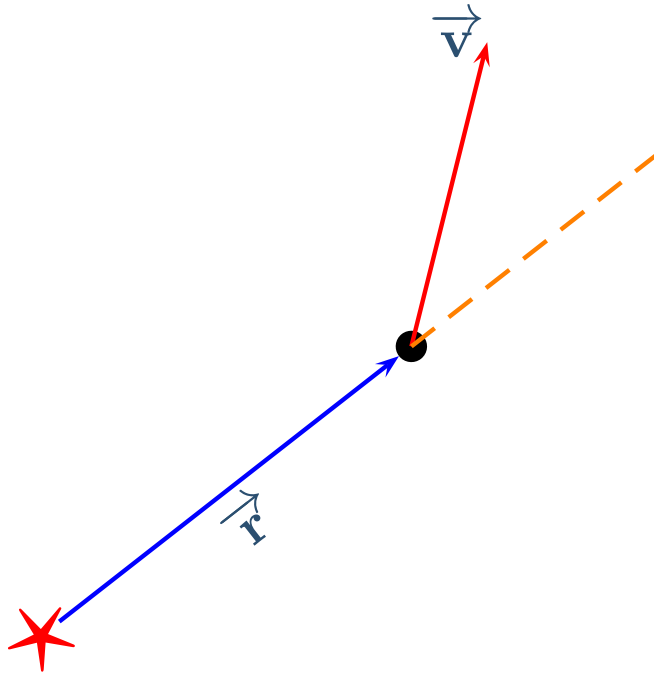


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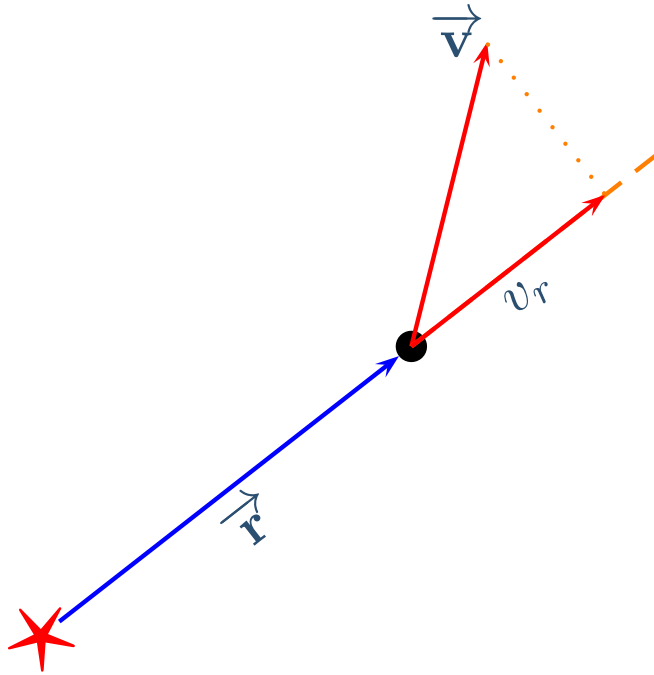


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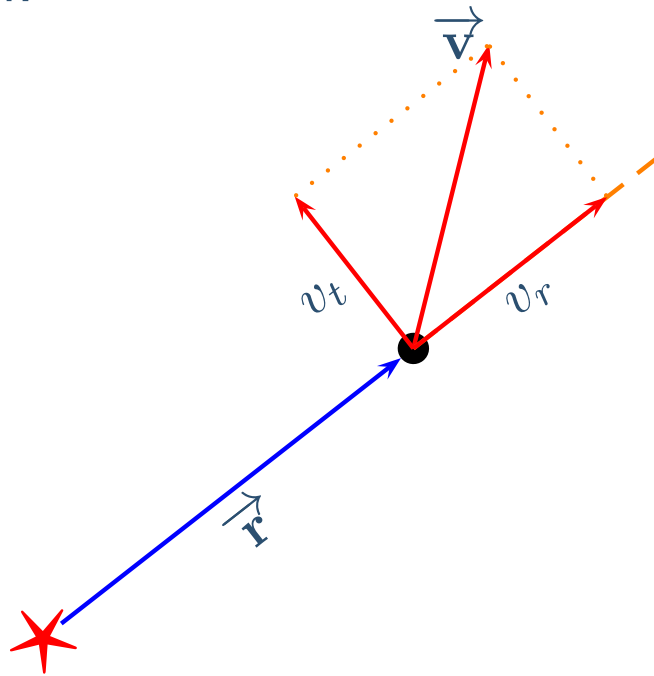


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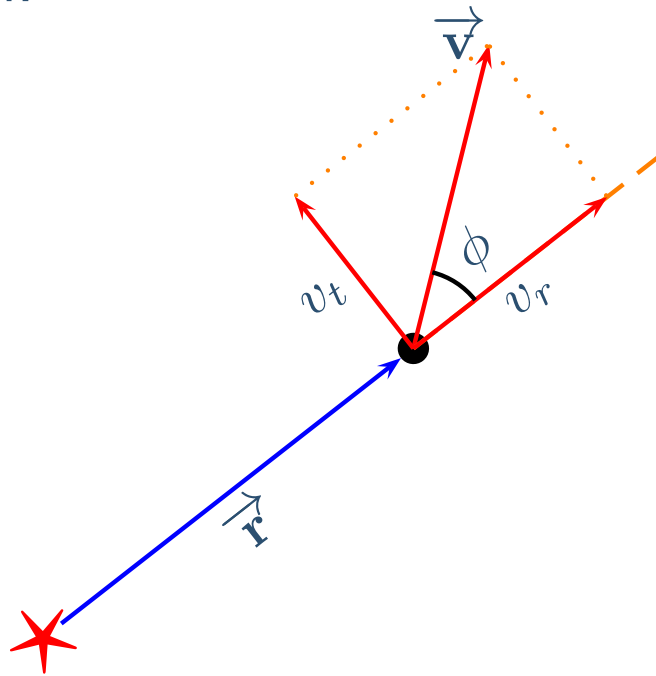


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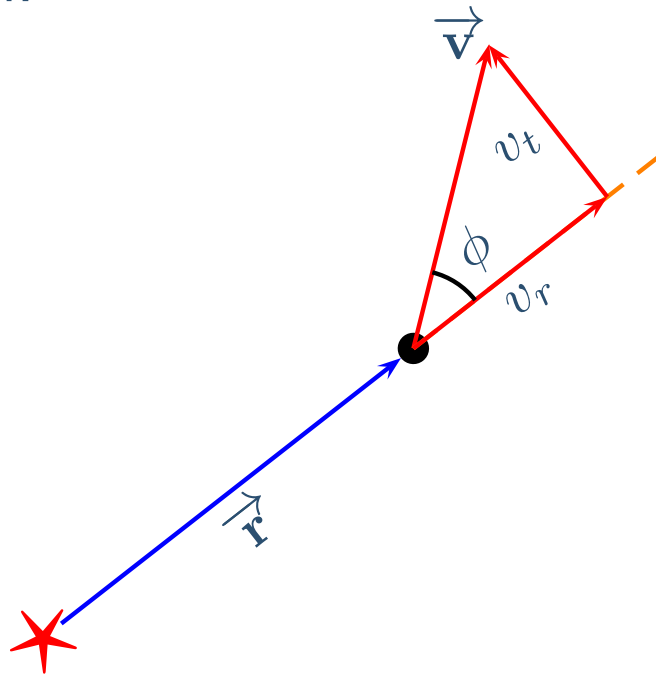


