April 30, Week 15

Today: Chapter 13, Newton's Law of Gravity

Homework #11 due Monday, May 7 at 11:59pm Mastering Physics: 5 problems from chapter 13 Written: none.

Kepler's Laws

Before Newton, all astronomical work had been observational. Using the data of Danish astronomer Tycho Brahe (1546-1601), the German mathematician Johannes Kepler (1571-1630) was able to deduce (but not explain), three statements about planetary motion.

Kepler's Laws

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Kepler's Laws:

- 1: Each planet's orbit traces out the shape of an ellipse with the sun located at one focus.
- 2: The imaginary line from the sun to a planet sweeps out equal areas in equal times.
- 3: The period of the planet's motion is proportional to the orbit's semi-major axis to the $\frac{3}{2}$ power.

Ellipses - The Algebraic Approach

Ellipse - ovals.



Kepler's Laws:

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2: The imaginary line from the sun to a planet sweeps out equal areas in equal times.



If these areas are equal then the planet takes the same amount of time going from *A* to *B* as it does going from *C* to *D*

Kepler's Laws:

2: The imaginary line from the sun to a planet sweeps out equal areas in equal times.



If these areas are equal then the planet takes the same amount of time going from *A* to *B* as it does going from *C* to *D*

The speed of *any* object in an elliptical orbit is NOT constant

Proof:

Proof:



Newton's Law of Gravity - p. 6/10

Proof:



Proof:





Proof:



Proof:











$$\frac{dA}{dt} = \frac{1}{2}(r)v_t$$



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$$\frac{dA}{dt} = \frac{1}{2}rv\sin\phi$$



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On an ellipse, $\overrightarrow{\mathbf{r}}$ and $\overrightarrow{\mathbf{v}}$ are *not* perpendicular!

$$v_t = v \sin \phi$$
$$\frac{dA}{dt} = \frac{1}{2}rv \sin \phi$$

Angular Momentum of a Satellite: $L = Mvr \sin \phi$



$$\frac{dA}{dt} = \frac{1}{2}(r)v_t$$

On an ellipse, $\overrightarrow{\mathbf{r}}$ and $\overrightarrow{\mathbf{v}}$ are *not* perpendicular!

$$v_t = v \sin \phi$$

$$\frac{dA}{dt} = \frac{1}{2M}Mvr\sin\phi$$

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$$\frac{dA}{dt} = \frac{L}{2M}$$
$$\overrightarrow{\tau} = \frac{\overrightarrow{dL}}{dt}$$

$$\overrightarrow{\boldsymbol{\tau}} = \overrightarrow{\mathbf{r}} \times \overrightarrow{\mathbf{F}}_g \Rightarrow \tau = rF_g \sin 180^\circ$$



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On satellites, gravity causes no torque

$$\overrightarrow{\boldsymbol{\tau}} = \overrightarrow{\mathbf{r}} \times \overrightarrow{\mathbf{F}}_g \Rightarrow \tau = rF_g \sin 180^\circ = 0$$

The angular momentum of a satellite is constant



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$$\vec{\tau} = \frac{\vec{dL}}{dt}$$

On satellites, gravity causes no torque

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The angular momentum of a satellite is constant

$$\frac{dA}{dt} = constant \Rightarrow$$

equal ΔA for equal Δt











At perihelion and aphelion the angle ϕ for satellite motion is 90° .



 $L = M v r \sin \phi$

At Perihelion: $L = M v_p r_p$

At Aphelion: $L = M v_a r_a$



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Conservation of Angular Momentum

$$\Rightarrow M v_p r_p = M v_a r_a$$
$$\Rightarrow v_p r_p = v_a r_a$$

Kepler's Laws:

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