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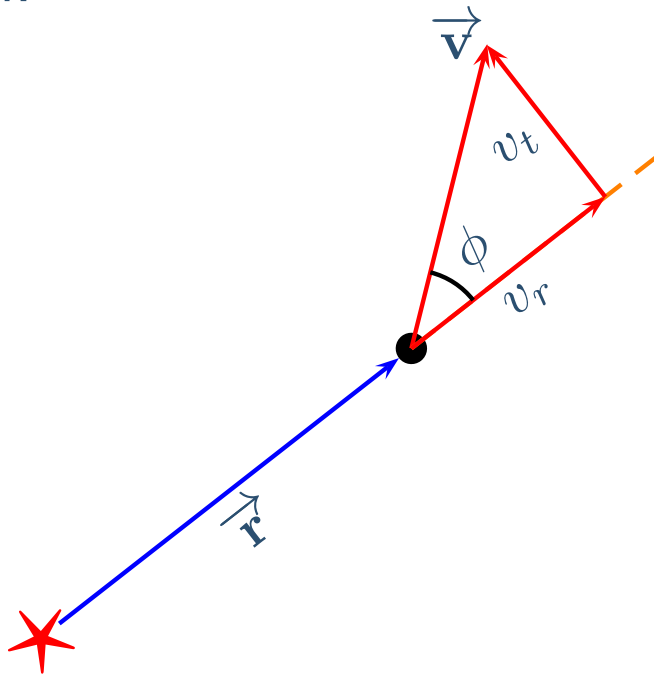


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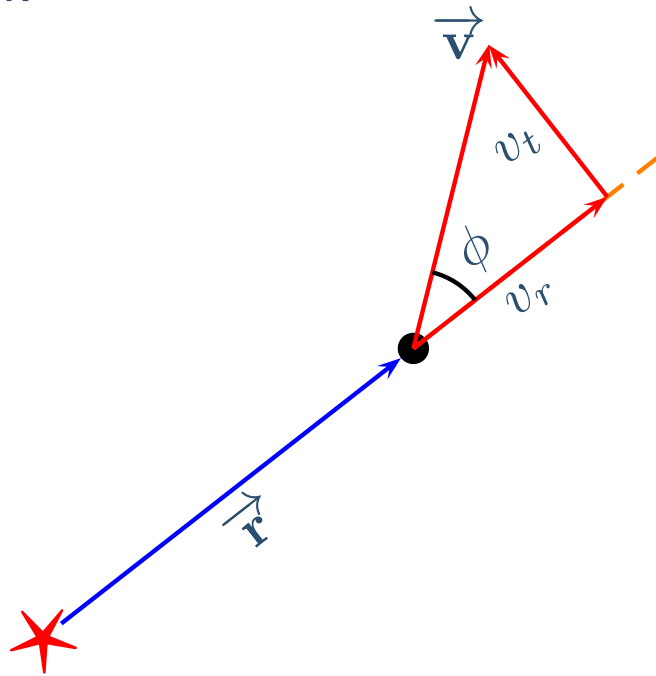
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On an ellipse, \vec{r} and \vec{v} are *not* perpendicular!

$$v_t = v \sin \phi$$

Kepler's Second Law III

Proof:



$$\frac{dA}{dt} = \frac{1}{2}(r)v_t$$

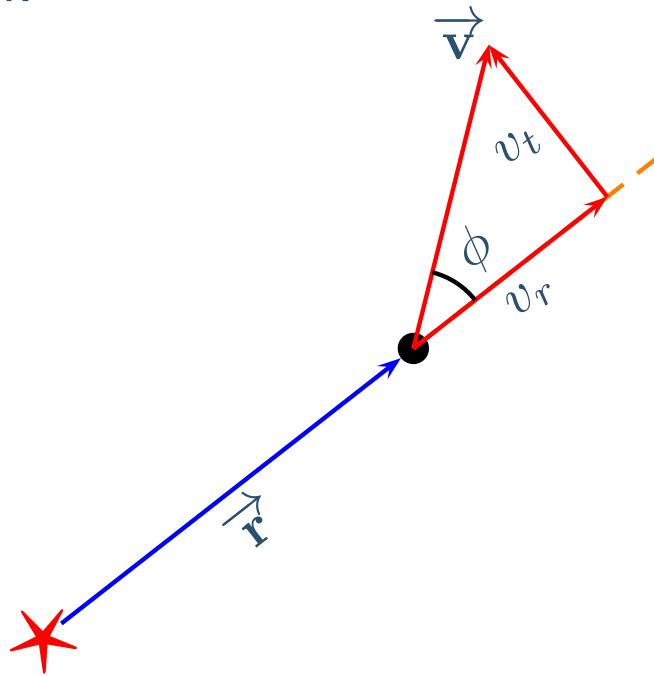
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$$\frac{dA}{dt} = \frac{1}{2}rv \sin \phi$$

Kepler's Second Law III

Proof:



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On an ellipse, \vec{r} and \vec{v} are *not* perpendicular!

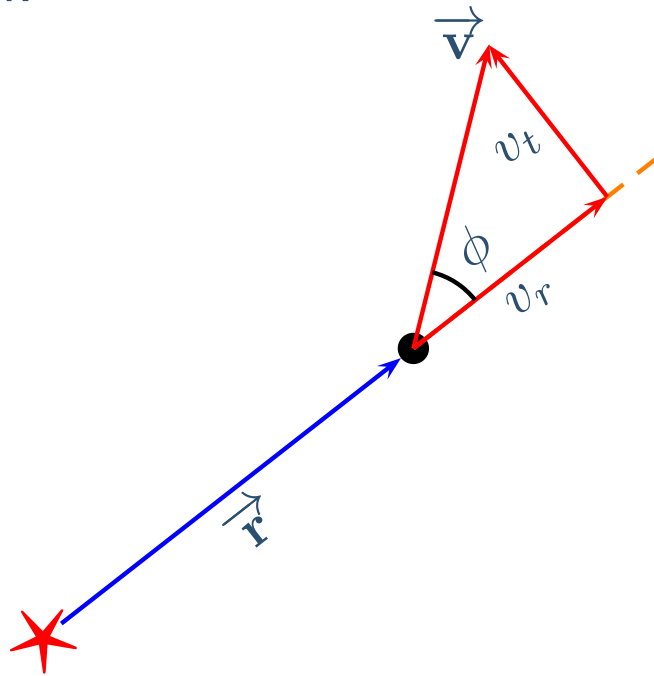
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Angular Momentum of a
Satellite: $L = Mvr \sin \phi$

Kepler's Second Law III

Proof:



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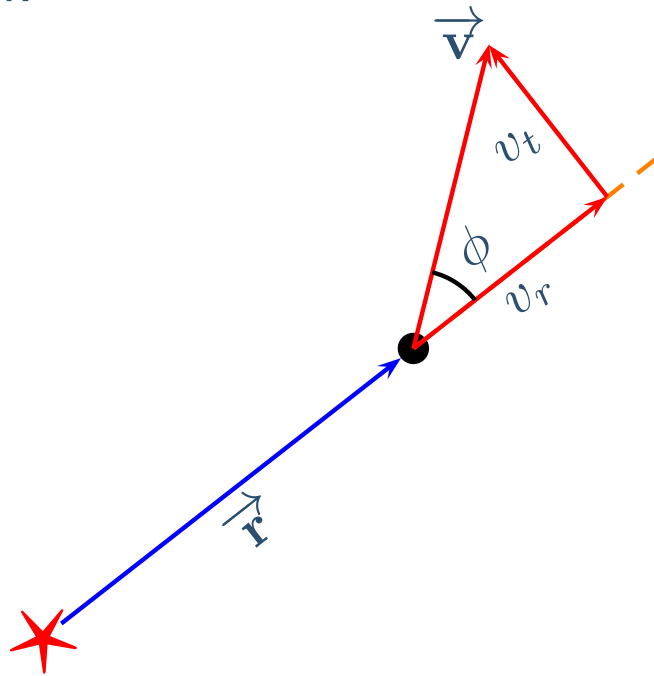
$$v_t = v \sin \phi$$

$$\frac{dA}{dt} = \frac{1}{2M} M v r \sin \phi$$

Angular Momentum of a
Satellite: $L = M v r \sin \phi$

Kepler's Second Law III

Proof:



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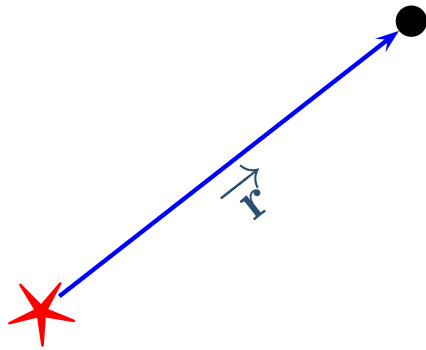
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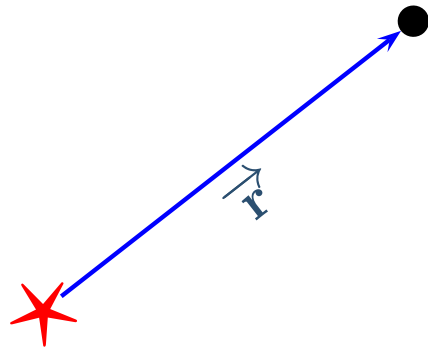
$$\frac{dA}{dt} = \frac{L}{2M}$$

Kepler's Second Law IV

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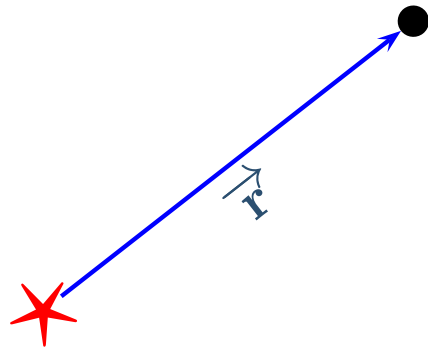
Kepler's Second Law IV



$$\frac{dA}{dt} = \frac{L}{2M}$$

$$\vec{\tau} = \frac{d\vec{L}}{dt}$$

Kepler's Second Law IV

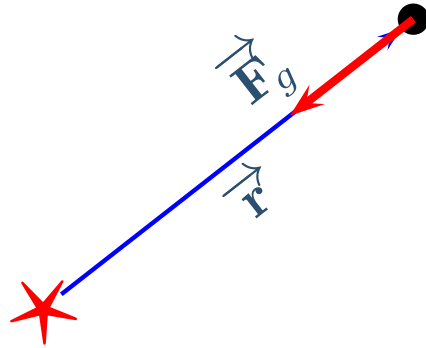


$$\frac{dA}{dt} = \frac{L}{2M}$$

$$\vec{\tau} = \frac{d\vec{L}}{dt}$$

On satellites, gravity causes no torque

Kepler's Second Law IV

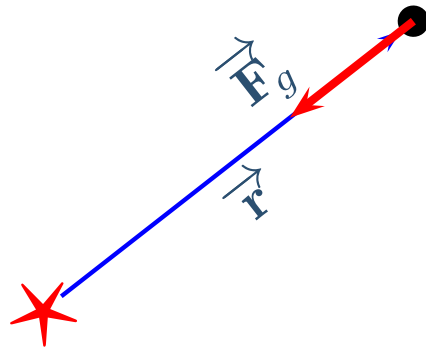


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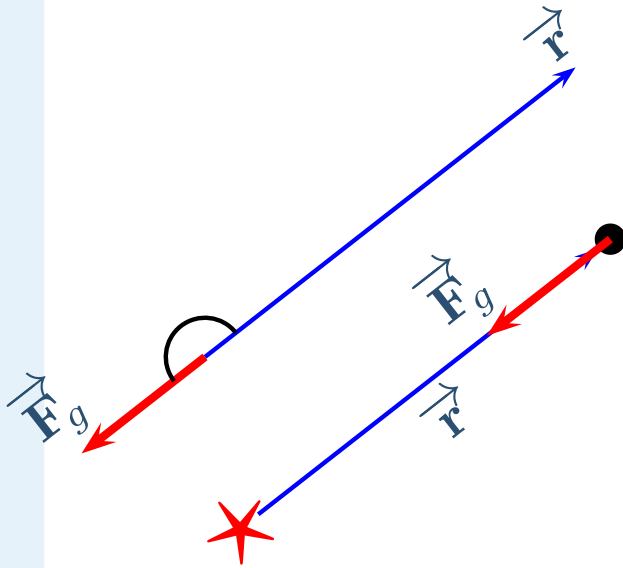
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$$\vec{\tau} = \vec{r} \times \vec{F}_g$$

Kepler's Second Law IV



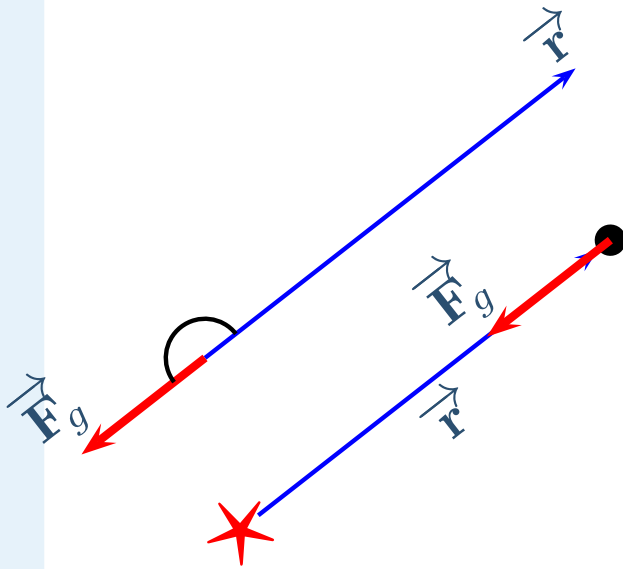
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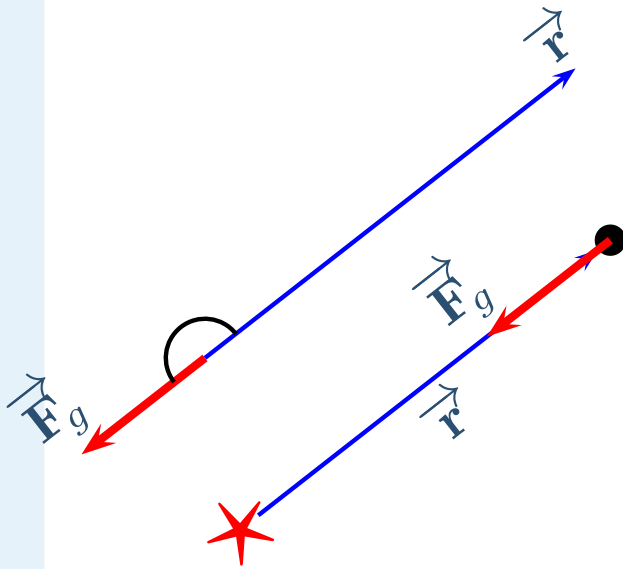
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Kepler's Second Law IV



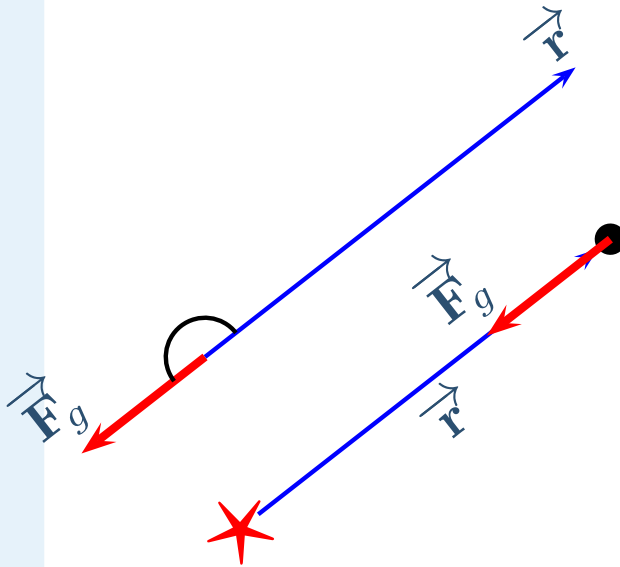
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On satellites, gravity causes no torque

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Kepler's Second Law IV



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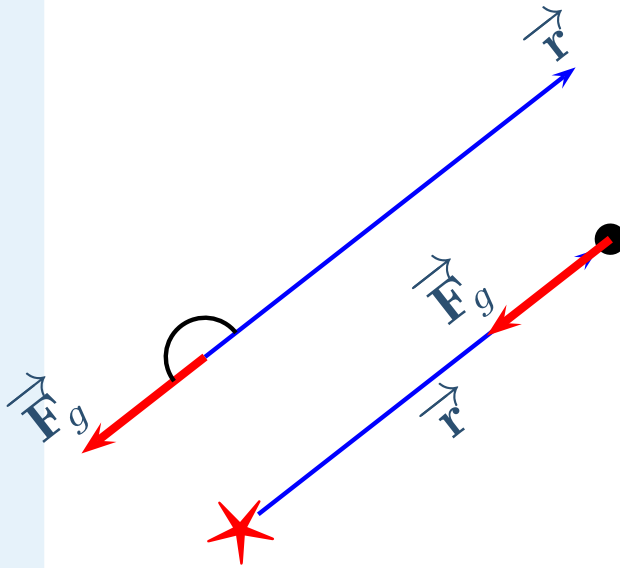
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The angular momentum of a satellite is constant

Kepler's Second Law IV



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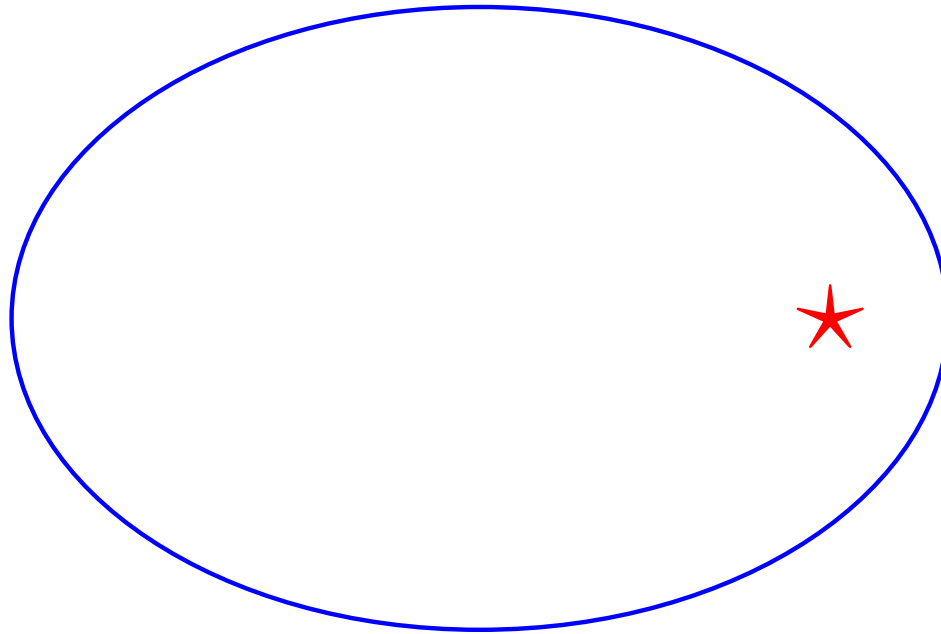
$$\frac{dA}{dt} = \text{constant} \Rightarrow \text{equal } \Delta A \text{ for equal } \Delta t$$

Kepler's Second Law V

At perihelion and aphelion the angle ϕ for satellite motion is 90° .

Kepler's Second Law V

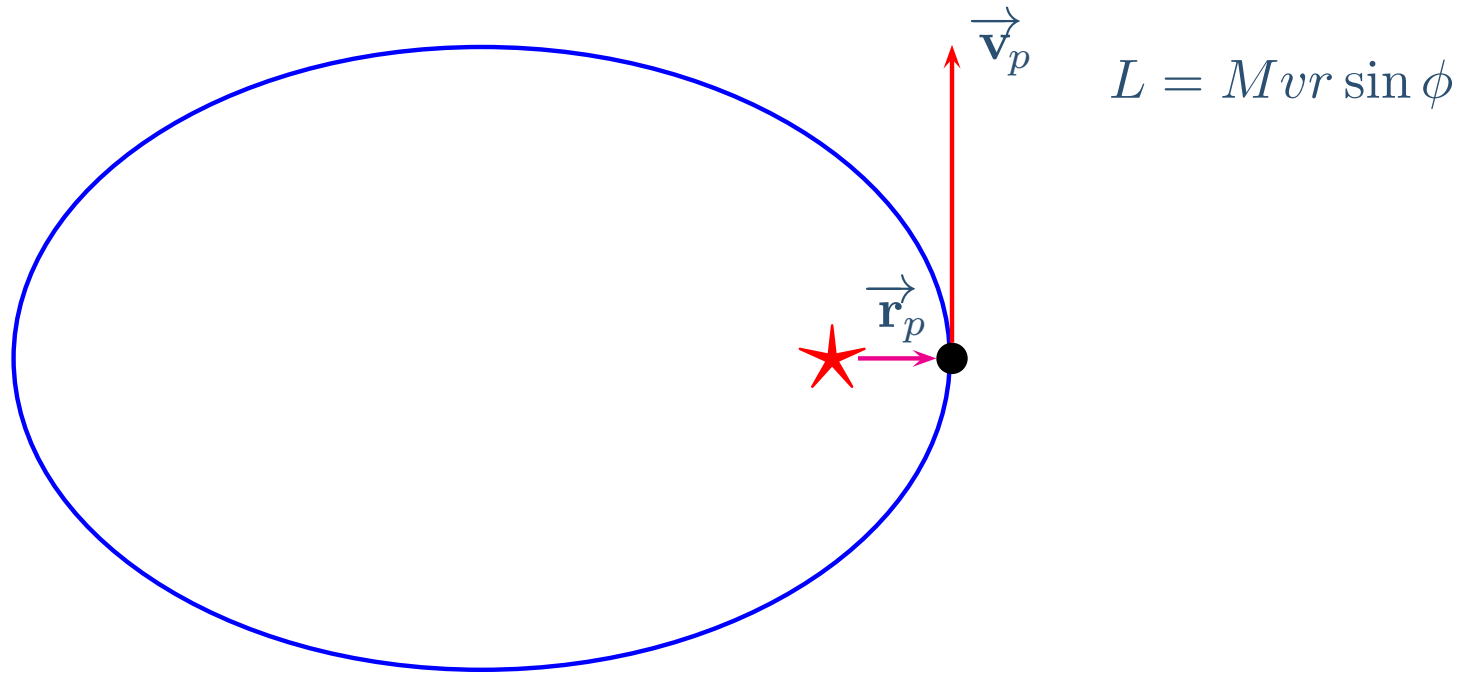
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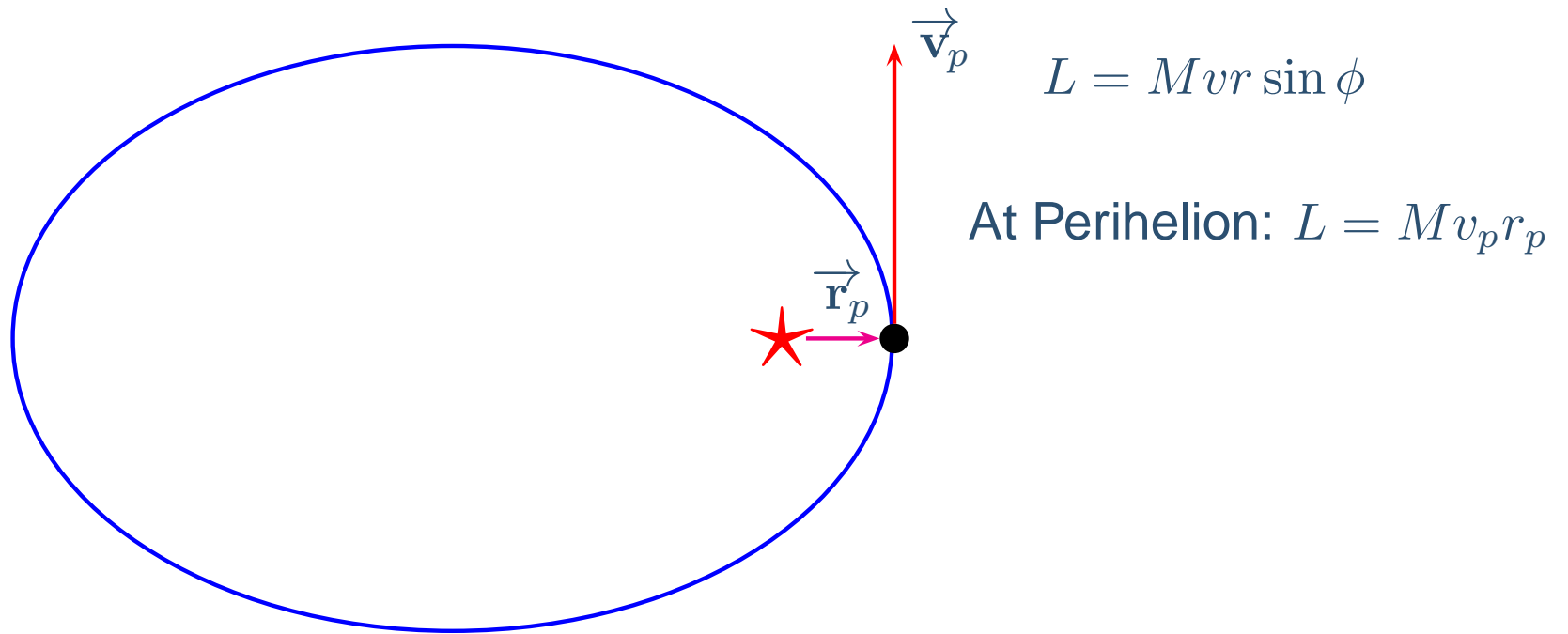
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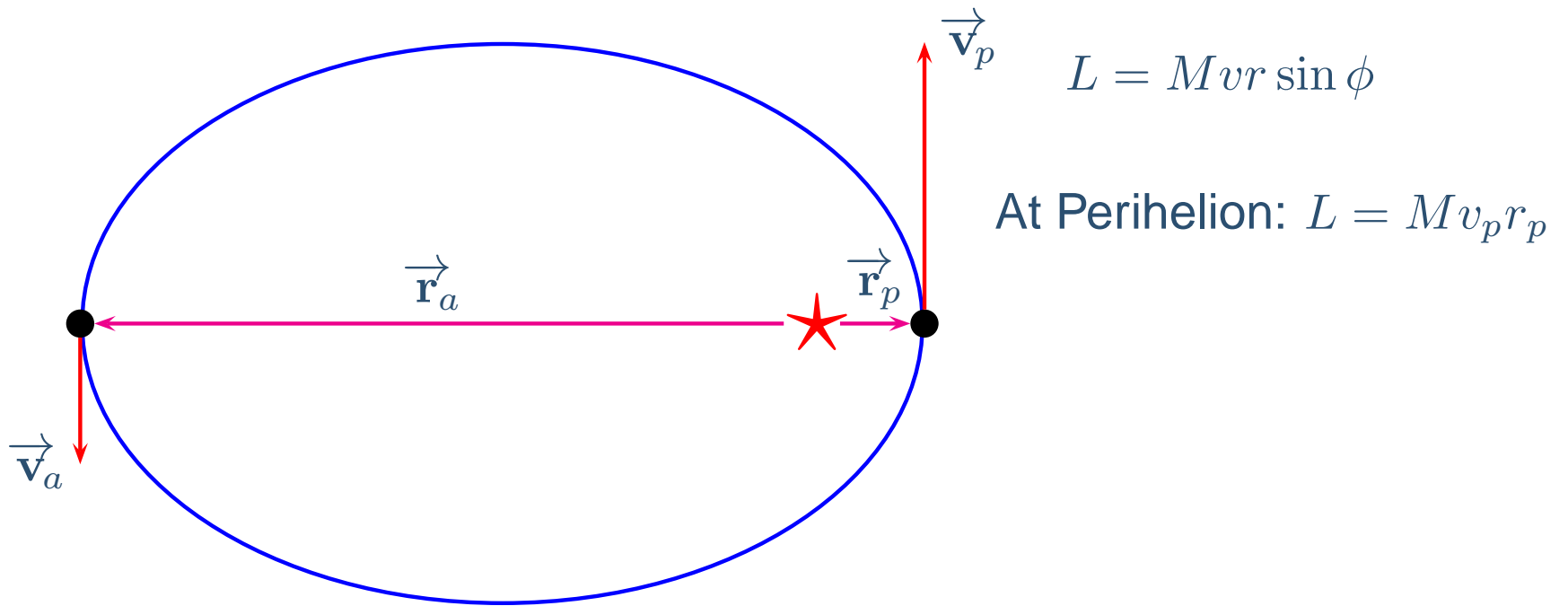
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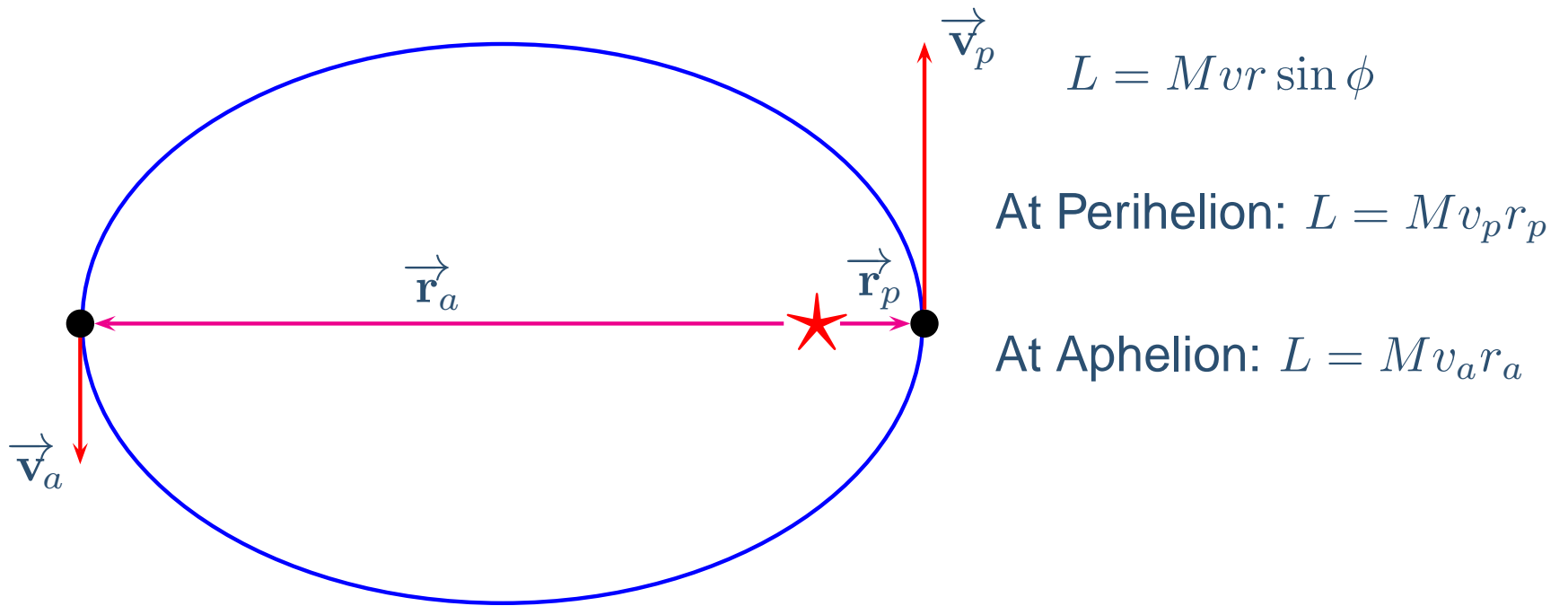
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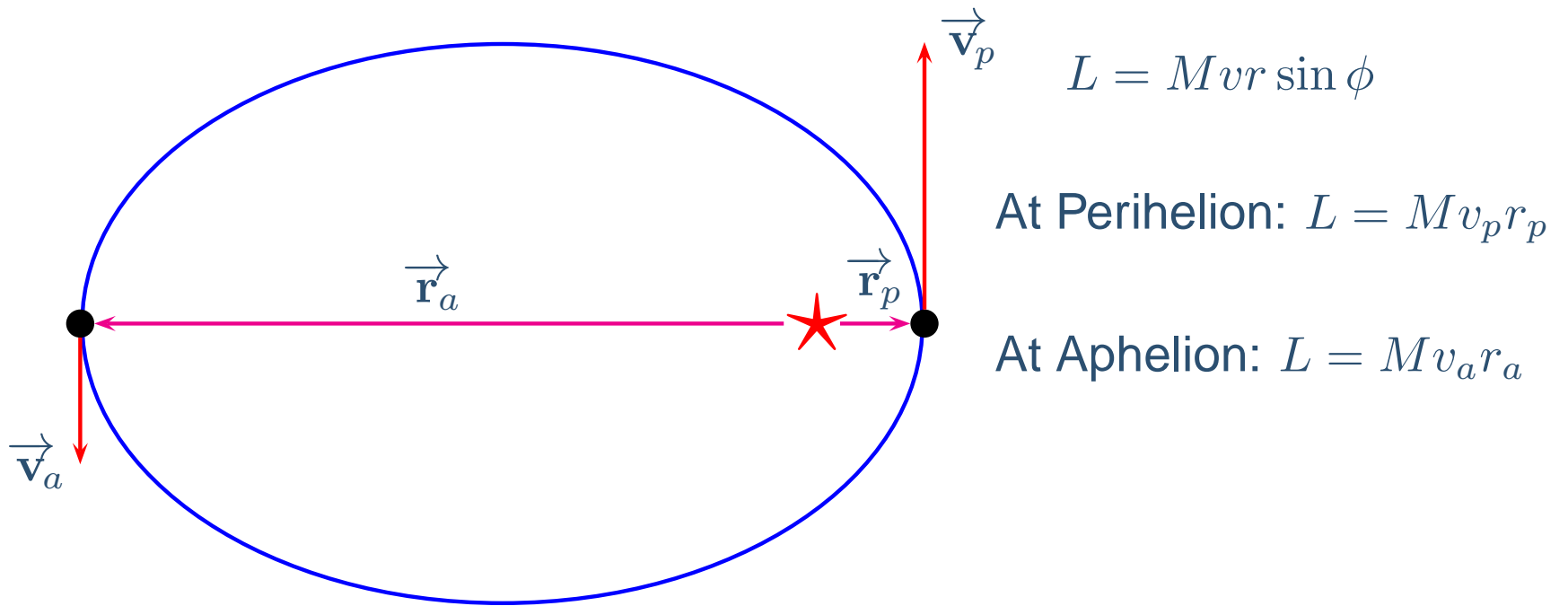
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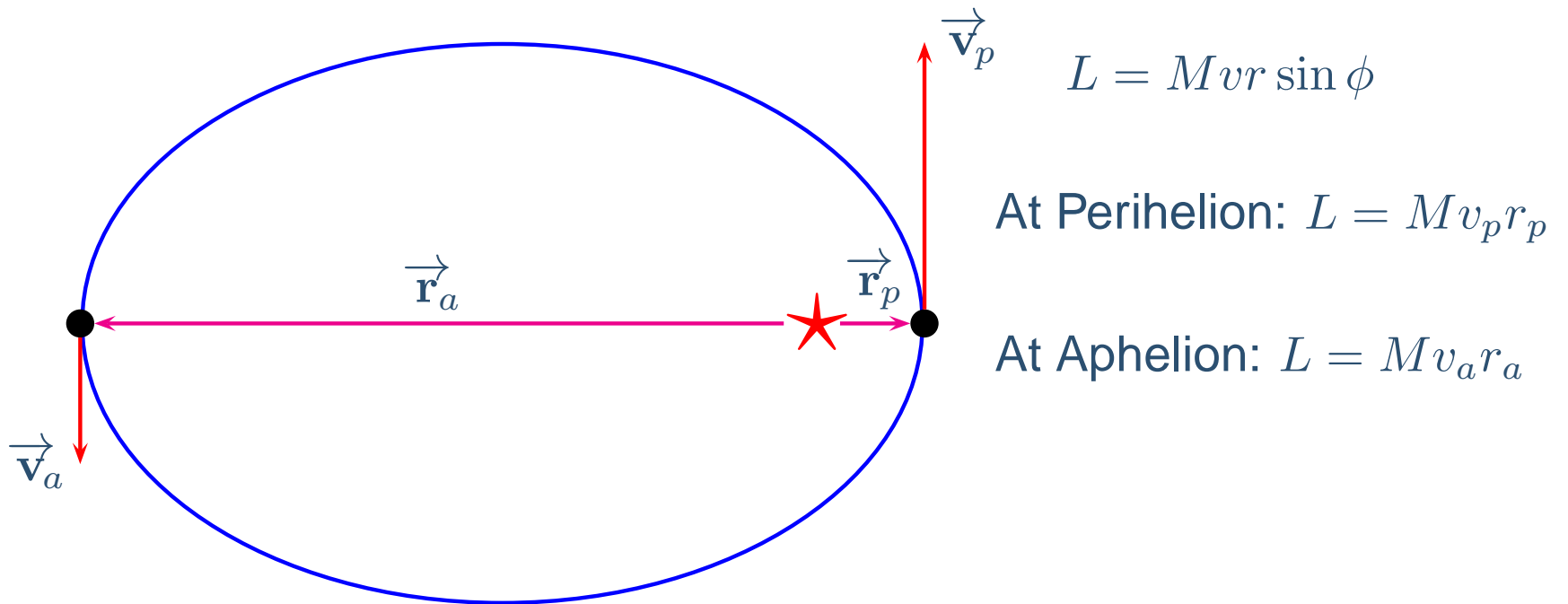


Conservation of Angular Momentum

$$\Rightarrow Mv_p r_p = Mv_a r_a$$

Kepler's Second Law V

At perihelion and aphelion the angle ϕ for satellite motion is 90° .



Conservation of Angular Momentum

$$\Rightarrow Mv_p r_p = Mv_a r_a$$

$$\Rightarrow \boxed{v_p r_p = v_a r_a}$$

Kepler's Third Law

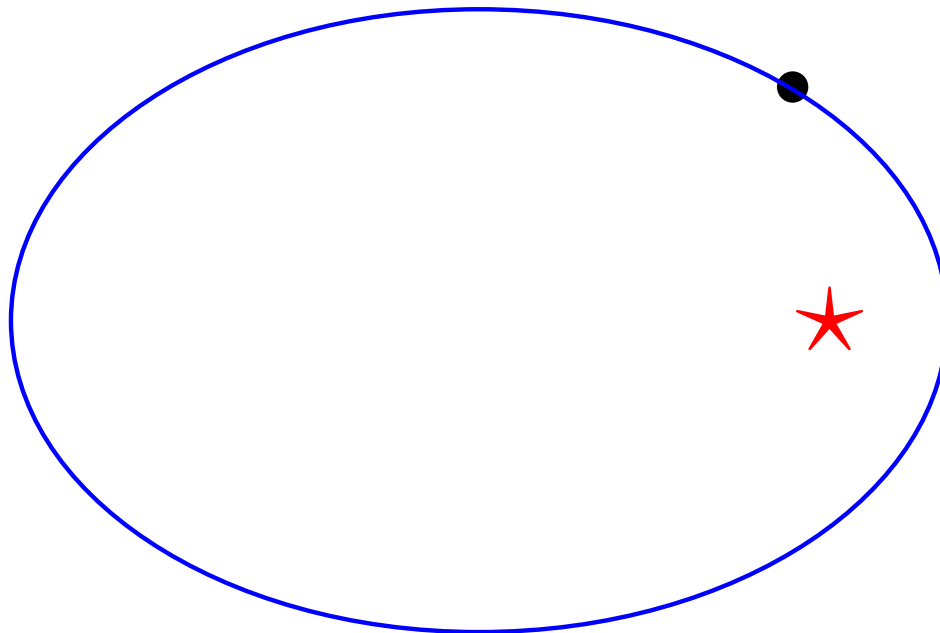
Kepler's Laws:

3: The period of the planet's motion is proportional to the orbit's semi-major axis to the $\frac{3}{2}$ power.

Kepler's Third Law

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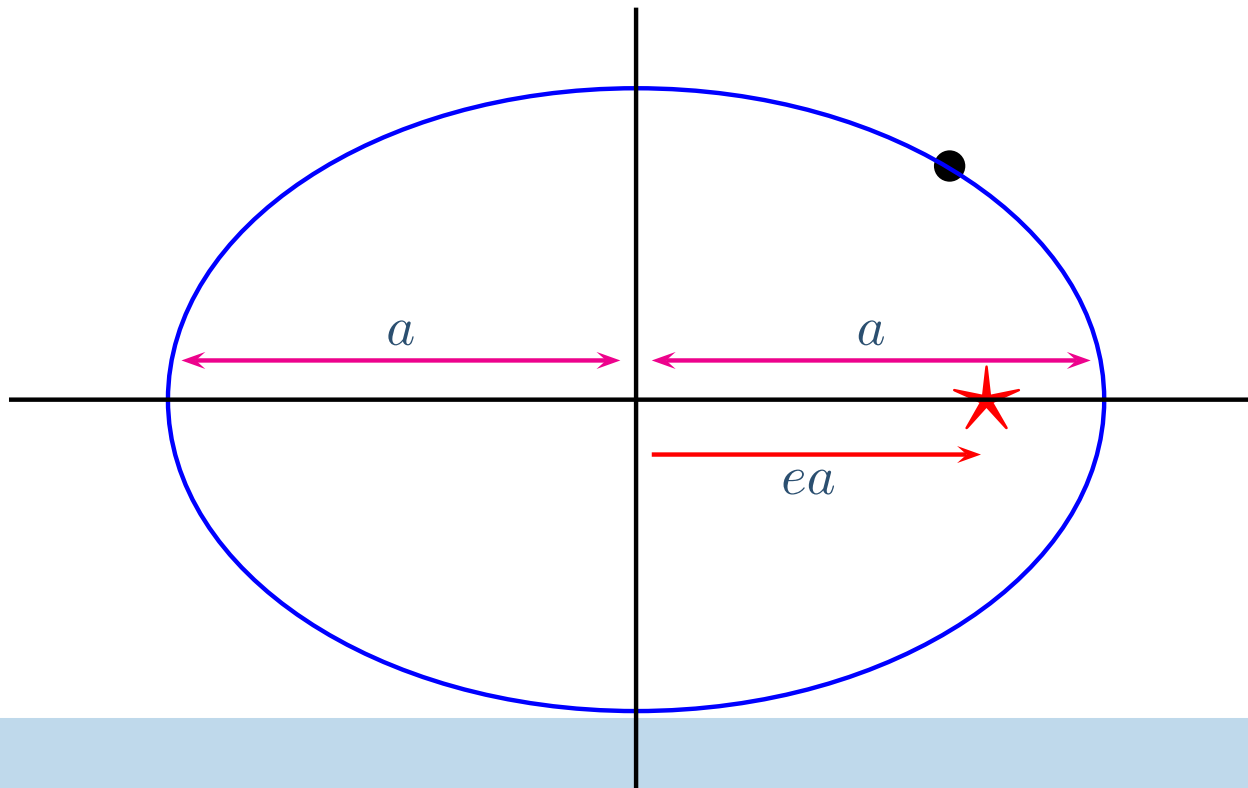
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Kepler's Third Law

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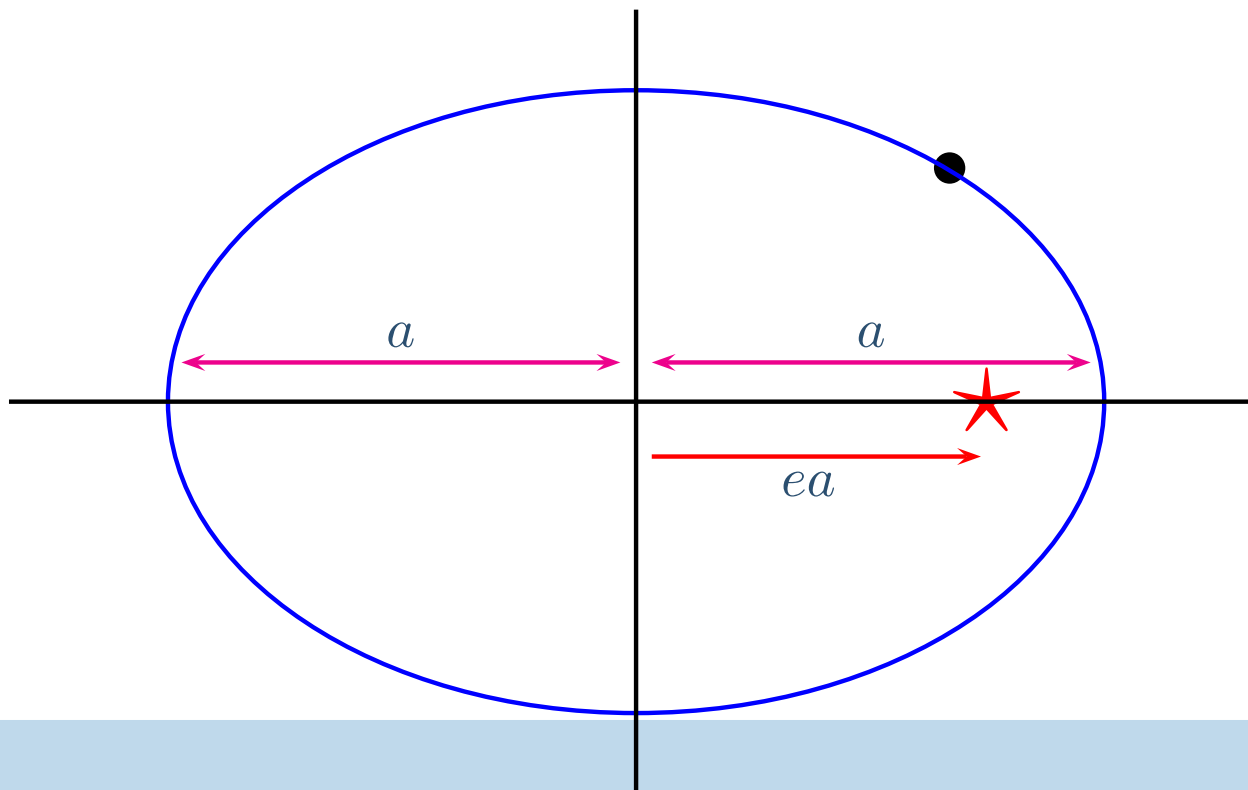
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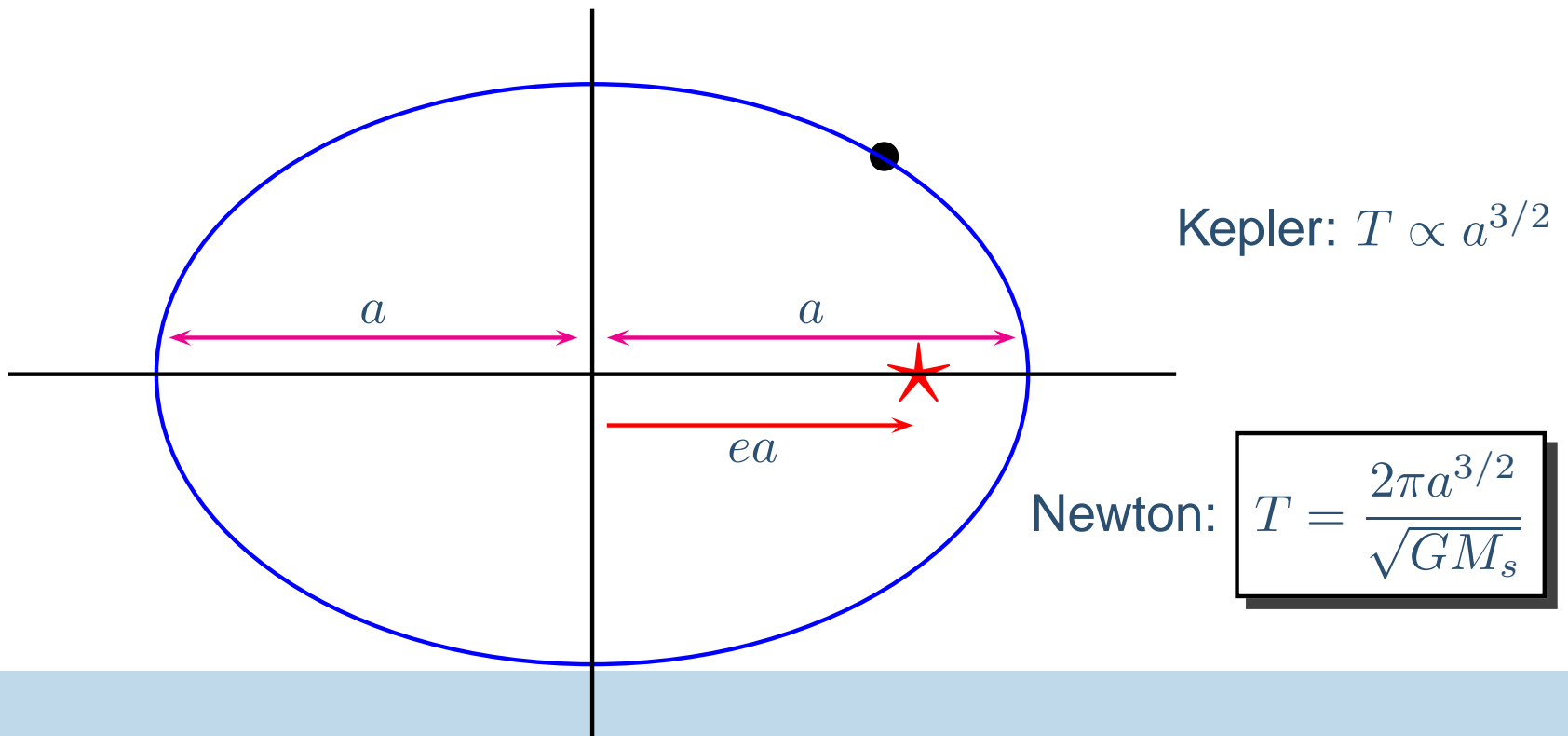


$$\text{Kepler: } T \propto a^{3/2}$$

Kepler's Third Law

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